

# 2025 33 Years **Chapter-Wise** Solved Papers (1992-2024)

# Electrical Engineering



Title : GATE 2025 : Electrical Engineering - 33 Years' Chapterwise Solved Papers (1992-2024)

Language : English

Editor's Name : GKP Editorial Team

Copyright © : 2025 CLIP

No part of this book may be reproduced in a retrieval system or transmitted, in any form or by any means, electronics, mechanical, photocopying, recording, scanning and or without the written permission of the Author/Publisher.

#### Typeset & Published by :

# Career Launcher Infrastructure (P) Ltd.

A-45, Mohan Cooperative Industrial Area, Near Mohan Estate Metro Station, New Delhi - 110044

#### Marketed by :

#### G.K. Publications (P) Ltd.

Plot No. 63, Sector-27A, Near Sector - 28 Metro Station, Faridabad, Haryana-121003

#### ISBN : 978-93-56817-20-3

Printer's Details : Printed in India, New Delhi.

For product information : Visit *www.gkpublications.com* or email to *gkp@gkpublications.com* 

# GATE Electrical Engineering

| Preface                                   | (ix)    |
|---|---------|
| About GATE                                | (xi)    |
| GATE Syllabus                             | (xxi)   |
| <ul> <li>Chapter-Wise Analysis</li> </ul> | (xxiii) |

# Solved Papers (Chapter-Wise)

# Verbal Ability

| 1. | English Grammar          | 1.1 - 1.4  |
|----|--------------------------|------------|
|    | MCQ Type Questions       | 1.1        |
|    | – Answers                | 1.4        |
|    | - Explanations           | 1.4        |
| 2. | Sentence Completion      | 2.1 - 2.6  |
|    | MCQ Type Questions       | 2.1        |
|    | – Answers                | 2.5        |
|    | - Explanations           | 2.5        |
| 3. | Synonyms                 | 3.1 - 3.2  |
|    | MCQ Type Questions       | 3.1        |
|    | – Answers                | 3.2        |
|    | - Explanations           | 3.2        |
| 4. | Antonyms                 | 4.1 - 4.1  |
|    | MCQ Type Questions       | 4.1        |
|    | - Answers                | 4.1        |
|    | - Explanations           | 4.1        |
| 5. | <b>Reasoning Ability</b> | 5.1 - 5.18 |
|    | MCQ Type Questions       | 5.1        |
|    | Numerical Type Questions | 5.12       |
|    | – Answers                | 5.13       |
|    | - Explanations           | 5.13       |

# Numerical Ability

**Contents** 

| 1. | Numbers and Algebra             | 1.1 - 1.14 |
|----|---------------------------------|------------|
|    | MCQ Type Questions              | 1.1        |
|    | Numerical Type Questions        | 1.6        |
|    | – Answers                       | 1.7        |
|    | - Explanations                  | 1.8        |
| 2. | Percentage and Its Applications | 2.1 - 2.4  |
|    | MCQ Type Questions              | 2.1        |
|    | Numerical Type Questions        | 2.2        |
|    | – Answers                       | 2.2        |
|    | - Explanations                  | 2.3        |
| 3. | Time and Work                   | 3.1 - 3.4  |
|    | MCQ Type Questions              | 3.1        |
|    | Numerical Type Questions        | 3.2        |
|    | – Answers                       | 3.2        |
|    | - Explanations                  | 3.3        |
| 4. | Ratio, Proportion and Mixtures  | 4.1 - 4.2  |
|    | MCQ Type Questions              | 4.1        |
|    | – Answers                       | 4.1        |
|    | - Explanations                  | 4.2        |

| 5. | Permutations and | d Combinations & |
|----|------------------|------------------|
|    | Probability      | 5.1 - 5.6        |

|    | •                        |           |
|----|--------------------------|-----------|
|    | MCQ Type Questions       | 5.1       |
|    | Numerical Type Questions | 5.3       |
|    | – Answers                | 5.3       |
|    | - Explanations           | 5.4       |
| 6. | Miscellaneous            | 6.1 - 6.8 |
|    | MCQ Type Questions       | 6.1       |
|    | Numerical Type Questions | 6.4       |
|    | – Answers                | 6.4       |
|    | - Explanations           | 6.5       |

# **Engineering Mathematics**

| 1. | Linear Algebra                    | 1.1 - 1.12 |
|----|-----------------------------------|------------|
|    | MCQ Type Questions                | 1.1        |
|    | Numerical Type Questions          | 1.4        |
|    | – Answers                         | 1.5        |
|    | - Explanations                    | 1.6        |
| 2. | Calculus & Vector Analysis        | 2.1 - 2.10 |
|    | MCQ Type Questions                | 2.1        |
|    | Numerical Type Questions          | 2.3        |
|    | – Answers                         | 2.4        |
|    | - Explanations                    | 2.5        |
| 3. | <b>Differential Equations</b>     | 3.1 - 3.6  |
|    | MCQ Type Questions                | 3.1        |
|    | Numerical Type Questions          | 3.2        |
|    | – Answers                         | 3.3        |
|    | – Explanations                    | 3.3        |
| 4. | <b>Complex Variables</b>          | 4.1 - 4.6  |
|    | MCQ Type Questions                | 4.1        |
|    | – Answers                         | 4.2        |
|    | – Explanations                    | 4.3        |
| 5. | <b>Probability and Statistics</b> | 5.1 - 5.6  |
|    | MCQ Type Questions                | 5.1        |
|    | Numerical Type Questions          | 5.2        |
|    | – Answers                         | 5.3        |
|    | – Explanations                    | 5.3        |
| 6. | Numerical Methods                 | 6.1 - 6.2  |
|    | MCQ Type Questions                | 6.1        |
|    | Numerical Type Questions          | 6.1        |
|    | – Answers                         | 6.1        |
|    | - Explanations                    | 6.1        |

| 7.  | Transform Theory              | 7.1 - 7.2   |
|-----|-------------------------------|-------------|
|     | MCQ Type Questions            | 7.1         |
|     | – Answers                     | 7.1         |
|     | - Explanations                | 7.2         |
| Тес | hnical Section                |             |
| 1.  | Electric Circuits             | 1.1 - 1.66  |
|     | MCQ Type Questions            | 1.1         |
|     | Numerical Type Questions      | 1.21        |
|     | – Answers                     | 1.28        |
|     | - Explanations                | 1.29        |
| 2.  | Electromagnetic Fields        | 2.1 - 2.16  |
|     | MCQ Type Questions            | 2.1         |
|     | Numerical Type Questions      | 2.6         |
|     | – Answers                     | 2.8         |
|     | - Explanations                | 2.8         |
| 3.  | Signals and Systems           | 3.1 - 3.32  |
|     | MCQ Type Questions            | 3.1         |
|     | Numerical Type Questions      | 3.13        |
|     | – Answers                     | 3.14        |
|     | - Explanations                | 3.15        |
| 4.  | <b>Electrical Machines</b>    | 4.1 - 4.78  |
|     | MCQ Type Questions            | 4.1         |
|     | Numerical Type Questions      | 4.28        |
|     | – Answers                     | 4.35        |
|     | – Explanations                | 4.37        |
| 5.  | Power Systems                 | 5.1 - 5.65  |
|     | MCQ Type Questions            | 5.1         |
|     | Numerical Type Questions      | 5.27        |
|     | – Answers                     | 5.32        |
|     | - Explanations                | 5.33        |
| 6.  | Control Systems               | 6.1 - 6.53  |
|     | MCQ Type Questions            | 6.1         |
|     | Numerical Type Questions      | 6.23        |
|     | – Answers                     | 6.25        |
|     | - Explanations                | 6.26        |
| 7.  | Electrical and Electronics Me | easurements |
|     |                               | 7.1 - 7.28  |
|     | MCQ Type Questions            | 7.1         |
|     | Numerical Type Questions      | 7.12        |
|     | – Answers                     | 7.15        |

- Explanations 7.16

| 8. | Analog Circuits   | 8.1 - 8.50                                     |
|----|---|--|
|    | MCQ Type Questions  | 8.1  |
|    | Numerical Type Questions  | 8.21   |
|    | – Answers   | 8.25   |
|    | - Explanations  | 8.26   |
|    |   |  |
| 9. | <b>Digital Circuits</b>   | 9.1 - 9.20                                     |
| 9. | <b>Digital Circuits</b><br>MCQ Type Questions   | <b>9.1 - 9.20</b><br>9.1                       |
| 9. | <b>Digital Circuits</b><br>MCQ Type Questions<br>Numerical Type Questions                         | <b>9.1 - 9.20</b><br>9.1<br>9.9                |
| 9. | <b>Digital Circuits</b><br>MCQ Type Questions<br>Numerical Type Questions<br>– Answers            | <b>9.1 - 9.20</b><br>9.1<br>9.9<br>9.9         |
| 9. | Digital Circuits<br>MCQ Type Questions<br>Numerical Type Questions<br>– Answers<br>– Explanations | <b>9.1 - 9.20</b><br>9.1<br>9.9<br>9.9<br>9.10 |

| 10. | 8085 Microprocessor          | 10.1 - 10.6  |
|-----|------------------------------|--------------|
|     | MCQ Type Questions           | 10.1         |
|     | Numerical Type Questions     | 10.4         |
|     | – Answers                    | 10.4         |
|     | - Explanations               | 10.4         |
| 11. | Power Electronics and Drives | 11.1 - 11.50 |
|     | MCQ Type Questions           | 11.1         |
|     | Numerical Type Questions     | 11.19        |
|     | – Answers                    | 11.25        |
|     | - Explanations               | 11.26        |
| •   | Solved Paper 2024            | 1 - 29       |



# **MCQ TYPE QUESTIONS**

# 1992

**1.** A unity feedback system has the open loop transfer function

$$G(s) = \frac{1}{(s-1)(s+2)(s+3)}$$

The Nyquist plot of G encircles the origin

- (a) Never (b) Once
- (c) Twice (d) Thrice
- **2.** The Nyquist plot encloses the origin only once from the above figure. Hence choice B is correct. The overall transfer function of the system in Figure, is



**3.** Which of the following figure(s) represent valid root loci in the s-plane for positive K? Assume that the system has a transfer function with real co-efficients.





4. For what values of 'a' does the system shown in Figure have a zero steady state error

**Control Systems** 

[i.e.,  $\lim \epsilon([t])$  for a step input?



- (d) for no value of 'a'
- 5. Match the following transfer functions and impulse responses

Transfer functions Impulse Responses



6.2 Control Systems

# 1994

**6.** The matrix of any state–space equations for the transfer function  $c(s)/\mathbf{R}(s)$  of the system, shown below in Figure. is



7. The pole-zero configuration of a phase-lead compensator is given by



8. Match the polar plots for the following functions on the left hand side



# 1995

**9.** The closed-loop transfer function of a control system is given by

$$\frac{C(s)}{R(s)} = \frac{2(s-1)}{(s+2)(s+1)}$$

For a unit step input the output is

(a)  $-3 e^{-2t} + 4 e^{-t} - 1 (b) -3 e^{-2t} - 4 e^{-t} + 1$ 

(c) zero (d) infinity

10. A system is described by the state equation XX = AX + BU.

The output is given by Y = C X

where 
$$A = \begin{bmatrix} -4 & -1 \\ 3 & -1 \end{bmatrix}$$
  $B = \begin{bmatrix} 1 \\ 1 \end{bmatrix}$   $C = [1, 0].$ 

Transfer function  $G(\boldsymbol{s})$  of the system is

(a) 
$$\frac{s}{s^2 + 5s + 7}$$
 (b)  $\frac{1}{s^2 + 5s + 7}$   
(c)  $\frac{s}{s^2 + 3s + 2}$  (d)  $\frac{1}{s^2 + 3s + 2}$ 

# 1996

**11.** The unit-impulse response of a unit-feedback control system is given by

$$c(t) = -te^{-t} + 2 e^{-t}, (t \ge 0)$$

the open loop transfer function is equal to

(a) 
$$\frac{s+1}{(s+2)^2}$$
 (b)  $\frac{2s+1}{s^2}$   
(c)  $\frac{s+1}{(s+1)^2}$  (d)  $\frac{s+1}{s^2}$ 

12. Consider the unit-step response of a unityfeedback control system whose open-loop transfer functions is  $G(s) = \frac{1}{s(s+1)}$ . The

maximum overshoot is equal to

- (a) 0.143 (b) 0.153
- $(c) \ 0.163 \qquad \qquad (d) \ 0.173$
- **13.** For a feedback control system of type 2, the steady state error for a ramp input is
  - (a) infinite (b) constant
  - (c) zero (d) indeterminate
- **14.** The closed-loop transfer function of a control system is given by  $\frac{C(s)}{R(S)} = \frac{1}{1+s}$ . For the input  $r(t) = \sin t$ , the steady state value of c(t) is equal to 1

(a) 
$$\frac{1}{\sqrt{2}} \cos t$$
 (b) 1  
(c)  $\frac{1}{\sqrt{2}} \sin t$  (d)  $\frac{1}{\sqrt{2}} \sin \left(1 - \frac{\pi}{4}\right)$ 

15. For the system shown in Figure, with a damping ratio  $\zeta$  of 0.7 and an undamped natural frequency  $\omega_n$  of 4 rad/sec, the values of K and a are



- (a) K = 4, a = 0.35 (b) K = 8, a = 0.455
- (c) K = 16, a = 0.225 (d) K = 64, a = 0.9
- **16.** The unit impulse response of a system is given as  $c(t) = -4 e^{-t} + 6 e^{-2t}$ . The step response of the same system for  $t \ge 0$  is equal to
  - $(a) 3e^{-2t} 4e^{-t} + 1$
  - (b) 3e  $^{-2t}$  + 4e  $^{-t}$  1

$$(c) - 3e^{-2t} - 4e^{-t} - 1$$

(d)  $3e^{-2t} + 4e^{-t} - 1$ 

## 1997

- **17.** Introduction of integral action in the forward path of a unity feedback system results in a
  - (a) marginally stable system
  - (b) system with no steady state error
  - $(c)\;$  system with increased stability margin
  - (d) system with better speed of response

# 1998

**18.** The output of a linear time invariant control systme is c(t) for a certain input r(t). If r(t) is modified by passing it through a block whose transfer function is  $e^{-s}$  and then applied to the systme, the modified output of the system would be

(a) 
$$\frac{c(t)}{1+e^t}$$
 (b)  $\frac{c(t)}{1+e^{-t}}$   
(c)  $c(t-1)u(t-1)$  (d)  $c(t)u(t-1)$ 

- **19.** None of the poles of a lonear control system lie in the right half of s-plane. For a bounded input, the output of this system
  - (a) is always bounded
  - (b) could be unbounded
  - (c) always tends to zero
  - $\left( d\right)$  none of the above
- 20. The phase lead compensation is used to
  - (a) increase rise time and decrease overshoot
  - (b) decrease both rise time and overshoot
  - $(c) \,$  increase both rise time and overshoot
  - (d) decrease rise time and increase overshoot

- 21. A set of linear equations is represented by the matrix equation Ax = b. The necessary condition for the existence of a solution for this system is(a) A must be invertible
  - (b) b must be linearly depended on the columns
  - (c) b mut be linearly independent of the columns of A
  - (d) none of the above

of A

- **22.** The vector  $\begin{bmatrix} 1\\ 2\\ -1 \end{bmatrix}$  is an eigen vector of A
  - $= \begin{bmatrix} -2 & 2 & -3 \\ 2 & 1 & -6 \\ -1 & -2 & 0 \end{bmatrix}.$  One of the eigen values of A is

(*b*) 2

c) 5 
$$(d) -1$$

23.  $A = \begin{bmatrix} 2 & 0 & 0 & -1 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 3 & 0 \\ -1 & 0 & 0 & 4 \end{bmatrix}$ . The sum of the eigen values of the matrix A is (a) 10 (b) = 10

**24.** For block diagram shown in Figure C(s)/R(s) is given by



25. The number of roots on the equation

 $2S^4 + S^3 + 3S^2 + 5S + 7 = 0$  that lie in the right half of S plane is ;

- (a) Zero
- (b) One
- (c) Two
- (d) Three

6.4 Control Systems

26. 
$$A = \begin{bmatrix} 5 & 0 & 2 \\ 0 & 3 & 0 \\ 2 & 0 & 1 \end{bmatrix}$$
. The inverse of A is  
(a) 
$$\begin{bmatrix} 1 & 0 & -2 \\ 0 & \frac{1}{3} & 0 \\ -2 & 0 & 5 \end{bmatrix}$$
(b) 
$$\begin{bmatrix} 5 & 0 & 2 \\ 0 & -\frac{1}{3} & 0 \\ 2 & 0 & 1 \end{bmatrix}$$
(c) 
$$\begin{bmatrix} \frac{1}{5} & 0 & \frac{1}{2} \\ 0 & \frac{1}{3} & 0 \\ \frac{1}{2} & 0 & 1 \end{bmatrix}$$
(d) 
$$\begin{bmatrix} \frac{1}{5} & 0 & -\frac{1}{2} \\ 0 & \frac{1}{3} & 0 \\ -\frac{1}{2} & 0 & 1 \end{bmatrix}$$

1999

**27.** The function corresponding to the Bode plot of Figure, is



(a) 
$$A = j f/f_1$$
 (b)  $A = 1/(1 - j f_1/f)$ 

(c) 
$$A = 1/(1 + jf_1/f)$$
 (d)  $A = 1 + j f/f_1$ 

# 2000

28. Feedback control systems are

- (a) insensitive to both forward-and feedback-path parameter changes
- (b) less sensitive to feedback-path parameter changes than to forward-path parameter changes
- (c) less sensitive to forward-path parameter changes than to feedback-path parameter changes
- (d) equally sensitive to forward-and feed back- path parameter changes
- **29.** A unity feedback system has open-loop transfer function G(s). The steady-state error is zero for
  - (a) step input and type-1G(s)
  - (b) ramp input and type-1G(s)
  - (c) step input and type-G(s)
  - (d) ramp input and type-0 G(s)
- **30.** A linear time-invariant system initially at rest, when subjected to a unit-step input, gives a response  $y(t) = te^{-t}$ , t > 0. The transfer function of the system is

(a) 
$$\frac{1}{(s+1)^2}$$
 (b)  $\frac{1}{s(s+1)^2}$   
(c)  $\frac{s}{(s+1)^2}$  (d)  $\frac{1}{s(s+1)}$ 

**31.** The characteristic equation of a feedback control system is

$$2s^4 + s^3 + 3s^2 + 5s + 10 = 0$$

The number of roots in the right half of s-plane are

- (a) zero (b) 1 (c) 2 (d) 3
- **32.** A unity feedback system has open-loop transfer function  $G(s) = \frac{25}{s(s+6)}$ .

The peak overshoot in the step-input response of the system is approximately equal to

- (a) 5% (b) 10%
- (c) 15% (d) 20%
- **33.** Maximum phase-lead of the compensator  $\begin{pmatrix} 0 & 5a \\ -1 \end{pmatrix}$

$$D(s) = \frac{(0.5s + 1)}{(0.05s + 1)}$$
, is

- (a) 52 deg at 4 rad/sec (b) 52 deg at 10 rad/sec
- (c)  $55 \deg \operatorname{at} 12 \operatorname{rad/sec} (d)$  None of these
- **34.** Open-loop transfer function of a unity-feedback system is

G(s) = G<sub>1</sub>(s). 
$$e^{-r\tau D} = \frac{e^{-s\tau_D}}{s(s+1)(s+2)}$$

Given :  $|G_1(j\omega)| \approx 1$  when  $\omega = 0.466$ .

What is the phase margin when  $\tau_D = 0$ ?

- (a)  $51.9^{\circ}$  (b)  $61.9^{\circ}$
- (c)  $41.9^{\circ}$  (d)  $71.9^{\circ}$
- **35.** A unity feedback system has open-loop transfer function

$$G(s) = {K(s+5)\over s(s+2)}$$
;  $K \ge 0$ 

What is the value of K(if it exists) so that the damping  $\xi$  of the complex closed loop poles is 0.3?

- (*a*) 1/2 (*b*) 5
- (c) 2 (d) does not exist

2001

**36.** The polar plot of a type-1, 3-pole, open-loop system is shown in the figure given below. The closed-loop system is



- (a) always stable
- (b) marginally stable
- (c) unstable with one pole on the right half s-plane
- (d) unstable with two poles on the right half s-plane
- 37. Given the homogeneous state-space equation

$$\dot{x} = \begin{bmatrix} -3 & 1\\ 0 & -2 \end{bmatrix} x$$

The steady state value of  $x_{ss} = \lim_{t\to\infty} x(t)$ , given the initial state value of  $x(0) = [10 - 10]^{T}$ , is

(a) 
$$x_{ss} = \begin{bmatrix} 0 \\ 0 \end{bmatrix}$$
 (b)  $x_{ss} = \begin{bmatrix} -3 \\ -2 \end{bmatrix}$   
(c)  $x_{ss} = \begin{bmatrix} -10 \\ 10 \end{bmatrix}$  (d)  $x_{ss} = \begin{bmatrix} \infty \\ \infty \end{bmatrix}$ 

- 38. The conductors of a 10 km long, single phase, two wire line are separated by a distance of 1.5 m. The diameter of each conductor is 1 cm. If the conductors are of copper, then inductance of the circuit is
  - (a) 50.0 mH (b) 45.3 mH
  - (c) 23.8 mH (d) 19.6 mH
- **39.** The asymptotic approximation of the logmagnitude versus frequency plot of a minimum phase system with real poles and one zero is shown in the figure given below. Its transfer functions is



#### <u>Common Data Q. (40 – 42)</u>

A unity feedback system has an open-loop transfer function of

$$G(s) = \frac{10000}{s(s+10)^2}$$

**40.** What is the magnitude of  $G(j\omega)$  in *d*B at an angular frequency of  $\omega = 20$  rad/sec?

(a) 
$$0 \, dB$$
 (b)  $10 \, dB$ 

 $(c) \quad 20 \text{ dB} \qquad \qquad (d) \text{ none of these}$ 

- 41. What is the phase margin in degrees?
  - (a)  $36.86^{\circ}$  (b)  $56.86^{\circ}$
  - (c)  $-36.86^{\circ}$  (d) none of these
- **42.** What is the gain margin in dB?
  - (a)  $-13.97 \,\mathrm{dB}$  (b)  $13.97 \,\mathrm{dB}$

(c)  $43.97 \,\mathrm{dB}$  (d)  $-43.97 \,\mathrm{dB}$ 

# 2002

**43.** Let s(t) be the step response of a linear system with zero initial conditions; then the response of this system to an input u(t) is

(a) 
$$\int_{0}^{t} s(t-\tau)u(\tau)d\tau$$
  
(b) 
$$\frac{d}{dt} \left[ \int_{0}^{t} s(t-\tau)u(\tau)d\tau \right]$$
  
(c) 
$$\int_{0}^{t} s(t-\tau) \left[ \int_{0}^{t} u(\tau_{1})d\tau_{1} \right] d\tau$$
  
(d) 
$$\int_{0}^{1} s(t-\tau)^{2} u(\tau) d\tau$$

**44.** Let Y(s) be the Laplace transformation of the function y(t), then final value of the function is

(a) 
$$\lim_{s \to 0} Y(s)$$
 (b)  $\lim_{s \to \infty} Y(s)$ 

(c) 
$$\lim_{s \to 0} sY(s)$$
 (d)  $\lim_{s \to \infty} sY(s)$ 

45. The determinant of the matrix

|     | 1   | 0   | 0   | 0          |     |
|-----|-----|-----|-----|------------|-----|
|     | 100 | 1   | 0   | 0          |     |
|     | 100 | 200 | 1   | 0          | is  |
|     | 100 | 200 | 300 | 1          |     |
| 100 |     |     |     | <i>(b)</i> | 200 |
| 1   |     |     |     | (d)        | 300 |

- 46. The state transition matrix for the system
  - $\dot{\chi}$  = AX with initial state X(0) is
  - (a)  $(sI A)^{-1}$
  - (b)  $e^{At} X(0)$

(a)(c)

- (c) Laplace inverse of  $[(sI A)^{-1}]$
- (d) Laplace inverse of  $[(sI A)^{-1}X(0)]$
- **47.** For the system  $\dot{\chi} = \begin{bmatrix} 2 & 3 \\ 0 & 5 \end{bmatrix} X + \begin{bmatrix} 1 \\ 0 \end{bmatrix} u$ , which of the following statements is true ?
  - (a) The system is controllable but unstable
  - (b) The system is uncontrollable and unstable
  - (c) The system is controllable and stable
  - (d) The system is uncontrollable and stable

#### 6.6 Control Systems

48. A unity feedback system has an open loop transfer

function,  $G(s) = \frac{K}{s^2}$ . The root locus plot is



49. The transfer function of the system described by

 $\frac{d^2y}{dt^2} + \frac{dy}{dt} = \frac{du}{dt} + 2u$  with u as input and y as output is

(a) 
$$\frac{(s+2)}{(s^2+s)}$$
 (b)  $\frac{(s+1)}{(s^2+s)}$   
(c)  $\frac{2}{(s^2+s)}$  (d)  $\frac{2s}{(s^2+s)}$ 

**50.** For the system  $\dot{\chi} = \begin{bmatrix} 2 & 0 \\ 0 & 4 \end{bmatrix} X + \begin{bmatrix} 1 \\ 1 \end{bmatrix} u$ ;  $y = \begin{bmatrix} 4 & 0 \end{bmatrix} X$ , with u as unit impulse and with zero initial state,

- the output, y, becomes
- (a)  $2 e^{2t}$  (b)  $4 e^{2t}$ (c)  $2 e^{4t}$  (d)  $4 e^{4t}$
- 51. The eigen values of the system represented by

$$\dot{\mathbf{X}} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & 0 & 0 & 1 \end{bmatrix} \mathbf{X} \text{ are}$$
  
a) 0, 0, 0, 0, 0 (b) 1, 1, 1, 1  
c) 0, 0, 0, -1 (d) 1, 0, 0,

## 2003

(

**52.** A control system is defined by the following mathematical relationship

$$\frac{d^2x}{dt^2} + 6\frac{dx}{dt} + 5x = 12(1 - e^{-2t})$$

The response of the system as  $t \to \infty$  is

(a) x = 6 (b) x = 2

(c) x = 2.4 (d) x = -2

**53.** A lead compensator used for a closed loop controller has the following transfer function

$$\frac{\mathrm{K}\left(1+\frac{s}{a}\right)}{\left(1+\frac{s}{b}\right)}$$

For such a lead compensator

| (a)          | a < b             | <i>(b)</i> | b < a  |
|--------------|-------------------|------------|--------|
| ( <i>c</i> ) | $a > \mathrm{Kb}$ | (d)        | a < Kb |

54. A second order system starts with an initial

condition of  $\begin{bmatrix} 2\\ 3 \end{bmatrix}$  without any external input. The

state transition matrix for the system is given  $\begin{bmatrix} e^{-2t} & 0 \end{bmatrix}$ 

by  $\begin{bmatrix} t & 0 \\ 0 & e^{-t} \end{bmatrix}$ . The state of the system at the end of 1 second is given by

|     | 0.271                            |     | 0.135   |  |
|-----|----------------------------------|-----|---------|--|
| (a) | 1.100                            | (b) | 0.368   |  |
| (c) | $\left\lceil 0.271 \right\rceil$ |     | [0.135] |  |
|     | 0.736                            | (d) | 1.100   |  |

**55.** A control system with certain excitation is governed by the following mathematical equation

$$\frac{d^2x}{dt^2} + \frac{1}{2}\frac{dx}{dt} + \frac{1}{18}x = 10 + 5e^{-4t} + 2e^{-5t}$$

The natural time constants of the response of the system are

- (a) 2s and 5s (b) 3s and 6s
- (c) 4s and 5s (d) 1/3s and 1/6s

#### Common Data Q. (56 – 57)

The block diagram shown in the figure given below gives a unity feedback closed loop control system.



**56.** The steady state error in the response of the above system to unit step input is

| (a)          | 25~% | <i>(b)</i> | 0.75~% |
|--------------|------|------------|--------|
| ( <i>c</i> ) | 6 %  | (d)        | 33~%   |

- **57.** The roots of the closed loop characteristic equation of the system are
  - (a) -1 and -15 (b) 6 and 10

(c) 
$$-4$$
 and  $-15$  (d)  $-6$  and  $-10$ 

**58.** The following equation defines a separately excited dc motor in the form of a differential equation

$$\frac{d^2\boldsymbol{\omega}}{dt^2} + \frac{B}{J}\frac{d\boldsymbol{\omega}}{dt} + \frac{K^2}{LJ}\boldsymbol{\omega} = \frac{K}{LJ}V_a$$

The above equation may be organized in the state-space form as follows

$$\begin{bmatrix} \frac{d^2 \omega}{dt^2} \\ \frac{d \omega}{dt} \end{bmatrix} = \mathbf{P} \begin{bmatrix} \frac{d \omega}{dt} \\ \omega \end{bmatrix} + \mathbf{Q} \mathbf{V}_a$$

where P matrix is given by

(a) 
$$\begin{bmatrix} -\frac{B}{J} - \frac{K^2}{LJ} \\ 1 & 0 \end{bmatrix}$$
  
(b) 
$$\begin{bmatrix} -\frac{K^2}{LJ} - \frac{B}{J} \\ 0 & 1 \end{bmatrix}$$
  
(c) 
$$\begin{bmatrix} 0 & 1 \\ -\frac{K^2}{LJ} - \frac{B}{J} \end{bmatrix}$$
  
(d) 
$$\begin{bmatrix} 1 & 0 \\ -\frac{B}{J} - \frac{K^2}{LJ} \end{bmatrix}$$

59. The loop gain GH of a closed loop system is given

by the expression 
$$\frac{K}{s(s+2)(s+4)}$$

The value of K for which the system just becomes unstable is

....

- (a) K = 6 (b) K = 8
- (c) K = 48 (d) K = 96
- ${\bf 60.}\,$  The asymptotic Bode plot of the transfer function

 $\frac{K}{1+\frac{s}{2}}$  is given in the figure given below. The

error in phase angle and dB gain at a frequency of  $\omega = 0.5$  a are respectively



**61.** The block diagram of a control system is shown in the figure given below. The transfer function G(s) = Y(s)/U(s) of the system is



#### **2004**

- **62.** The Nyquist plot of loop transfer function G(s) H(s) of a closed loop control system passes through the point (-1, j0) in the G(s) H(s) plane. The phase margin of the system is (a)  $0^{\circ}$  (b)  $45^{\circ}$ 
  - (c) 90° (d) 180°
- **63.** Consider the function,  $F(s) = \frac{5}{s(s^2 + 3s + 2)}$ where F(s) is Laplace transform of the function f(t).

The initial value of f(t) is equal to

- (a) 5 (b)  $\frac{5}{2}$ (c)  $\frac{5}{3}$  (d) 0
- 64. For a tachometer, if  $\theta(t)$  is the rotor displacement is radians, e(t) is the output voltage and  $K_t$  is the tachometer constant in V/rad/sec, then the transfer function E(s) will be

(a) 
$$K_t s^2$$
 (b)  $\frac{K_t}{s}$   
(c)  $K_t s$  (d)  $K_t$ 

**65.** For the equation,  $s^3 - 4s^2 + s + 6 = 0$ 

the number of roots in the left half of s-plane will be

- (a) zero (b) one
- (c) two (d) three
- 66. For the block diagram shown in the figure given

below the transfer function 
$$\frac{C(s)}{R(s)}$$
 is equal to



**67.** The state variable description of a linear autonomous system is,

$$X = AX,$$

where X is two dimensional state vector, and

A is the system matrix given by  $A = \begin{bmatrix} 0 & 2 \\ 2 & 0 \end{bmatrix}$ . The roots of the characteristic equation are

(a) -2 and +2 (b) -j2 and +j2(c) -2 and -2 (d) +2 and +2

#### 6.8 Control Systems

**68.** The block diagram of a closed loop control system is given in the figure given below. The values of K and P such that the system has a damping ratio of 0.7 and an undamped natural frequency  $\omega_{p}$  of 5 rad/sec, are respectively equal to



- (c) 25 and 0.3 (d) 25 and 0.2
- **69.** The unit impulse response of a second order under-damped system starting from rest is given by

$$c(t) = 12.5 e^{-6t} \sin 8 t, \quad t \ge 0$$

The steady-state value of the unit step response of the system is equal to

**70.** In the system shown in the figure given below the input is  $x(t) = \sin t$ .

In the steady-state, the response y(t) will be

$$(a) \quad \frac{1}{\sqrt{2}} \sin(t - 45^{\circ}) \quad (b) \quad \frac{1}{\sqrt{2}} \sin(t + 45^{\circ})$$

$$(c) \quad \sin(t - 45^{\circ}) \quad (d) \quad \sin(t + 45^{\circ})$$

**71.** The open loop transfer function of a unity feedback

control system is given as  $G(s) = \frac{as+1}{s^2}$ . The value

of 'a' to give a phase margin of  $45^\circ$  is equal to

- (a) 0.141 (b) 0.441
- (c) 0.841 (d) 1.141

#### 2005

**72.** A system with zero initial conditions has the closed loop transfer function.

$$T(s) = \frac{s^2 + 4}{(s+1)(s+4)}$$

The system output is zero at the frequency  $(a) \quad 0.5 \text{ rad/sec}$ 

- (b) 1 rad/sec
- (c)  $2 \operatorname{rad/sec}$
- (d)  $4 \operatorname{rad/sec}$
- **73.** Figure given below shows the root locus plot (location of poles not given) of a third order system whose open loop transfer function is

(a) 
$$\frac{K}{s^3}$$
  
(b)  $\frac{K}{s^2(s+1)}$   
(c)  $\frac{K}{s(s^2+1)}$   
(d)  $\frac{K}{s(s^2-1)}$ 

**74.** The gain margin of a unity feedback control system with the open loop transfer function

$$G(s) = \frac{(s+1)}{s^2} \text{ is}$$
(a) 0
(b)  $\frac{1}{\sqrt{2}}$ 
(c)  $\sqrt{2}$ 
(d)  $\infty$ 

75. A unity feedback system, having an open loop

gain 
$$G(s)H(s) = \frac{K(1-s)}{(1+s)}$$
, becomes stable when

- (a)  $|\mathbf{K}| > 1$  (b)  $\mathbf{K} > 1$
- (c)  $|\mathbf{K}| < 1$  (d)  $\mathbf{K} < -1$
- **76.** When subjected to a unit step input, the closed loop control system shown in the figure given below will have a steady state error of



**77.** In the GH(s) plane, the Nyquist plot of the loop

transfer function  $G(s)H(s) = \frac{\pi e^{-0.25}}{s}$  passes through the negative real axis at the point

(a) 
$$(-0.25, j0)$$
 (b)  $(-0.5, j0)$ 

- (c) (-1, j0) (d) (-2, j0)
- **78.** If the compensated system shown in the figure given below has a phase margin of 60° at the crossover frequency of 1 rad/sec, then value of the gain K is



**79.** For the matrix  $p = \begin{bmatrix} 3 & -2 & 2 \\ 0 & -2 & 1 \\ 0 & 0 & 1 \end{bmatrix}$ , one of the eigen values is equal to -2. Which of the following is an

eigen vector?  $(a) \begin{vmatrix} -2 \\ -2 \\ 1 \end{vmatrix}$ 2 -1 *(b)*  $(d)\begin{bmatrix} 2\\5\end{bmatrix}$ (*c*) –2 3

- **80.** If u(t) is the unit step and  $\delta(t)$  is the unit impulse function, the inverse *z*-transform of  $F(z) = \frac{1}{z+1}$ 
  - for k > 0 is
  - (a)  $(-1)^k \, \delta(k)$ (b)  $\delta(k) - (-1)^k$ (c)  $(-1)^k u(k)$ (d)  $u(k) - (-1)^k$

#### Linked Answer Q. (81 - 82)

A state variable system

$$\mathbf{X}(t) = \begin{bmatrix} 0 & 1 \\ 0 & -3 \end{bmatrix} \mathbf{X}(t) + \begin{bmatrix} 1 \\ 0 \end{bmatrix} u(t),$$

with initial condition  $X(0) \begin{bmatrix} -1 & 3 \end{bmatrix}^T$  and the unit step input u(t) has

81. The state transition matrix

(a) 
$$\begin{bmatrix} 1 & \frac{1}{3}(1-e^{-3t}) \\ 0 & e^{-3t} \end{bmatrix}$$
 (b)  $\begin{bmatrix} 1 & \frac{1}{3}(e^{-t}-e^{-3t}) \\ 0 & e^{-t} \end{bmatrix}$   
(c)  $\begin{bmatrix} 1 & \frac{1}{3}(e^{-t}-e^{-3t}) \\ 0 & e^{-3t} \end{bmatrix}$  (d)  $\begin{bmatrix} 1 & (1-e^{-t}) \\ 0 & e^{-t} \end{bmatrix}$ 

82. The state transition equation

(a) 
$$X(t) = \begin{bmatrix} t - e^{-t} \\ e^{-t} \end{bmatrix}$$
  
(b) 
$$X(t) = \begin{bmatrix} t - e^{-3t} \\ e^{-t} \end{bmatrix}$$
  
(c) 
$$X(t) = \begin{bmatrix} t - e^{-3t} \\ 3e^{-3t} \end{bmatrix}$$
  
(d) 
$$X(t) = \begin{bmatrix} t - e^{-3t} \\ e^{-t} \end{bmatrix}$$

## 2006

(

83. For a system with the transfer function H (s) =  $\frac{3(s-2)}{4s^2-2s+1}$ , the matrix A in the state space form x = Ax + Bu is equal to

$$(a) \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ -1 & 2 & -4 \end{bmatrix} (b) \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -1 & 2 & -4 \end{bmatrix}$$
$$(c) \begin{bmatrix} 0 & 1 & 0 \\ 3 & -2 & 1 \\ 1 & -2 & 4 \end{bmatrix} (d) \begin{bmatrix} 1 & 0 & 0 \\ 0 & 0 & 1 \\ -1 & 2 & -4 \end{bmatrix}$$

- 84. A discrete real all pass system has a pole at  $z = 2 \angle 30^\circ$ : it, therefore
  - (a) also has a pole at  $\frac{1}{2} \angle 30^{\circ}$
  - (b) has a constant phase response over the *z*plane: arg |H(z)| = constant
  - (c) is stable only, if it is anticausal
  - (d) has a constant phase response over the unit circle: arg  $|H(e^{j\Omega})| = constant$

#### 85. The Bode magnitude plot of



6.10 Control Systems

**86.** A closed-loop system has the characteristic function  $(s^2 - 4) (s + 1) + K (s - 1) = 0$ . Its root locus plot against K is



87. The algebraic equation

$$F(s) = s^5 - 3s^4 + 5s^3 - 7s^2 + 4s + 20$$
 is given.

 $\mathbf{F}(s) = 0$  has

- (a) a single complex root with the remaining roots being real
- (b) one positive real root and four complex roots, all with positive real parts
- (c) one negative real root, two imaginary roots, and two roots with positive real parts
- (d) one positive real root, two imaginary roots, and two roots with negative real parts
- **88.** Consider the following Nyquist plots of loop transfer functions over  $\omega = 0$  to  $\omega = \infty$ . Which of these plots represents a stable closed loop system ?





# 2007

**89.** The system shown in the figure given below is



(a) stable

(b) unstable

(c) conditionally stable

(*d*) stable for input  $u_1$ , but unstable for input  $u_2$ **90.** If  $x = \text{Re } G(j\theta)$ , and  $y = \text{Im } G(j\omega)$  then for  $\omega \to 0^+$ ,

the Nyquist plot for  $G(s) = \frac{1}{s(s+1)(s+2)}$  becomes asymptotic to the line

(a) 
$$x = 0$$
  
(b)  $x = -\frac{3}{4}$   
(c)  $x = y - \frac{1}{6}$   
(d)  $x = \frac{y}{\sqrt{3}}$ 

**91.** The system  $\frac{900}{(s+1)(s+9)}$  is be compensated such that its gain-crossover frequency becomes same

as its uncompensated phase-crossover frequency and provides a 45° phase margin. To achieve this, one may use

- (a) a lag compensator that provides an attenuation of 20 dB and a phase lag of  $45^{\circ}$  at the frequency of  $3\sqrt{3}$  rad/s
- (b) a lead compensator that provides an amplification of 20 dB and a phase lead of  $45^{\circ}$  at the frequency of 3 rad/s
- (c) a lag-lead compensator that provides an amplification of 20 dB and a phase lag of  $45^{\circ}$  at the frequency of  $\sqrt{3}$  rad/s.
- (d) a lag-lead compensator that provides an attenuation of 20 dB and phase lead of  $45^{\circ}$  at the frequency of 3 rad/s

**92.** Consider the discrete-time system shown in the figure where the impulse response of G(z) is g(0) = 0, g(1) = g(2) = 1, g(3) = g(4) = ... = 0.



This system is stable for range of values of K

(a) 
$$\left\lfloor -1, \frac{1}{2} \right\rfloor$$
 (b)  $\left[-1, 1\right]$   
(c)  $\left[-\frac{1}{2}, 1\right]$  (d)  $\left[-\frac{1}{2}, 2\right]$ 

**93.** If the loop gain *K* of a negative feedback system having a loop transfer function  $\frac{K(s+3)}{(s+8)^2}$  is to be

adjusted to induce a sustained oscillation then

- (a) The frequency of this oscillation must be  $\frac{4}{\sqrt{3}}$  rad/s
- (b) The frequency of this oscillation must be must be 4 rad/s
- (c) The frequency of this oscillation must be must be 4 or  $\frac{4}{\sqrt{3}}$  rad/s
- (d) such a K does not exist
- 94. The system shown in the figure below



can be reduced to the form



with

(a) 
$$X = c_0 s + c_1$$
,  $Y = \frac{1}{(s^2 + a_0 s + a_1)}$ ,  $Z = b_0 s + b_1$ 

(b) 
$$X = 1$$
,  $Y = \frac{(c_0 s + c_1)}{(s^2 + a_0 s + a_1)}$ ,  $Z = b_0 s + b_1$ 

(c) 
$$X = c_1 s + c_0$$
,  $Y = \frac{(b_1 s + b_0)}{(s^2 + a_1 s + a_0)}$ ,  $Z = 1$   
(d)  $X = c_1 s + c_0$ ,  $Y = \frac{1}{(s^2 + a_1 s + a_0)}$ ,  $Z = b_1 s + b_0$ 

**95.** The R-L-C series circuit shown is supplied from a variable frequency voltage source. The admittance - locus of the R-L-C network at terminals AB for increasing frequency  $\omega$  is



96. Consider the feedback control system shown below which is subjected to a unit step input. The system is stable and has the following parameters  $k_p = 4$ ,  $k_i = 10 \omega = 500$  and  $\xi = 0.7$ 



The steady state value of z is

| ( <i>a</i> ) 1 | ( <i>b</i> ) 0.25 |
|----------------|-------------------|
| (c) <b>0.1</b> | ( <i>d</i> ) 0    |

#### 2008

**97.** The characteristic equation of a  $(3 \times 3)$  matrix P is defined as

 $\alpha \left( \lambda \right) = \left| \lambda I - P \right| = \lambda^3 + \lambda^2 + 2\lambda + 1 = 0.$ 

If I denotes identity matrix, then the inverse of matrix P will be

- (a)  $(P^2 + P + 2I)$
- (b)  $(P^2 + P + I)$
- (c)  $-(P^2 + P + I)$
- $(d) (P^2 + P + 2I)$

#### 6.12 Control Systems

**98.** The transfer function of a linear time invariant system is given as

$$G(s) = \frac{1}{s^2 + 3s + 2}$$

The steady state value of the output of this system for a unit impulse input applied at time instant t = 1 will be

 $\mathbf{2}$ 

(a) 
$$0$$
 (b)  $0.5$ 

(c) 1 
$$(d)$$

99. The transfer functions of two compensators are

$$C_1 = \frac{10 (s + 1)}{(s + 10)}, C_2 = \frac{s + 10}{10 (s + 1)}$$

Which one of the following statements is correct?

- (a)  $C_1$  is a lead compensator and  $C_2$  is a lag compensator
- (b)  $C_1$  is a lag compensator and  $C_2$  is a lead compensator
- (c) Both  $C_1$  and  $C_2$  are lead compensators
- (d) Both  $C_1$  and  $C_2$  are lag compensators
- **100.** The asymptotic Bode magnitude plot of a minimum phase transfer function is shown in the figure



This transfer function has

- (a) Three poles and one zero
- (b) Two poles and one zero
- (c) Two poles and two zeros
- (d) One pole and two zeros
- 101. Figure shows a feedback system where K > 0. The range of K for which the system is stable will be given by



102. The transfer function of a system is given as

$$\frac{100}{s^2 + 20s + 100}$$

This system is

- (a) an overdamped system
- (b) an underdamped system
- (c) a critically damped system
- (d) an unstable system

# 2009

**103.** The measurement system shown in the figure uses three sub-systems in cascade whose gains

are specified as  $G_{1,}G_{2}$  and  $\frac{1}{G_{3}}$ . The relative small errors associated with each respective subsystem  $G_{1,}G_{2}$  and  $G_{3}$  are  $\varepsilon_{1}$ ,  $\varepsilon_{2}$  and  $\varepsilon_{3}$ . The error associated with the output is



**104.** The polar plot of an open loop stable system is shown below. The closed loop system is



- (a) always stable
- (b) marginally stable
- (c) unstable with one pole on the RH s-plane
- (*d*) unstable with two poles on the RH s-plane
- **105.** The first two rows of Routh's tabulation of a third order equation are as follows.

$${f s}^3 \ 2 \ 2 \ {f s}^3 \ 4 \ 4$$

This means there are

- (a) two roots at  $s = \pm j$  and one root in right half s-plane
- (b) two roots at  $s = \pm j2$  and one root in left half s-plane
- (c) two roots at  $s = \pm j2$  and one root in right half s-plane
- (d) two roots at  $s = \pm j$  and one root in left half s-plane

**106.** The asymptotic approximation of the logmagnitude *vs* frequency plot of a system containing only real poles and zeros is shown in the figure given below. Its transfer function is



**107.** The unit-step response of a unity feedback system with open loop transfer function



The value of *K* is

**108.** The open loop transfer function of a unity feedback system is given by

$$\mathbf{G}(s) = \frac{(e^{-0.1s})}{s} \, .$$

The gain margin of this system is

(a) 
$$11.95 \,\mathrm{dB}$$
 (b)  $17.67 \,\mathrm{dB}$ 

(c)  $21.33 \,\mathrm{dB}$  (d)  $-23.9 \,\mathrm{dB}$ 

#### 2010

109. As shown in the figure, a negative feedback system has an amplifier of gain 100 with  $\pm 10 \%$  tolerance in the forward path, and an attenuator

of value  $\frac{9}{100}$  in the feedback path. The overall

system gain is approximately



110. For the system  $\frac{2}{(s+1)}$ , the approximate time taken for a step response to reach 98% of its final value is

- (c) 4 s (d) 8 s
- **111.** If the electrical circuit of Fig.(b) is an equivalent of the coupled tank system of Fig.(a), then



(a) Coupled tank (b) Electrical equivalent

- (a) A, B are resistances and C, D capacitances
- (b) A, C are resistances and B, D capacitances
- (c) A, B are capacitances and C, D resistances
- (d) A, C are capacitances and B, D resistances
- **112.** The frequency response of  $G(s) = \frac{1}{[s(s+1)(s+2)]}$ plotted in the complex  $G(j\omega)$  plane (for  $0 < \omega < \infty$ ) is



- **113.** The system  $\dot{x} = Ax + Bu$  with  $A = \begin{bmatrix} -1 & 2 \\ 0 & 2 \end{bmatrix}, B = \begin{bmatrix} 0 \\ 1 \end{bmatrix}$  is
  - (a) stable and controllable
  - (b) stable but uncontrollable
  - (c) unstable but controllable
  - (d) unstable and uncontrollable
- **114.** The characteristic equation of a closed-loop system is s(s + 1)(s + 3) + k(s + 2) = 0, k > 0. Which of the following statements is true?
  - (a) Its roots are always real
  - (b) It cannot have a break away point in the range -1 < Re[s] < 0
  - (c) Two of its roots tend to infinity along the asymptotes Re[s] = -1
  - (d) It may have complex roots in the right half plane

# 2011

**115.** The frequency response of a linear system  $G(j\omega)$  is provided in the tabular form below

$$|G(j\omega)| = 1.3 = 1.2 = 1.0 = 0.8 = 0.5 = 0.3$$
  
$$\angle G(j\omega) = -130^{\circ} - 140^{\circ} - 150^{\circ} - 160^{\circ} - 180^{\circ} - 200^{\circ}$$

The gain margin and phase margin of the system are

(a) 6 dB and  $30^{\circ}$  (b) 6 dB and  $-30^{\circ}$ 

- (c) -6 dB and  $30^{\circ}$  (d) -6 dB and  $-30^{\circ}$
- **116.** The steady state error of a unity feedback linear system for a unit step input is 0.1. The steady state error of the same system, for a pulse input r(t) having a magnitude of 10 and a duration of one second, as shown in the figure is

- (*b*) 0.1
- (c) 1 10

r(t)

**117.** A point z has been plotted in the complex plane, as shown in figure below.



The plot of the complex number  $y = \frac{1}{x}$  is



- **118.** An open loop system represented by the transfer
  - function G(s) =  $\frac{(s-1)}{(s+2)(s+3)}$  is
  - (a) stable and of the minimum phase type
  - $(b)\ {\rm stable}\ {\rm and}\ {\rm of}\ {\rm the}\ {\rm non-minimum}\ {\rm phase}\ {\rm type}$
  - (c) unstable and of the minimum phase type
  - (d) unstable and of the non-minimum phase type
- **119.** Let the Laplace transform of a function f(t) which exists for t > 0 be  $F_1(s)$  and the Laplace transform of its delayed version  $f(t \tau)$  be  $F_2^*(s)$ . Let  $F_1^*(s)$  be the complex conjugate of  $F_1(s)$  with the Laplace variable set as  $s = \sigma + j\omega$ .

If  $G(s) = \frac{F_2(s).F_1^*(s)}{|F_1(s)|^2}$ , then the inverse Laplace transform of G(s) is

- (a) an ideal impulse  $\delta(t)$
- (b) an ideal delayed impulse  $\delta(t-\tau)$
- (c) an ideal step function u(t)

(d) an ideal delayed step function impulse  $u(t - \tau)$ 

**120.** The open loop transfer function G(s) of a unity feedback control system is given as,

$$\mathbf{G}(s) = \frac{k\left(s + \frac{2}{3}\right)}{s^2\left(s + 2\right)}$$

From the root locus, it can be inferred that when k tends to positive infinity,

- (a) three roots with nearly equal real parts exist on the left half of the *s*-plane
- (b) one real root is found on the right half of s-plane
- (c) the root loci cross the  $j\omega$  axis for a finite value of  $k; k \neq 0$
- (d) three real roots are found on the right half of the s-plane
- **121.** The matrix  $[A] = \begin{bmatrix} 2 & 1 \\ 4 & -1 \end{bmatrix}$  is decomposed into a product of a lower triangular matrix [L] and an upper triangular matrix [U]. The properly decomposed [L] and [U] matrices respectively are

$$(a)\begin{bmatrix}1&0\\4&-1\end{bmatrix}\&\begin{bmatrix}1&1\\0&-2\end{bmatrix}(b)\begin{bmatrix}2&0\\4&-1\end{bmatrix}\&\begin{bmatrix}1&1\\0&1\end{bmatrix}$$
$$(c)\begin{bmatrix}1&0\\4&1\end{bmatrix}\&\begin{bmatrix}2&1\\0&-1\end{bmatrix}(d)\begin{bmatrix}2&0\\4&-3\end{bmatrix}\&\begin{bmatrix}1&0.5\\0&1\end{bmatrix}$$

122. Atwo-loop position control system is shown below.



The gain k of the Tacho-generator influences mainly

- (a) peak overshoot
- (b) natural frequency of oscillation
- (c) phase shift of the closed loop transfer function at very low frequencies  $(\omega \rightarrow 0)$
- (*d*) phase shift of the closed loop transfer function at very high frequencies  $(\omega \rightarrow \infty)$

# 2012

123. A system with transfer function

$$G(s) = \frac{(s^2 + 9)(s + 2)}{(s + 1)(s + 3)(s + 4)}$$

is excited by  $\sin(\omega t)$ . The steady-state output of the system is zero at

- (a)  $\omega = 1 \text{ rad/s}$  (b)  $\omega = 2 \text{ rad/s}$
- (c)  $\omega = 3 \text{ rad/s}$  (d)  $\omega = 4 \text{ rad/s}$

**124.** The unilateral Laplace transform of f(t) is  $\frac{1}{s^2 + s + 1}$ .

The unilateral Laplace transform of t f(t) is

(a) 
$$-\frac{s}{(s^2+s+1)^2}$$
 (b)  $-\frac{2s+1}{(s^2+s+1)^2}$   
(c)  $\frac{s}{(s^2+s+1)^2}$  (d)  $\frac{2s+1}{(s^2+s+1)^2}$ 

**125.** The state variable description of an LTI system is given by

$$\begin{pmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{pmatrix} = \begin{pmatrix} 0 & a_1 & 0 \\ 0 & 0 & a_1 \\ a_3 & 0 & 0 \end{pmatrix} \begin{pmatrix} x_1 \\ x_2 \\ x_3 \end{pmatrix} \begin{pmatrix} 0 \\ 0 \\ 1 \end{pmatrix} u, \ y = (1 \ 0 \ 0) \begin{pmatrix} x_1 \\ x_2 \\ x_3 \end{pmatrix}$$

where y is the output and u is the input. The system is controllable for

- (a)  $a_1 \neq 0, a_2 = 0, a_3 \neq 0$ (b)  $a_1 = 0, a_2 \neq 0, a_3 \neq 0$
- (c)  $a_1 = 0, a_2 \neq 0, a_3 = 0$
- (d)  $a_1 \neq 0, a_2 \neq 0, a_3 = 0$
- **126.** The feedback system shown below oscillates at 2 rad/s when



- (a) K = 2 and a = 0.75 (b) K = 3 and a = 0.75
- (c) K = 4 and a = 0.5 (d) K = 2 and a = 0.5

**127.** The input x(t) and output y(t) of a system are

related as  $y(t) = \int_{-\infty}^{t} x(\tau) \cos(3\tau) d\tau$ . The system is

- (a) time-invariant and stable
- (b) stable and not time-invariant
- (c) time-invariant and not stable
- (d) not time-invariant and not stable

#### 2013

**128.** The transfer function  $\frac{V_2(s)}{V_1(s)}$  of the circuit shown



**129.** Assuming zero initial condition, the response y(t) of the system given below to a unit step input u(t) is



**130.** The impulse response of a system is h(t) = t u(t). For an input u(t - 1), the output is

(a) 
$$\frac{t^2}{2}u(t)$$
 (b)  $\frac{t(t-1)}{2}u(t-1)$   
(c)  $\frac{(t-1)^2}{2}u(t-1)$  (d)  $\frac{t^2-1}{2}u(t-1)$ 

- **131.** Which one of the following statements is NOT TRUE for a continuous time causal and stable LTI system?
  - (a) All the poles of the system must lie on the left side of the  $j\omega$  axis.
  - (b) Zeros of the system can lie anywhere in the splane.
  - (c) All the poles must lie within |s| = 1.
  - (d) All the roots of the characteristic equation must be located on the left side of the  $j\omega$  axis.
- **132.** Two systems with impulse responses  $h_1(t)$  and  $h_2(t)$  are connected in cascade. Then the overall impulse response of the cascaded system is given by
  - (a) product of  $h_1(t)$  and  $h_2(t)$
  - (b) sum of  $h_1(t)$  and  $h_2(t)$
  - (c) convolution of  $h_1(t)$  and  $h_2(t)$
  - (d) subtraction of  $h_2(t)$  from  $h_1(t)$
- **133.** A source  $v_s(t) = V \cos 100\pi t$  has an internal impedance of  $(4 + j3)\Omega$ . If a purely resistive load connected to this source has to extract the maximum power out of the source, its value in  $\Omega$  should be

**134.** The Bode plot of a transfer function G (*s*) is shown in the figure below.



#### 6.16 Control Systems

The gain  $(20 \log |G(s)|)$  is 32 dB and -8 dB at 1 rad/s and 10 rad/s respectively. The phase is negative for all  $\omega$ . Then G(s) is

(a)  $\frac{39.8}{s}$  (b)  $\frac{39.8}{s^2}$ (c)  $\frac{32}{s}$  (d)  $\frac{32}{s^2}$ 

135. The signal flow graph for a system is given below.



**136.** The impulse response of a continuous time system is given by  $h(t) = \delta(t-1) + \delta(t-3)$ . The value of the step response at t = 2 is

| (a) | 0 | ( <i>b</i> ) | 1 |
|-----|---|--------------|---|
| (c) | 2 | (d)          | 3 |

# 2014

- **137.** In the formation of Routh-Hurwitz array for a polynomial, all the elements of a row have zero values. This premature termination of the array indicates the presence of
  - (a) Only one root at the origin
  - (b) Imaginary roots
  - (c) Only positive real roots
  - (d) Only negative real roots
- **138.** The root locus of a unity feedback system is shown in the figure



The closed loop transfer function of the system is

(a) 
$$\frac{C(s)}{R(s)} = \frac{K}{(s+1)(s+2)}$$

(b) 
$$\frac{C(s)}{R(s)} = \frac{-K}{(s+1)(s+2) + K}$$
  
(c) 
$$\frac{C(s)}{R(s)} = \frac{K}{(s+1)(s+2) - K}$$
  
(d) 
$$\frac{C(s)}{R(s)} = \frac{K}{(s+1)(s+2) + K}$$

139. The state transition matrix for the system

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ 1 & 1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 1 \\ 1 \end{bmatrix} u \text{ is}$$

$$(a) \begin{bmatrix} e^t & 0 \\ e^t & e^t \end{bmatrix} \qquad (b) \begin{bmatrix} e^t & 0 \\ t^2 e^t & e^t \end{bmatrix}$$

$$(c) \begin{bmatrix} e^t & 0 \\ t e^t & e^t \end{bmatrix} \qquad (d) \begin{bmatrix} e^t & t e^t \\ 0 & e^t \end{bmatrix}$$

140. For the transfer function

$$G(s) = \frac{5(s+4)}{s(s+0.25)(s^2+4s+25)}$$

5(a + 4)

The values of the constant gain term and the highest corner frequency of the Bode plot respectively are

141. The second order dynamic system

$$\frac{d\mathbf{X}}{dt} = \mathbf{P}\mathbf{X} + \mathbf{Q}\mathbf{u} \qquad \mathbf{y} = \mathbf{R}\mathbf{X}$$

has the matrices P, Q and R as follows:

$$\mathbf{P} = \begin{bmatrix} -1 & 1\\ 0 & -3 \end{bmatrix} \mathbf{Q} = \begin{bmatrix} 0\\ 1 \end{bmatrix} \mathbf{R} = \begin{bmatrix} 0 & 1 \end{bmatrix}$$

The system has the following controllability and observability properties:

- (a) Controllable and observable
- (b) Not controllable but observable
- (c) Controllable but not observable
- (d) Not controllable and not observable
- **142.** The signal flow graph of a system is shown below. U(s) is the input and C(s) is the output



Assuming,  $h_1 = b_1$  and  $h_0 = b_0 - b_1 a_1$ , then inputoutput transfer function,  $G(s) = \frac{C(s)}{U(s)}$  of the system is given by

(a) 
$$G(s) = \frac{b_0 s + b_1}{s^2 + a_0 s + a_1}$$
 (b)  $G(s) = \frac{a_1 s + a_0}{s^2 + b_1 s + b_0}$ 

(c) 
$$G(s) = \frac{b_1 s + b_0}{s^2 + a_1 s + a_0}$$
 (d)  $G(s) = \frac{a_0 s + a_1}{s^2 + b_0 s + b_1}$ 

- 143. A single-input single-output feedback system has forward transfer function G(s) and feedback transfer function H(s). It is given that |G(s).H(s)| < 1. Which of the following is true about the stability of the system?
  - (a) The system is always stable
  - (b) The system is stable if all zeros of G(s).H(s) are in left half of the s-plane
  - (c) The system is stable if all poles of G(s).H(s) are in left half of the s-plane
  - (d) It is not possible to say whether or not the system is stable from the information given
- **144.** The block diagram of a system is shown in the figure



If the desired transfer function of the system is

(b) s

$$\frac{\mathcal{C}(s)}{\mathcal{R}(s)} = \frac{s}{s^2 + s + 1}$$

Then G(s) is

(a) 
$$1$$

(c) 
$$1/s$$
 (d)  $\frac{1}{s}$ 

**145.** Consider the system described by following state space equations

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -1 & -1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u; \ \mathbf{y} = \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

If u is unit step input, then the steady state error of the system is

1

(a) 0 (b) 
$$\frac{1}{2}$$

(c) 
$$\frac{2}{3}$$
 (d)

**146.** The magnitude Bode plot of a network is shown in the figure



The maximum phase angle  $\phi_{\rm m}$  and the corresponding gain  $G_{\rm m}$  respectively, are

| (a) $-30^{\circ}$ and $1.73$ dB | (b) $-30^{\circ}$ and $4.77$ dB                 |
|---------------------------------|---|
| (c) $+30^{\circ}$ and $4.77$ dB | $(d) + 30^{\circ} \text{ and } 1.73 \text{ dB}$ |

## 2015

147. For the signal-flow graph shown in the figure, which one of the following expressions is equal



**148.** A Bode magnitude plot for the transfer function G(s) of a plant is shown in the figure. Which one of the following transfer functions best describes the plant?



(c) 
$$\frac{s+1000}{10s(s+10)}$$
 (d)  $\frac{s+1000}{10(s+10)}$ 

(a)

**149.** In the signal flow diagram given in the figure,  $u_1$  and  $u_2$  are possible inputs whereas  $y_1$  and  $y_2$  are possible outputs. When would the SISO system derived from this diagram be controlable and observable ?



#### 6.18 Control Systems

- (a) When u<sub>1</sub> is the only input and y<sub>1</sub> is the only output
- (b) When  $u_2$  is the only input and  $y_1$  is the only output
- (c) When u<sub>1</sub> is the only input and y<sub>2</sub> is the only output
- (d) When  $\mathbf{u}_2$  is the only input and  $y_2$  is the only output

**150.** Find the transfer function  $\frac{Y(s)}{X(s)}$  of the system given below.



- **151.** The transfer function of a second order real system with a perfectly flat magnitude response of unity has a pole at (2 j3). List all the poles and zeroes.
  - (a) Poles at  $(2 \pm j3)$ , no zeroes
  - (b) Poles at  $(\pm 2 j3)$ , one zero at origin
  - (c) Poles at (2 j3), (-2 + j3), zeroes at (-2 j3), (2 + j3)
  - (d) Poles at  $(2 \pm j3)$ , zeroes at  $(-2 \pm j3)$
- **152.** The open loop poles of a third order unity feedback system are at 0, -1, -2. Let the frequency corresponding to the point where the root locus of the system transits to unstable region be K. Now suppose we introduce a zero in the open loop transfer function at -3, while keeping all the earlier open loop poles intact. Which one of the following is TRUE about the point where the root locus of the modified system transits to unstable region?
  - $(a) \,$  It corresponds to a frequency greater than K
  - (b) It corresponds to a frequency less than K
  - (c) It corresponds to a frequency K
  - (d) Root locus of modified system never transits to unstable region

**153.** Nyquist plot of the functions  $G_1(s)$  and  $G_2(s)$  are shown in figure.



Nyquist plot of the product of  $G_1(s)$  and  $G_2(s)$  is



**154.** An open loop transfer function G(s) of a system is

$$G(s) = \frac{K}{s(s+1)(s+2)}$$

For a unity feedback system, the breakaway point of the root loci on the real axis occurs at,

(a) 
$$-0.42$$
 (b)  $-1.58$   
(c)  $-0.42$  and  $-1.58$  (d) None of the above

2016

**155.** The transfer function of a system is 
$$\frac{Y(s)}{R(s)} = \frac{s}{s+2}$$

The steady state output y(t) is A  $\cos(2t + \phi)$  for the input  $\cos(2t)$ . The values of A and  $\phi$ , respectively are

(a) 
$$\frac{1}{\sqrt{2}}, -45^{\circ}$$
 (b)  $\frac{1}{\sqrt{2}}, +45^{\circ}$   
(c)  $\sqrt{2}, -45^{\circ}$  (d)  $\sqrt{2}, +45^{\circ}$ 

156. The phase cross-over frequency of the transfer

function 
$$G(s) = \frac{100}{(s+1)^3}$$
 in rad/s is

- (a)  $\sqrt{3}$  (b)  $\frac{1}{\sqrt{3}}$
- (c) 3 (d)  $3\sqrt{3}$

157. Consider the following asymptotic Bode magnitude plot ( $\omega$  is in rad/s).



Which one of the following transfer functions is best represented by the above Bode magnitude plot?

(a) 
$$\frac{2s}{(1+0.5s)(1+0.25s)^2}$$
 (b)  $\frac{4(1+0.5s)}{s(1+0.25s)}$   
(c)  $\frac{2s}{(1+2s)(1+4s)}$  (d)  $\frac{4s}{(1+2s)(1+4s)^2}$ 

158. Loop transfer function of a feedback system is

 $\mathbf{G}(s)\mathbf{H}\left(s\right)=\frac{s+3}{s^{2}(s-3)}$  . Take the Nyquist contour in the clockwise direction. Then the Nyquist plot of G(s) encircles -1 + i0

- (a) once in clockwise direction
- (b) twice in clockwise direction
- (c) once in anticlockwise direction
- (d) twice in anticlockwise direction

**159.** The value of the integral 
$$2\int_{-\infty}^{\infty} \left(\frac{\sin 2\pi t}{\pi t}\right) dt$$
 is equal to  
(a) 0 (b) 0.5  
(c) 1 (d) 2

(c) 1

**160.** Let  $P = \begin{bmatrix} 3 & 1 \\ 1 & 3 \end{bmatrix}$ . Consider the set S of all vectors

$$\begin{pmatrix} x \\ y \end{pmatrix}$$
 such that  $a^2 + b^2 = 1$  where  $\begin{pmatrix} a \\ b \end{pmatrix} = P \begin{pmatrix} x \\ y \end{pmatrix}$ .

Then S is

- (a) a circle of radius  $\sqrt{10}$
- (b) a circle of radius  $\frac{1}{\sqrt{10}}$
- (c) an ellipse with major axis along  $\begin{pmatrix} 1 \\ 1 \end{pmatrix}$

(d) an ellipse with minor axis along  $\begin{pmatrix} 1 \\ 1 \end{pmatrix}$ 

161. The open loop transfer function of a unity feedback control system is given by

$$G(s) = \frac{K(s+1)}{s(1+Ts)(1+2s)}, K > 0, T > 0$$

The closed loop system will be stable if,

$$\begin{array}{ll} (a) & 0 < T < \frac{4\left(K+1\right)}{K-1} & (b) & 0 < K < \frac{4\left(T+2\right)}{T-2} \\ (c) & 0 < K < \frac{T+2}{T-2} & (d) & 0 < T < \frac{8\left(K+1\right)}{K-1} \end{array}$$

- 162. A second-order real system has the following properties :
  - A. the damping ratio  $\xi = 0.5$  and undamped natural frequency  $\omega_n = 10 \text{ rad/s}$ ,
  - B. the steady state value of the output, to a unit step input, is 1.02.

The transfer function of the system is

(a) 
$$\frac{1.02}{s^2 + 5s + 100}$$
 (b)  $\frac{1.02}{s^2 + 10s + 100}$   
(c)  $\frac{100}{s^2 + 10s + 100}$  (d)  $\frac{102}{s^2 + 5s + 100}$ 

163. The gain at the breakaway point of the root locus of a unity feedback system with open loop transfer

| function $G(s)$ - | Ks is          |
|-------------------|----------------|
|                   | (s-1)(s-4) 15  |
| ( <i>a</i> ) 1    | (b) 2          |
| (c) 5             | ( <i>d</i> ) 9 |

## 2017

164. A closed loop system has the characteristic equation given by  $s^3 + Ks^2 + (K+2)s + 3 = 0$ . For this system to be stable, which one of the following conditions should be satisfied?

(a) 
$$0 < K < 0.5$$
 (b)  $0.5 < K < 1$   
(c)  $0 < K < 1$  (d)  $K > 1$ 

165. The transfer function of a system is given by,

$$\frac{V_0(s)}{V_1(s)} = \frac{1-s}{1+s}$$
. Let the output of the system be

 $v_{\scriptscriptstyle 0}(t) {=} V_{\scriptscriptstyle m} \sin \left( \omega t + \varphi \right)$  for the input,  $v_{\scriptscriptstyle i}(t) {=} V_{\scriptscriptstyle m} \sin$  $(\omega t)$ . Then the minimum and maximum values of  $\phi$  (in radians) are respectively

(a) 
$$\frac{-\pi}{2}$$
 and  $\frac{\pi}{2}$  (b)  $\frac{-\pi}{2}$  and 0

(c) 0 and 
$$\frac{\pi}{2}$$
 (d)  $-\pi$  and 0

166. The transfer function of the system Y(s)/U(s)whose state-space equations are given below is:

$$\begin{bmatrix} \dot{x}_1(t) \\ \dot{x}_2(t) \end{bmatrix} = \begin{bmatrix} 1 & 2 \\ 2 & 0 \end{bmatrix} \begin{bmatrix} x_1(t) \\ x_2(t) \end{bmatrix} + \begin{bmatrix} 1 \\ 2 \end{bmatrix} u(t)$$
$$y(t) = \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} x_1(t) \\ x_2(t) \end{bmatrix}$$

6.20 Control Systems

(a) 
$$\frac{(s+2)}{(s^2-2s-2)}$$
 (b)  $\frac{(s-2)}{(s^2+s-4)}$   
(c)  $\frac{(s-4)}{(s^2+s-4)}$  (d)  $\frac{(s+4)}{(s^2-s-4)}$ 

**167.** In the system whose signal flow graph is shown in figure,  $U_1(s)$  and  $U_2(s)$  are inputs. The transfer



## 2018

**168.** Consider a lossy transmission line with  $V_1$  and  $V_2$  as the sending and receiving end voltages, respectively. Z and X are the series impedance and reactance of the line, respectively. The steady-state stability limit for the transmission line will be

(a) greater than 
$$\left| \frac{V_1 V_2}{X} \right|$$
 (b) less than  $\left| \frac{V_1 V_2}{X} \right|$   
(c) equal to  $\left| \frac{V_1 V_2}{X} \right|$  (d) equal to  $\left| \frac{V_1 V_2}{Z} \right|$ 

**169.** Match the transfer functions of the second-order systems with the nature of the system given below.

#### Transfer functions

#### Nature of system



**170.** The number of roots of polynomial,  $s^7 + s^6 + 7s^5 + 14s^4 + 31s^3 + 73s^2 + 25s + 200$ , in the open left half of the complex plane is

| (a) | 3 | <i>(b)</i> | 4 |
|-----|---|------------|---|
|     |   |            |   |

(c) 5 (d) 6

# 2019

**171.** The characteristic equation of a linear timeinvariant (LTI) system is given by

$$\Delta(s) = s^4 + 3s^3 + 3s^2 + s + k = 0$$

The system BIBO stable if

(a) 
$$k > 3$$
  
(b)  $0 < k < \frac{8}{6}$   
(c)  $0 < k < \frac{12}{9}$   
(d)  $k > 6$ 

172. A system transfer function is

$$H(s) = \frac{a_1 s^2 + b_1 s + c_1}{a_2 s^2 + b_2 s + c_2}.$$

If  $\mathbf{a}_1 = \mathbf{b}_1 = 0$ , and all other coefficients are positive, the transfer function represents a

- (a) high pass filter (b) notch filter
- (c) low pass filter (d) band pass filter
- **173.** The open loop transfer function of a unity feedback system is given by

$$\mathbf{G(s)} = \frac{\pi \mathbf{e}^{-0.25s}}{s},$$

In G(s) plane, the Nyquist plot of G(s) passes through the negative real axis at the point.

174. The asymptotic Bode magnitude plot of a minimum phase transfer function G(s) is shown below.



Consider the following two statements.

Statement I : Transfer function G(s) has three poles and one zero.

Statement II : At very high frequency  $(\omega \rightarrow \infty)$ ,

the phase angle 
$$\angle G(j\omega) = -\frac{3\pi}{2}$$

Which one of the following option is correct?

- (a) Statement I is false and statement II is true.
- (b) Both the statements are true.
- (c) Both the statements are false.
- (*d*) Statement I is true and statement II is false. **175.** The transfer function of a phase lead compensator

is given by D(s) = 
$$\frac{3\left(s + \frac{1}{3T}\right)}{\left(s + \frac{1}{T}\right)}$$

The frequency (in rad/sec), at which  $\angle D(j\omega)$  is maximum, is

(a) 
$$\sqrt{3T^2}$$
 (b)  $\sqrt{\frac{3}{T^2}}$   
(c)  $\sqrt{3T}$  (d)  $\sqrt{\frac{1}{3T^2}}$ 

 ${\bf 176.}\ Consider a state-variable model of a system$ 

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -\alpha & -2\beta \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ \alpha \end{bmatrix} \mathbf{r}; \ \mathbf{y} = \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$
  
Where y is the output, and r is the input. The

damping ratio  $\xi$  and the undamped natural frequency  $\omega_n$  (rad/sec) of the system are given by

(a) 
$$\xi = \sqrt{\alpha}; \omega_n = \frac{\beta}{\sqrt{\alpha}}$$
 (b)  $\xi = \frac{\sqrt{\alpha}}{\beta}; \omega_n = \sqrt{\beta}$   
(c)  $\xi = \sqrt{\beta}; \omega_n = \sqrt{\alpha}$  (d)  $\xi = \frac{\beta}{\sqrt{\alpha}}; \omega_n = \sqrt{\alpha}$ 

# 2020

**177.** Which of the options is an equivalent representation of the signal flow graph shown here





**178.** Consider a permanent magnet dc (PMDC) motor which is initially at rest. At t = 0, a dc voltage of 5 V is applied to the motor. Its speed monotonically increases from 0 rad/s to 6.32 rad/s in 0.5 s and finally settles to 10 rad/s. Assuming that the armature inductance of the motor is negligible, the transfer function of the motor is

(a) 
$$\frac{10}{0.5s+1}$$
 (b)  $\frac{2}{0.5s+1}$   
(c)  $\frac{2}{s+0.5}$  (d)  $\frac{10}{s+0.5}$ 

**179.** Which of the following options is correct for the system shown below?



- (a)  $4^{th}$  order and stable
- (b) 3<sup>rd</sup> order and stable
- (c)  $4^{th}$  order and unstable
- (d)  $3^{rd}$  order and unstable
- **180.** Consider a negative unity feedback system with the forward path transfer function

 $\frac{s^2 + s + 1}{s^3 + 2s^2 + 2s + K}$ , where K is a positive real number. The value of K for which the system will have some of its poles in the imaginary axis is \_\_\_\_.

(c) 7 (d) 6

181. For the closed-loop system shown, the transfer function  $\frac{E(s)}{R(s)}$  is



# 6.22 Control Systems

2022

182. The transfer function of a real system, H(s), is

given as : 
$$H(s) = \frac{As + B}{s^2 + Cs + D}$$
, where A, B, C and

D are positive constants. This system cannot operate as

- (a) low pass filter. (b) high pass filter.
- (c) band pass filter. (d) an integrator.
- 183. The Bode magnitude plot of a first order stable system is constant with frequency. The asymptotic value of the high frequency phase, for the system, is  $-180^{\circ}$ . This system has



- (a) one LHP pole and one RHP zero at the same frequency.
- (b) one LHP pole and one LHP zero at the same frequency.
- (c) two LHP poles and one RHP zero.
- (d) two RHP poles and one LHP zero.
- 184. The open loop transfer function of a unity gain negative feedback system is given by k

 $G(s) = \frac{k}{s^2 + 2s - 5}$ . The range of k for which the system is stable, is

(a) k > 3 (b) k < 3

(c) k > 5 (d) k < 5

**185.** The open loop transfer function of a unity gain negative feedback system is given as

$$G(s) = \frac{1}{s(s+1)} \label{eq:G}$$

The Nyquist contour in the s-plane encloses the entire right half plane and a small neighbourhood around the origin in the left half plane, as shown in the figure below. The number of encirclements of the point (-1 + j0) by the Nyquist plot of G(s), corresponding to the Nyquist contour, is denoted as N. Then N equals to



**186.** The damping ratio and undamped natural frequency of a closed loop system as shown in the figure, are denoted as  $\xi$  and  $\omega_n$ , respectively. The values of  $\xi$  and  $\omega_n$  are



(a)  $\xi = 0.5$  and  $\omega_n = 10$  rad/s

(b)  $\xi = 0.1 \text{ and } \omega_n = 10 \text{ rad/s}$ 

(c) 
$$\xi = 0.707$$
 and  $\omega_n = 10$  rad/s

(d)  $\xi = 0.707$  and  $\omega_n = 100$  rad/s

#### 2023

187. For the block diagram shown in the figure, the



**188.** In the Nyquist plot of the open-loop transfer function

$$G(s)H(s) = \frac{3s+5}{s-1}$$

corresponding to the feedback loop shown in the figure, the infinite semi-circular arc of the Nyquist contour in s-plane is mapped into a point at



**189.** Consider a unity-gain negative feedback system consisting of the plant G(s) (given below) and a proportional-integral controller. Let the proportional gain and integral gain be 3 and 1, respectively. For a unit step reference input, the final values of the controller output and the plant output, respectively, are

$$G(s) = \frac{1}{s-1}$$
(a)  $\infty, \infty$  (b) 1, 0  
(c) 1, -1 (d) -1, 1

190. Consider a lead compensator of the form

$$K(s) = \frac{1 + \frac{s}{\alpha}}{1 + \frac{s}{\beta\alpha}}, \beta > 1, \alpha > 0$$

The frequency at which this compensator produces maximum phase lead is 4 rad/s. At this frequency, the gain amplification provided by the controller, assuming asymptotic Bode-magnitude plot of K(s), is 6 dB. The values of  $\alpha$ ,  $\beta$ , respectively, are

| (a) 1, 16 | (b) 2, 4       |
|-----------|----------------|
| (c) 3, 5  | (d) 2.66, 2.25 |

# NUMERICAL TYPE QUESTIONS

#### 1992

1. For what range of K is the following system (Figure) asymptotically stable ? Assume  $K \ge 0$ 



#### 1994

**Direction (Q. 2) :** Indicate whether the following statement is TRUE or FALSE. Write the indicating work fully and legibly. A 'FALSE' answer must be accompanied by a very brief (preferably one or two sentences) justification.

2. The closed loop system, of Figure, is stable if the

transfer function T (s) = 
$$\frac{C(s)}{R(s)}$$
 is stable.



3. The number of positive real roots of the equation  $s^3 - 2s + 2 = 0$  is \_\_\_\_\_

#### 1995

4. Closed loop stability implies that [1 + G(s) H(s)] has only \_\_\_\_\_ in the left half of the s-plane.

# 2014

5. For the given system, it is desired that the system be stable. The minimum value of  $\alpha$  for this condition is \_\_\_\_\_.



6. The Bode magnitude plot of the transfer function

G(s) = 
$$\frac{K(1+0.5s)(1+as)}{s\left(1+\frac{s}{8}\right)(1+bs)\left(1+\frac{s}{36}\right)}$$
 I shown below:

Note that -6 dB/octave = -20 dB/decade. The value of  $\frac{a}{b\text{K}}$  is \_\_\_\_\_



7. The closed-loop transfer function of a system is

 $T(S) = \frac{4}{(s^2 + 0.4s + 4)}$ . The steady state error due to unit step input is \_\_\_\_\_.

8. A system with the open loop transfer function

G(s) = 
$$\frac{K}{s(s+2)(s^2+2s+2)}$$

is connected in a negative feedback configuration with a feedback gain of unity. For the closed loop system to be marginally stable, the value of K is

2015

**9.** An open loop control system results in a response of  $e^{-2t}(\sin 5t + \cos 5t)$  for a unit impulse input. The DC gain of the control system is \_\_\_\_\_.

# 6.24 Control Systems

# 2016

10. Consider a linear time-invariant system with transfer function

$$\mathbf{H}(s) = \frac{1}{(s+1)}$$

If the input is  $\cos(t)$  and the steady state output is  $A \cos(t + \alpha)$ , then the value of A is \_\_\_\_\_.

11. The line integral of the vector field

$$\mathbf{F} = 5xz\hat{i} + \left(3x^2 + 2y\right)j + x^2zk$$

along a path from (0, 0, 0) to (1, 1, 1)parameterized by  $(t, t^2, t)$  is \_\_\_\_\_.

# 2017

12. Consider the unity feedback control system shown. The value of K that results in a phase margin of the system to be 30° is \_\_\_\_\_. (Give the answer up to two decimal places.)



**13.** For a system having transfer function  $G(s) = \frac{-s+1}{s+1}$ , a unit step input is applied at time t = 0. The value of the response of the system at t = 1.5 sec (rounded off to three decimal places) is\_\_\_\_\_.

# 2018

14. Consider a unity feedback system with forward transfer function given by

$$G(s) = \frac{1}{(s+1)(s+2)}$$

The steady-state error in the output of the system for a unit-step input is \_\_\_\_\_ (up to 2 decimal places).

# 2021

- 17. The Bode magnitude plot for the transfer function  $\frac{V_o(s)}{V_i(s)}$  of the circuit is as shown. The value of R is \_\_\_\_\_  $\Omega$ . (Round off to 2 decimal places.)



**15.** The unit step response y(t) of a unity feedback system with open-loop transfer function G(s)H(s)

 $=\frac{K}{(s+1)^{2}(s+2)}$  is shown in the figure. The

value of K is \_\_\_\_\_ (upto 2 decimal places).



# 2020

16. Consider a negative unity feedback system with forward path transfer function Κ  $G(s) = \frac{\pi}{(s+a)(s-b)(s+c)}$ , where K, a, b, c are positive real numbers. For a Nyquist path enclosing the entire imaginary axis and right half of the s-plane in the clockwise direction, the Nyquist plot of [1 + G(s)], encircles the origin of [1 + G(s)]-plane once in the clockwise direction and never passes through this origin for a certain value of 'K'. Then the number of poles of  $\displaystyle \frac{G(s)}{1+G(s)}$ lying in the open right half of the s-plane is \_\_\_\_\_.

18. Consider a closed-loop system as shown,

 $G_{_{p}}(s) \!=\! \frac{14.4}{s(1\!+\!0.1s)}$  is the plant transfer function

and  $G_c(s) = 1$  is the compensator. For a unit-step input, the output response has damped oscillations. The damped natural frequency is \_\_\_\_\_\_ rad/s. (Round off to 2 decimal places).



19. In the given figure, plant

**Numerical Type Questions** 

$$\begin{split} G_{p}(S) &= \frac{2.2}{(1+0.1s)(1+0.4s)(1+1.2s)} \text{ and compensator} \\ G_{c}(S) &= K \Biggl( \frac{1+T_{1}s}{1+T_{2}s} \Biggr). \end{split}$$

The external disturbance input is D(s). It is desired that when the disturbance is a unit step, the steady-state error should not exceed 0.1 unit. The minimum value of K is \_\_\_\_\_.

(Round off to 2 decimal places.)



**20.** The state space representation of a first-order system is given as

 $\dot{\mathbf{x}} = -\mathbf{x} + \mathbf{u}$ 

y = x

where, x is the state variable, u is the control input and y is the controlled output. Let u = -Kx be the control law, where K is the controller gain. To place a closed-loop pole at -2, the value of K is \_\_\_\_\_.

|                          |                          |                          |                         | ANS                      | WERS                     |                          |                          |                          |                          |
|--------------------------|--------------------------|--------------------------|-------------------------|--------------------------|--------------------------|--------------------------|--------------------------|--------------------------|--------------------------|
| MCQ Ty                   | pe Questic               | ons                      |                         |                          |                          |                          |                          |                          |                          |
| 1.(b)                    | <b>2.</b> ( <i>b</i> )   | <b>3.</b> ( <i>a</i> )   | <b>4.</b> ( <i>a</i> )  | <b>5.</b> (a)            | <b>6.</b> (c)            | <b>7.</b> ( <i>a</i> )   | <b>8.</b> (*)            | <b>9.</b> ( <i>a</i> )   | <b>10.</b> ( <i>a</i> )  |
| <b>11.</b> $(b)$         | <b>12.</b> (c)           | 13.(c)                   | 14.(d)                  | 15.(c)                   | 16.(b)                   | 17.(b)                   | 18.(d)                   | 19.(b)                   | <b>20.</b> $(d)$         |
| <b>21.</b> ( <i>b</i> )  | <b>22.</b> (c)           | <b>23.</b> ( <i>a</i> )  | <b>24.</b> ( <i>a</i> ) | <b>25.</b> $(c)$         | <b>26.</b> ( <i>a</i> )  | <b>27.</b> $(d)$         | <b>28.</b> (c)           | <b>29.</b> ( <i>a</i> )  | <b>30.</b> (c)           |
| <b>31.</b> $(c)$         | <b>32.</b> $(b)$         | <b>33.</b> (d)           | <b>34.</b> ( <i>a</i> ) | <b>35.</b> $(d)$         | <b>36.</b> (c)           | <b>37.</b> ( <i>a</i> )  | <b>38.</b> (c)           | <b>39.</b> $(d)$         | <b>40.</b> ( <i>a</i> )  |
| <b>41.</b> (c)           | <b>42.</b> ( <i>a</i> )  | <b>43.</b> $(b)$         | <b>44.</b> (c)          | <b>45.</b> (c)           | <b>46.</b> (c)           | <b>47.</b> ( <i>b</i> )  | <b>48.</b> ( <i>b</i> )  | <b>49.</b> ( <i>a</i> )  | <b>50.</b> ( <i>b</i> )  |
| <b>51.</b> $(d)$         | <b>52.</b> (c)           | <b>53.</b> ( <i>a</i> )  | <b>54.</b> ( <i>a</i> ) | <b>55.</b> $(b)$         | <b>56.</b> ( <i>a</i> )  | <b>57.</b> $(d)$         | <b>58.</b> ( <i>a</i> )  | <b>59.</b> (c)           | <b>60.</b> ( <i>a</i> )  |
| <b>61.</b> ( <i>b</i> )  | <b>62.</b> ( <i>a</i> )  | <b>63.</b> $(d)$         | <b>64.</b> (c)          | <b>65.</b> (c)           | <b>66.</b> ( <i>b</i> )  | <b>67.</b> ( <i>a</i> )  | <b>68.</b> $(d)$         | <b>69.</b> ( <i>d</i> )  | <b>70.</b> ( <i>b</i> )  |
| <b>71.</b> (c)           | <b>72.</b> (c)           | <b>73.</b> ( <i>a</i> )  | <b>74.</b> $(d)$        | <b>75.</b> (c)           | <b>76.</b> (c)           | <b>77.</b> ( <i>b</i> )  | <b>78.</b> (c)           | <b>79.</b> ( <i>d</i> )  | <b>80.</b> ( <i>b</i> )  |
| <b>81.</b> ( <i>a</i> )  | <b>82.</b> (c)           | <b>83.</b> (b)           | <b>84.</b> $(d)$        | <b>85.</b> ( <i>a</i> )  | <b>86.</b> ( <i>b</i> )  | <b>87.</b> (c)           | <b>88.</b> (a)           | <b>89.</b> $(d)$         | <b>90.</b> ( <i>b</i> )  |
| <b>91.</b> ( <i>d</i> )  | <b>92.</b> ( <i>b</i> )  | <b>93.</b> ( <i>d</i> )  | <b>94.</b> ( <i>d</i> ) | <b>95.</b> ( <i>a</i> )  | <b>96.</b> ( <i>a</i> )  | <b>97.</b> ( <i>d</i> )  | <b>98.</b> ( <i>b</i> )  | <b>99.</b> (a)           | <b>100.</b> ( <i>c</i> ) |
| <b>101.</b> ( <i>c</i> ) | <b>102.</b> (c)          | <b>103.</b> $(d)$        | 104.(d)                 | <b>105.</b> $(d)$        | 106.(b)                  | 107.(d)                  | 108.(d)                  | <b>109.</b> ( <i>a</i> ) | <b>110.</b> ( <i>c</i> ) |
| <b>111.</b> $(d)$        | <b>112.</b> ( <i>a</i> ) | <b>113.</b> $(c)$        | <b>114.</b> $(c)$       | <b>115.</b> ( <i>a</i> ) | <b>116.</b> ( <i>a</i> ) | 117.(d)                  | <b>118.</b> $(b)$        | 119.(b)                  | <b>120.</b> ( <i>a</i> ) |
| 121.(d)                  | <b>122.</b> ( <i>a</i> ) | 123.(c)                  | 124.(d)                 | 125.(d)                  | <b>126.</b> ( <i>a</i> ) | 127.(b)                  | 128.(d)                  | 129.(b)                  | <b>130.</b> $(c)$        |
| 131.(c)                  | <b>132.</b> $(c)$        | <b>133.</b> $(c)$        | 134.(b)                 | <b>135.</b> $(a)$        | 136.(b)                  | 137.(b)                  | <b>138.</b> $(c)$        | 139.(c)                  | <b>140.</b> ( <i>a</i> ) |
| 141.(c)                  | 142.(c)                  | <b>143.</b> ( <i>a</i> ) | 144.(b)                 | 145.(a)                  | 146.(c)                  | 147.(b)                  | 148.(d)                  | 149.(b)                  | 150.(c)                  |
| 151.(d)                  | 152.(d)                  | 153.(b)                  | 154.(a)                 | 155.(b)                  | <b>156.</b> ( <i>a</i> ) | <b>157.</b> ( <i>a</i> ) | <b>158.</b> ( <i>a</i> ) | 159.(d)                  | <b>160.</b> $(c)$        |
| <b>161.</b> ( <i>c</i> ) | <b>162.</b> $(b)$        | <b>163.</b> ( <i>a</i> ) | 164.(d)                 | <b>165.</b> $(d)$        | <b>166.</b> $(d)$        | <b>167.</b> ( <i>a</i> ) | <b>168.</b> $(b)$        | <b>169.</b> ( <i>c</i> ) | <b>170.</b> ( <i>a</i> ) |
| <b>171.</b> (b)          | <b>172.</b> (c)          | <b>173.</b> (b)          | 174. (a)                | <b>175.</b> (d)          | 176. (d)                 | <b>177.</b> (c)          | 178.(b)                  | <b>179.</b> (c)          | <b>180.</b> ( <i>b</i> ) |
| <b>181.</b> (c)          | <b>182.</b> (b)          | <b>183.</b> (a)          | <b>184.</b> (c)         | <b>185.</b> (b)          | <b>186.</b> (a)          | <b>187.</b> (b)          | <b>188.</b> (c)          | <b>189.</b> (d)          | <b>190.</b> (b)          |
|                          |                          |                          |                         |                          |                          |                          |                          |                          |                          |

| $1.(K \le 4/5)$           | <b>2.</b> (True) | <b>3.</b> (One)           | <b>4.</b> (roots)              | <b>5.</b> (0.618)   | <b>6.</b> (0.75)   |
|---------------------------|------------------|---------------------------|--------------------------------|---------------------|--------------------|
| <b>7.</b> (0 to 0)        | 8.(5 to 5)       | <b>9.</b> 0.241           | <b>10.</b> (0.707)             | <b>11.</b> (4.41)   | <b>12.</b> (1.047) |
| 13.(0.554)                | <b>14.</b> 0.66  | <b>15.</b> 8              | <b>16.</b> $(2 \text{ to } 2)$ | 17.(0.09  to  0.11) |                    |
| <b>18.</b> (10.80 to11.00 | )                | <b>19.</b> (9.50 to 9.60) | <b>20.</b> (1 to 1)            |                     |                    |

# **EXPLANATIONS**

# **MCQ TYPE QUESTIONS**

1. Given :  $G(s) = \frac{1}{(s-1)(s+2)(s+3)}$ 

Since the open-loop transfer function has no poles at the origin, the Nyquist path is as shown below :



**Path I:** 
$$G(j\omega) = \frac{1}{(j\omega - 1)(j\omega + 2)(j\omega + 3)}$$
  
 $G(j0) = -\frac{1}{6} = \frac{1}{6} \lfloor \frac{189^{\circ}}{6} \rfloor$ 

**Path II:**  $G(j\omega) = \frac{1}{(j\omega)^3} = \frac{1}{(\operatorname{Re}^{j0})^3} \operatorname{R} \to \infty$ =  $\delta_e^{-j30} \qquad \delta \to 0$ 

Hence as  $\phi$  goes from – 90° to + 90° in path II, G(jw) goes from 270° to – 270° as shown below



or 
$$Y = \frac{2GU}{1 - GH}$$
  
or  $\frac{Y}{U} = \frac{2G}{1 - GH}$ 

 Only (a) is a valid root loci since in (b), (c) and (d) the root loci lies to the left of an even number of zeros and poles.

4. 
$$E_{(s)} = X_{(s)} - Y_{(s)}H$$

$$= X_{(s)} - \frac{G}{1 + GH} H.X_{(s)}$$

$$= X_{(s)} \frac{1}{1 + GH}$$

$$X_{(s)} = \frac{1}{s}$$

$$\xrightarrow{+ } \underbrace{E(t)}_{-} \underbrace{G = \frac{s + 1}{s^{2} + 5s + a}}_{-} \underbrace{H = \frac{1}{s^{4}}}_{+}$$
Hence 
$$E_{(s)} = \frac{1}{s(1 + GH)}$$

$$= \frac{1}{s} \left[ \frac{1}{1 + \frac{s + 1}{(s + 4)(s^{2} + 5s + a)}} \right]$$

The steady state error,

$$e_{ss} = \lim_{t \to \infty} \epsilon_{(t)}$$
  
=  $\lim_{t \to \infty} sE_{(s)}$   
=  $\lim_{t \to \infty} \frac{1}{1 + \frac{s+1}{(s+4)(s^2 + 5s + a)}}$   
=  $\frac{1}{1 + \frac{1}{4a}} = \frac{4a}{4a + 1} = 0$   
 $a = 0$   
 $H(s) = -\frac{s}{2} - \frac{s+1-1}{2} - 1 - \frac{1}{2}$ 

5. (a)  $H(s) = \frac{s}{s+1} = \frac{s+1-1}{s+1} = 1 - \frac{1}{s+1}$  $h(t) = \delta_{(t)} - e^{-t} u_{(t)}$ 

This matches best with (P)

or

(b) 
$$H(s) = \frac{s}{(s+1)^2}$$

 $h(t) = t e^{-t} u_{(t)}$ This matches best with (Q)

(c) 
$$H(s) = \frac{1}{s^2 + s + 1} = \frac{\omega_n^2}{s^2 + 2\delta\omega_n s + w_n^2}$$
  
Comparing (1) and (2), we get

Comparing 
$$(1)$$
 and  $(2)$ , we get

$$\omega_n = 1,$$
  

$$2\delta \omega_n = 1,$$
  

$$\Rightarrow \qquad \delta = \frac{1}{2}$$
  

$$h(t) = \frac{\omega_n}{\sqrt{1 - \delta^2}} e^{-\delta \omega_n t} \sin(\omega_n \sqrt{1 - \delta^2 t})$$
  

$$= \frac{1}{\sqrt{\frac{3}{4}}} e^{-1/2} \sin\left(\sqrt{\frac{3}{4}}t\right)$$

This matches best with (S)

(d) 
$$H(s) = \frac{1}{s^2 + 1}$$
$$h(t) = \sin(t)$$

Such matches best with (R)

Hence, a - P, b - Q, c - S, d - R

6.



From the figure,  $x_1 = -x_1 + 3r(t)$ 

The A matrix is (-1) and hence choice (c) is correct. The transfer function of a phase-lead compensator 7. is

$$\frac{E_2(s)}{E_1(s)} = \frac{1}{a} \frac{(1+aTs)}{1+Ts}, \quad a > 1$$
  
Zero is at  $s = -\frac{1}{aT}$ ;  
Pole is at  $s = -\frac{1}{T}$   
choice (a) is correct, since  $a > 1$ .

Hence

$$8. \quad a - \mathbf{R}, \qquad b - \mathbf{S}$$

9.

the choice (a) is correct, since 
$$a > b - S_{c} - Q$$
,  $d - P$   

$$\frac{C(s)}{R(s)} = \frac{2(s-1)}{(s+2)(s+1)};$$

$$R(s) = \frac{1}{s}$$
nnce  $C(s) = \frac{2(s-1)}{(s+1)(s+2)}$ 

Hence

$$= \frac{k_1}{s} + \frac{k_2}{s+1} + \frac{k_3}{s+3},$$

expanding in partial fractions

$$k_{1} = \frac{2(s-1)}{(s+1)(s+2)}|_{s=0} = \frac{-2}{2} = -1$$

$$k_{2} = \frac{2(s-1)}{s(s+2)}|_{s=-1} = \frac{2(-2)}{-1(1)} = 4$$

 $k_3 = \frac{2(s-1)}{s(s+2)}|_{s=-2} = \frac{2(-3)}{-2(-1)} = -3$ Hence  $C(s) = \frac{1}{s} + \frac{4}{s+1} - \frac{3}{s+2}$ and the output  $c(t) = [-1 + 4e^{-t} - 3e^{-2t}]u(t)$ . **10.** The transfer function *G*(*s*) is :  $G(s) = C(sI - A)^{-1} B$  $(sI - A) \begin{vmatrix} s & 0 \\ 0 & s \end{vmatrix} - \begin{vmatrix} -4 & -1 \\ 3 & -1 \end{vmatrix} = \begin{vmatrix} s+4 & 1 \\ -3 & s+1 \end{vmatrix}$  $(sI - A) = \frac{\begin{bmatrix} s+1 & 3\\ -1 & s+4 \end{bmatrix}}{s^2 + 5s + 4 + 3} = \frac{\begin{bmatrix} s+1 & -1\\ 3 & s+4 \end{bmatrix}}{s^2 + 5s + 7}$ Hence G(s)  $\frac{[10] \begin{bmatrix} s+1 & 3\\ -1 & s+4 \end{bmatrix}}{s^2 + 5s + 7} \begin{bmatrix} 1\\ 1 \end{bmatrix}$  $= \frac{[s+1-1]}{s^2+5s+7} \begin{bmatrix} 1\\1 \end{bmatrix} = \frac{s}{s^2+5s+7}$ 11.  $\mathbf{C}(t) ~=~ -te^{-t} + 2e$  $(t \geq 0)$  $C(s) = -\frac{1}{(s+1)^2} + \frac{2}{s+1^2} = \frac{2s+1}{(s+1)^2}$  $C(s) = \frac{G(s)}{1+G(s)},$  $G(s) = \frac{C(s)}{1 - C(s)} = \frac{\frac{2s + 1}{(s+1)^2}}{1 - \frac{2s + 1}{(s+1)^2}} = \frac{2s + 1}{s^2}$  $\frac{\mathcal{C}(s)}{\mathcal{R}(s)} = \frac{\mathcal{G}(s)}{1 + \mathcal{G}(s)},$ 12.  $\frac{G(s)}{1 - G(s)} = \frac{\frac{1}{s(s+1)}}{1 + \frac{1}{c(s+1)}} = \frac{1}{s^2 + s + 1}$  $\omega_n = 1,$   $\omega_n = \frac{1}{2},$   $\zeta = \frac{1}{2}$   $M_p = e^{-\xi\sqrt{1-\zeta^2}} = e^{(-\pi/2)/\sqrt{1-(1/4)}}$  $= e^{-\pi/\sqrt{3}} = 0.163$  $\frac{C(s)}{R(s)} = \frac{1}{1+s}, \qquad \frac{C(j\omega)}{R(j\omega)} = \frac{1}{1+j\omega}$ 14. From  $r(t) \sin t$ ,  $\omega = 1$ ,  $\left| \frac{C(j\omega)}{R(j\omega)} \right| = \frac{1}{\sqrt{1+1}} = \frac{1}{\sqrt{2}}$  $\left| \frac{\mathcal{C}(j\omega)}{\mathcal{R}(j\omega)} \right| = -\tan^{-1} 1 = -\frac{\pi}{4}$  $\therefore$  SS value of  $c(t) = \frac{1}{\sqrt{2}} \sin\left(t - \frac{\pi}{4}\right)$ 

#### 6.28 Control Systems

15.

$$\frac{C(s)}{R(s)} = \frac{s(s+2)}{1 + \frac{K}{s(s+2)}(1+as)}$$
  
=  $\frac{K}{\frac{K}{s(s+2) + K(1+as)}}$   
=  $\frac{K}{\frac{K}{s^2 + s(2+Ka) + K}}$   
K =  $\omega_n^2 = 16$   
 $2\zeta\omega_n = 2 + Ka = 2 \times 0.7 \times 4 = 5.6$   
 $\therefore \qquad a = \frac{5.6 - 2}{16} = 0.225$ 

Κ

16. Step response is integral of unit impulse response

$$\begin{split} u(t) &= \int_{o}^{t} (-4e^{-t} + 6e^{-2t}) \, dt \\ &= [4e^{-t} - 3e^{-2t}]_{0}^{t} \\ &= 4e^{-t} - 3e^{-2t} - 1. \end{split}$$

18. Linear time invariant control system is shown as below.



$$\Rightarrow$$
  $C_1(t) = C(t) u$ 

- 19. For a linear control system with no poles in R. H. S. of S-plane including roots on  $j\omega$  axis with bounded input, the output may be unbounded.
- **20.** System becomes fast and hence  $t_r$  is less
- **21.** Solution exists Ax = b, b should be in column space of A.

22. 
$$A = \begin{bmatrix} -2 & 2 & -3 \\ 2 & 1 & -6 \\ -1 & -2 & 0 \end{bmatrix}; \quad X_{1} = \begin{bmatrix} 1 \\ 2 \\ -1 \end{bmatrix}$$
  
From (A-\lambda I) X<sub>1</sub> = 0
$$\begin{bmatrix} -2 - \lambda & 2 & -3 \\ 2 & 1 - \lambda & -6 \\ -1 & -2 & -\lambda \end{bmatrix} \begin{bmatrix} 1 \\ 2 \\ -1 \end{bmatrix} = 0$$
  
or 
$$\begin{bmatrix} -2 - \lambda + 4 + 3 \\ 2 + 2\lambda + 6 \\ -1 - 4 + \lambda \end{bmatrix} = 0$$
  
or 
$$\lambda = 5$$

23. Sum of eigen values of A

$$2 + 1 + 3 + 4 = 10$$

**24.** Let output of summer be K(S)

$$K(S) = \frac{C(S)}{G_2G_3}$$
  
∴  $\frac{C(S)}{G_2G_3} = G_1 \left[ R(S) - \frac{C(S)H_1}{G_3} \right] - C(S)H_2$   
∴  $C(S) [1 + H_1 G_1 G_2 + H_2 G_2 G_3]$   
 $= G_1 G_2 G_3 R(S).$ 

 $\overline{7}$ 

25. Routh Hurwitz interion

| $\mathbf{S}^4$ | 2  | 3 |  |
|----------------|----|---|--|
| $\mathbf{S}^3$ | 1  | 5 |  |
| $\mathbf{S}^2$ | _7 | 7 |  |
| $\mathbf{S}^1$ | 6  |   |  |
| $\mathbf{S}^0$ | 7  |   |  |

Since sign changes twice, therefore 2 roots in RHP. ~ 1 ~ 1 о г. 01

**26.** det 
$$A = |A| = 5 [3 - 0] - 0 [0 - 2] + 2 [0 - 6]$$
  
=15 - 12 = 3

$$\begin{vmatrix} 3 & 0 \\ 0 & 1 \end{vmatrix} - \begin{vmatrix} 0 & 2 \\ 0 & 1 \end{vmatrix} \begin{vmatrix} 0 & 2 \\ 3 & 0 \end{vmatrix}$$
  
Adj  $A = -\begin{vmatrix} 0 & 0 \\ 2 & 1 \end{vmatrix} \begin{vmatrix} 5 & 2 \\ 2 & 1 \end{vmatrix} - \begin{vmatrix} 5 & 2 \\ 0 & 0 \end{vmatrix}$   
$$\begin{vmatrix} 0 & 3 \\ 2 & 0 \end{vmatrix} - \begin{vmatrix} 5 & 0 \\ 2 & 0 \end{vmatrix} \begin{vmatrix} 5 & 0 \\ 0 & 3 \end{vmatrix}$$
  
Adj  $A = \begin{bmatrix} 3 & 0 & -6 \\ 0 & 1 & 0 \\ -6 & 0 & 15 \end{bmatrix}$   
$$\therefore \qquad A^{-1} = \frac{AdjA}{|A|} = \frac{1}{3} \begin{bmatrix} 3 & 0 & -6 \\ 0 & 1 & 0 \\ -6 & 0 & 15 \end{bmatrix}$$
  
$$= \begin{bmatrix} 1 & 0 & -2 \\ 0 & 1/3 & 0 \\ -2 & 0 & 5 \end{bmatrix}$$
  
28. 
$$S^{T}_{G} = -\frac{GH}{1+GH}, GH >> 1$$
  
$$= -1 (Sensitivity with change in H)$$
  
$$SG^{T} = \frac{1}{1+GH} (Sensitivity with change in G)$$
  
29. 
$$= \begin{bmatrix} 4(s) \\ -(s) \end{bmatrix}$$

$$\mathbf{T}(s) = \frac{\mathbf{C}(s)}{\mathbf{R}(s)} = \frac{\mathbf{G}(s)}{1 + \mathbf{G}(s)\mathbf{H}(s)}$$

For H(s) = 1 [unity feedback system]

$${\rm E}(s) \,=\, \frac{{\rm R}(s) - {\rm C}(s)}{{\rm R}(s)} \,=\, \frac{1}{1 + {\rm G}(s)}$$

steady-state error,  $G_{ss} = \frac{lt}{s \rightarrow 0} sE(s)$ SR(s)1+

$$= \frac{1}{s \to 0} \frac{1}{1 + G(s)}$$
$$= \frac{1(s + Z_1)(s + Z_2)}{1 + G(s)}$$
[Type -1 sys]

For, G(s) = 
$$\frac{1(s+Z_1)(s+Z_2)}{s(s+P_1)(s+P_2)}$$
 [Type -1 system]

and 
$$R(s) = \frac{1}{s} [step function]$$

G.

 $y(t) = te^{-t}$ 

$$= \frac{lt}{s \to 0} \frac{s}{1+k} = 0 \ [k \text{ is a constant}]$$

30.

Б

$$y(s) = \sharp \left[ y(t) \right] = + \frac{1}{(s+1)^2}$$
$$T(s) = \frac{Y(s)}{R(s)} = \frac{s}{(s+1)^2}$$
$$[R(s) = \frac{1}{s}, \text{unit-step function}].$$

31. Constructing routh- array, we get

3  $5^4$  $\mathbf{2}$ 10  $5^3$ 1  $\mathbf{5}$ 0  $5^2$ -710  $5^1$ 10 457 7 $5^0$ 5200 45

As, there are no. of sign change is the first row is 2, therefore no. of roots in Rh-s plane is two.

32. Damping ratio =0.6, % overshoot

$$\mathbf{M}_{\mathbf{p}} = \left[ e^{-\pi r} / \sqrt{1 - r^2} \right] \times 100 = 10\%$$

**33.** Phase-angles  $\phi = \angle D(j\omega) = \tan^{-1}(0.5\omega) - \tan^{-1}(0.05\omega)$  $\phi$  will be maximum when

$$\frac{d\phi}{d\omega} = 0 = \frac{0.5}{1 + (0.5\omega)^2} - \frac{0.05}{1 + (0.05\omega)^2}$$
  

$$\Rightarrow 10 + 10 (0.05\omega)^2 = 1 + (0.5\omega)^2$$
  

$$\Rightarrow \omega^2 = \sqrt{\frac{9}{0.25 - 0.025}}$$
  

$$= \sqrt{40}$$
  

$$\omega = 6.32 \text{ rad/sec.}$$
  
and,  $\phi = \tan^{-1} (6.32 \times 0.5) - \tan^{-1} (0.05 \times 6.32)$   

$$= 55^{\circ}$$

**34.** Phase margin :

$$\phi = 180^{\circ} - \tan^{-1}\omega - \tan^{-1}0.5\omega - 90^{\circ}$$
$$= 180^{\circ} - \tan^{-1}(0.466) - \tan^{-1}(0.233) - 90^{\circ}$$
$$= 51.9^{\circ}$$

**35.** For  $\xi = 0.3$ , value of K comes out a complex number, so it doesn't exist.

$$a = + 1.42$$
GM (gain margin) = -20 log a
$$y' -270^{\circ}$$

$$\omega = \overline{y}$$

$$\omega = 0$$

$$-180^{\circ} -1.42$$

$$\omega = 0$$

$$-90^{\circ}$$
Since
$$a > 1$$

$$a > 1,$$

So GM will be – ve and system unstable,

 $\mathbf{G}(j\omega) \mathbf{H}(j\omega) = \frac{1 - j\omega T_1}{j\omega (1 + j\omega T_2)}$ Now, [assuming one pole in RHS plane].  $\angle G(j\omega) H(j\omega) = -90^{\circ} - \tan \omega T - \tan \omega T_{2}$  $\omega = 0, \angle G(j\omega) H(j\omega) = -90^{\circ}$ At

 $\omega = \infty, \angle G(j\omega) H(j\omega) = -270^{\circ}$ At both the conditions satisfies by polor plot given. 1 7

$$\dot{\mathbf{X}} = \begin{bmatrix} -3 & 1\\ 0 & -2 \end{bmatrix} \mathbf{X} = \mathbf{A}\mathbf{X}$$

Its solution is,

Nor,

37.

$$\begin{split} \mathbf{X}(t) &= \begin{bmatrix} \pounds^{-1} \begin{bmatrix} s\mathbf{I} - \mathbf{A} \end{bmatrix}^{-1} \end{bmatrix} \mathbf{X}(0) \\ &= \pounds^{-1} \begin{bmatrix} s+3 & -1 \\ 0 & s+2 \end{bmatrix}^{-1} \begin{bmatrix} 10 \\ -10 \end{bmatrix} \\ &= \begin{bmatrix} \pounds^{-1} \frac{\begin{bmatrix} s+2 & 1 \\ 0 & s+3 \end{bmatrix}}{(s+2)(s+3)} \end{bmatrix} \mathbf{X}(0) \\ &= \begin{bmatrix} \pounds^{-1} \frac{1}{s+3} & \pounds^{-1} \begin{bmatrix} \frac{1}{s+2} \cdot \frac{1}{s+3} \end{bmatrix} \\ 0 & \pounds^{-1} \begin{bmatrix} \frac{1}{s+2} \end{bmatrix} \end{bmatrix} \mathbf{X}(0) \\ &= \begin{bmatrix} e^{-3t} & e^{-2t} - e^{-3t} \\ 0 & e^{-2t} \end{bmatrix} \begin{bmatrix} 10 \\ -10 \end{bmatrix} \\ \mathbf{X}_{ss} &= \lim_{t \to \infty} \mathbf{X}(t) \end{split}$$

$$= \lim_{t \to \infty} \begin{bmatrix} 20e^{-3t} - 10e^{-2t} \\ -10e^{-2t} \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \end{bmatrix}$$

#### 6.30 Control Systems

**38.** Inductance of the circuit/km,

L = 4 × 10<sup>-7</sup>ln 
$$\frac{D}{r}$$
 mH/km  
= 4 × 10<sup>-7</sup>ln  $\frac{1}{0.7788 \times 5 \times 10^{-3}}$   
= 2.381 mH/km.  
∴ Inductance = 2.381 mH/km × 10  
= **23.81 mH**

**39.** Since at  $\omega = 2$  and  $\omega = 25$ , slope changes from 40 dB/dec to -60 dB/dec at both the valve. So there are poles at  $\omega = 2$  and  $\omega = 25$ . Also at  $\omega = 0.1$  slope is  $-40 \, dB$ /dec. It means, there is two poles at origin. Hence the transfer function, should be of the form,

|      | $\mathbf{T}(s) = \frac{\mathbf{K}(s+5)}{\mathbf{K}(s+5)}$ |
|------|---|
|      | $\Gamma(s) = \frac{1}{s^2(s+2)(s+25)}$                    |
|      | 5K  |
| Now, | $54 = 20 \log \overline{(0.1)^2 \times 50}$               |
| or   | $\mathbf{K}=50.$  |
| Thus | $T(s) = \frac{50(s+5)}{1-1}$                              |
| Inus | $r(s) = \frac{1}{s^2}(s+2)(s+25)$                         |

# Common Data Q. (40 - 42)

 $G(s) = \frac{10,000}{s(s+10)^2}$ **40.** Given,

$$\begin{split} {\rm H}(s) &= 1 \\ {\rm G}\left(s\right) {\rm H}\left(s\right) &= \frac{10,000}{{\rm S}\left({\rm S}+10\right)^2} \\ {\rm G}\left(j\omega\right) {\rm H}\left(j\omega\right) &= \frac{10,000}{j\omega\left(j\omega+10\right)^2} \end{split}$$

(a)  $G(j\omega)$  in decibels

= 20 log | G (j
$$\omega$$
) H (j $\omega$ ) | <sub>$\omega = 20$</sub>   
= 20 log  $\frac{10,000}{\{|20j||10+20j|^2\}}$   
= 20 log  $\frac{10,000}{10000}$  = 0 db.

**41.** First, gain cross over frequency,  $\omega = \omega_1$  should be calculated.

By definition,

As, 
$$|G(j\omega) H(j\omega)|_{\omega=\omega_1} = 1$$
  

$$\Rightarrow \frac{10,000}{|j \omega_1 | 10 + j\omega_1|^2} = 1$$

$$\Rightarrow 10,000 = \omega_1 (\omega_1^2 + 100)$$
Gives  $\omega_1 = 20$  rad/sec.  
Now, phase margin,

 $\phi = \angle \mathbf{G} (j\omega) \mathbf{H} (j\omega) \mid_{\omega = \omega_1} + 180^{\circ}$  $= -90^{\circ} - 2 \tan^{-1} \frac{\omega_1}{10} + 180^{\circ} = -36.86^{\circ}$ 

42. For, gain margin, first calculate phase-cross over frequency,  $\omega = \omega_2$ .

As per definition,  $\angle G(j\omega) H(j\omega)|_{\omega=\omega_2} = -180^{\circ}$ 

$$\Rightarrow -90^{\circ} - 2 \cdot \tan^{-1} \frac{\omega_2}{10} = -180^{\circ}$$

$$\Rightarrow \tan^{-1} \frac{\omega_2}{10} = 45^{\circ}$$
or
$$\alpha_2 = 10 \text{ rad/sec}$$
Now, Gain margin, GM = -20 log a
where
$$a = |G(j\omega) H(j\omega)|_{\omega = \omega_2}$$

$$\therefore \quad GM = -20 \log \frac{10,000}{\omega_2 (\omega_2^2 + 100)}$$

$$= -20 \log \frac{10,000}{10,200}$$

$$= -13.97 \text{ dB}$$

$$\dot{\chi} = \begin{bmatrix} 2 & 3\\ 0 & 5 \end{bmatrix} X + \begin{bmatrix} 1\\ 0 \end{bmatrix} u$$

$$B = \begin{bmatrix} 1\\ 0 \end{bmatrix}, \quad A = \begin{bmatrix} 2 & 3\\ 0 & 5 \end{bmatrix}$$

$$AB = \begin{bmatrix} 2 & 3\\ 0 & 5 \end{bmatrix} \begin{bmatrix} 1\\ 0 \end{bmatrix} = \begin{bmatrix} 2\\ 0 \end{bmatrix}$$

$$\Rightarrow \quad [B:AB] = \begin{bmatrix} 1 & 2\\ 0 & 0 \end{bmatrix}$$

This is not a  $2 \times 2$  matrix, hence system is uncontrollable

Now 
$$[sI-A] = \begin{bmatrix} s & 0 \\ 5 & s \end{bmatrix} - \begin{bmatrix} 2 & 3 \\ 0 & 5 \end{bmatrix}$$
$$= \begin{bmatrix} s-2 & -3 \\ 0 & s-5 \end{bmatrix} = (s-2)(s-5)$$

Two roots on positive half of the y-axis, hence unstable

47.

Angle of asymptote,  $\phi_{A} = \frac{\pm (2q+1)180}{2-0} = \pm 90$ 

Hence root locus can be

 $G(s) = \frac{K}{s^2}$ 



50.

ice root locus can be  

$$\dot{X} = \begin{bmatrix} 2 & 0 \\ 0 & 4 \end{bmatrix} X + \begin{bmatrix} 1 \\ 1 \end{bmatrix} u; \quad Y = \begin{bmatrix} 4 & 0 \end{bmatrix} X$$

$$[sI - A] = \begin{bmatrix} s - 2 & 0 \\ 0 & s - 4 \end{bmatrix}$$

$$[sI - A]^{-1} = \phi(s) \begin{bmatrix} \frac{1}{s - 2} & 0 \\ 0 & \frac{1}{s - 4} \end{bmatrix}$$

$$Y(s) = \phi(s) C(s) \cdot X(s)$$

$$\frac{Y(s)}{X(s)} = \begin{bmatrix} \frac{1}{s-2} & 0\\ 0 & \frac{1}{s-4} \end{bmatrix} [4, 0] = \frac{4}{s-2}$$
  
With input as  $\delta(t)$ , L  $[\delta(t)] = 1$   
 $Y(s) = \frac{4}{s-2} \cdot [1] = \frac{4}{s-2}$   
 $\Rightarrow \qquad Y(t) = 4 e^{2t}$ 

 $Y(t) = 4 e^{2t}$ 

52. Taking Laplace-transform of both-sides, we have

$$[s^{2} + 6s \times 5] X(s) = 12 \left(\frac{1}{s} - \frac{1}{s+2}\right) = \frac{24}{s(s+2)}$$
  
or 
$$X(s) = \frac{24}{s(s^{2} + 6s + 5)(s+2)}$$

then, 
$$\lim_{t \to \infty} x(t) = \lim_{s \to 0} sX(s)$$

$$= \lim_{s \to 0} \frac{24}{(s^2 + 6s + 5)(s + 2)} = 2.4$$

b

53.

$$\theta = \tan^{-1} \frac{\omega}{a} - \tan^{-1} \frac{\omega}{b}$$
  
For lead compensator,  $\theta > 0$   
then,  $b > a$ 

then, **54.** The state of the system at time *t* is.

$$= \begin{bmatrix} e^{-2t} & 0 \\ 0 & e^{-t} \end{bmatrix} \begin{bmatrix} 2 \\ 3 \end{bmatrix} = \begin{bmatrix} 2e^{-2t} \\ 3e^{-t} \end{bmatrix}$$
  
At  $t = 1, X(1) = \begin{bmatrix} 2e^{-2} \\ 3e^{-1} \end{bmatrix} = \begin{bmatrix} 0.271 \\ 1.100 \end{bmatrix}$ 

55.

$$\begin{aligned} \mathbf{X}(t) &= [s\mathbf{I} - \mathbf{A}]^{-1}\mathbf{X}(0) = \phi(t)\mathbf{X}(0) \\ &= \begin{bmatrix} e^{-2t} & 0 \\ 0 & e^{-t} \end{bmatrix} \begin{bmatrix} 2 \\ 3 \end{bmatrix} = \begin{bmatrix} 2e^{-2t} \\ 3e^{-t} \end{bmatrix} \\ \mathbf{At} \quad t = 1, \mathbf{X}(1) &= \begin{bmatrix} 2e^{-2} \\ 3e^{-1} \end{bmatrix} = \begin{bmatrix} 0.271 \\ 1.100 \end{bmatrix} \end{aligned}$$
  

$$\begin{aligned} \mathbf{At} \quad t = 1, \mathbf{X}(1) &= \begin{bmatrix} 2e^{-2} \\ 3e^{-1} \end{bmatrix} = \begin{bmatrix} 0.271 \\ 1.100 \end{bmatrix} \end{aligned}$$
  

$$\begin{aligned} \mathbf{At} \quad t = 1, \mathbf{X}(1) &= \begin{bmatrix} 2e^{-2} \\ 3e^{-1} \end{bmatrix} = \begin{bmatrix} 10.271 \\ 1.100 \end{bmatrix} \end{aligned}$$
  

$$\begin{aligned} \mathbf{At} \quad t = 1, \mathbf{X}(1) &= \begin{bmatrix} 2e^{-2t} \\ 3e^{-1} \end{bmatrix} = \begin{bmatrix} 10.271 \\ 1.100 \end{bmatrix} \end{aligned}$$
  

$$\begin{aligned} \mathbf{At} \quad t = 1, \mathbf{X}(1) &= \begin{bmatrix} 2e^{-2t} \\ 3e^{-1} \end{bmatrix} = \begin{bmatrix} 10.271 \\ 1.100 \end{bmatrix} \end{aligned}$$
  

$$\begin{aligned} \mathbf{At} \quad t = 1, \mathbf{X}(1) &= \begin{bmatrix} 2e^{-2t} \\ 3e^{-1} \end{bmatrix} = \begin{bmatrix} 10.271 \\ 1.100 \end{bmatrix} \end{aligned}$$
  

$$\begin{aligned} \mathbf{At} \quad t = 1, \mathbf{X}(1) &= \begin{bmatrix} 2e^{-2t} \\ 3e^{-1} \end{bmatrix} = \begin{bmatrix} 10.271 \\ 1.100 \end{bmatrix} \end{aligned}$$
  

$$\begin{aligned} \mathbf{At} \quad t = 1, \mathbf{X}(1) &= \begin{bmatrix} 2e^{-2t} \\ 3e^{-1} \end{bmatrix} = \begin{bmatrix} 0.271 \\ 1.100 \end{bmatrix} \end{aligned}$$
  

$$\begin{aligned} \mathbf{At} \quad t = 1, \mathbf{X}(1) &= \begin{bmatrix} 2e^{-2t} \\ 3e^{-1} \end{bmatrix} = \begin{bmatrix} 0.271 \\ 1.100 \end{bmatrix} \end{aligned}$$
  

$$\begin{aligned} \mathbf{At} \quad t = 1, \mathbf{X}(1) &= \begin{bmatrix} 2e^{-2t} \\ 3e^{-1} \end{bmatrix} = \begin{bmatrix} 0.271 \\ 1.100 \end{bmatrix} \end{aligned}$$
  

$$\begin{aligned} \mathbf{At} \quad t = 1, \mathbf{X}(1) &= \begin{bmatrix} 2e^{-2t} \\ 3e^{-1} \end{bmatrix} = \begin{bmatrix} 0.271 \\ 1.100 \end{bmatrix} \end{aligned}$$
  

$$\begin{aligned} \mathbf{At} \quad t = 1, \mathbf{X}(1) &= \begin{bmatrix} 2e^{-2t} \\ 3e^{-1} \end{bmatrix} = \begin{bmatrix} 0.271 \\ 1.100 \end{bmatrix} \end{aligned}$$
  

$$\begin{aligned} \mathbf{At} \quad t = 1, \mathbf{X}(1) &= \begin{bmatrix} 2e^{-2t} \\ 3e^{-1} \end{bmatrix} = \begin{bmatrix} 0.271 \\ 1.100 \end{bmatrix} \end{aligned}$$
  

$$\begin{aligned} \mathbf{At} \quad \mathbf{At} \quad \mathbf{X}(1) &= \begin{bmatrix} 2e^{-2t} \\ 3e^{-1} \end{bmatrix} = \begin{bmatrix} 0.271 \\ 1.100 \end{bmatrix} \end{aligned}$$
  

$$\begin{aligned} \mathbf{At} \quad \mathbf{At} \quad \mathbf{At} \quad \mathbf{At} \\ \mathbf{At} \quad \mathbf{At} \quad \mathbf{At} \quad \mathbf{At} \\ \mathbf{At} \quad \mathbf{At} \quad \mathbf{At} \quad \mathbf{At} \\ \mathbf{At} \quad \mathbf{At} \quad \mathbf{At} \\ \mathbf{At} \quad \mathbf{At} \quad \mathbf{At} \quad \mathbf{At} \quad \mathbf{At} \quad \mathbf{At} \\ \mathbf{At} \quad \mathbf{At$$

or 
$$1 + \frac{45}{(s+15)(s+1)} = 0$$
  
or  $s^2 + 16s + 60 = 0$   
or  $(s+6)(s+10) = 0$   
Thus,  $s = -6$ ,  $s = -10$  are the roots.

58. 
$$\frac{d^2\omega}{dt^2} = \frac{B}{J}\frac{d\omega}{dt} - \frac{K^2}{LJ}\omega + \frac{K}{LJ}V_a \qquad ...(i)$$
$$\frac{d\omega}{dt} = 1. \frac{d\omega}{dt} + 0.\omega \qquad ...(ii)$$
From equations (i) and (ii) we get

 $(\iota)$  and  $(\iota)$ , we get

$$\begin{bmatrix} \frac{d^2 \omega}{dt^2} \\ \frac{d \omega}{dt} \end{bmatrix} = \begin{bmatrix} -B/J & -K^2/LJ \\ 1 & 0 \end{bmatrix} \begin{bmatrix} \frac{d \omega}{dt} \\ \omega \end{bmatrix} + \begin{bmatrix} \frac{K}{LJ} \\ 0 \end{bmatrix} V_a$$

**59.** The characteristics equation is, 1 + GH = 0 $s^3 + 6s^2 + 8s + k = 0$ 

Constructing routh-Hurwitz array

$$s^{3} \begin{vmatrix} 1 & 8 & 0 \\ s^{2} & 6 & k \\ s^{1} & k - 48 & - \\ k \end{vmatrix}$$
  
The system will be just unstable  
 $k - 48 = 0$  or  $k = 48$ .  
**60.**  $\angle GH \mid_{\omega = 0.5a} = -\tan^{-1} \frac{\omega}{a} = -\tan^{-1} 0.5 = 26.56^{\circ}$   
From the plot,  
at  $\omega = 0$ .  $\angle GH = 22.5^{\circ}$   
Then error  $= -22.50^{\circ} (-26.56^{\circ}) = 4.06^{\circ}$   
At  $\omega = 0.5 a$ , 20 log | GH |  $= 20 \log k - 20 \log | 1 + j 0.5 | = 20 \log k - 0.97$   
From the plot, 20 log | GH |  $= 20 \log k$   
Then, error in dB gain = **0.97 dB**

9

**61.** Forward path gain, 
$$P_1 = \frac{2}{a^2}$$

For ward path gain,  $P_1 = \frac{1}{s^2}$   $\mu \Delta_1 = 1$ Individual loop gains  $= \left(\frac{-3}{s}\right), \left(\frac{12}{s}\right), \left(\frac{18}{s^2}\right)$   $P_{12}$  (Non-touching loop gain)  $= 36/s^2$ Then, T =  $\frac{P_1 \Delta_1}{1 - P_{11} - P_{21} - P_{23} + P_{12}}$ =  $\frac{2./s^2}{1 + \frac{15}{s} + \frac{18}{s^2} + \frac{36}{s^2}} = \frac{2}{s^2 + 15s + 54}$  $= \frac{2}{(s+6)(s+9)} = \frac{1}{27\left(1+\frac{s}{6}\right)\left(1+\frac{s}{9}\right)}$ 

#### 6.32 Control Systems

- **62.** Phase margin,  $\phi = \angle G(j\omega) H(j\omega) |_{\omega = \omega_1} + 180^\circ$ =  $-180^\circ + 180^\circ = 0^\circ$ 
  - [Since, at point (-1, j0),  $\angle G(j\omega)$ ]. H $(j\omega) \mid_{\omega = \omega_1} = 180^{\circ}$
- **63.** Initial value =  $\lim_{s \to \infty} sF(s) = \lim_{s \to \infty} \frac{5}{s^2 + 3s + 2} = 0$
- **64.** Transfer function is derived as,

$$K_t \cdot \frac{d\theta}{dt} = e(t)$$
$$sK_t \cdot \theta(s) = E(s)$$
$$\frac{E(s)}{\theta(s)} = sK_t$$

65. Constructing Routh-array,

*:*..

Number of sign changes in the first column is two, therefore the number of roots in the left half *s*-plane is 2.

66. Constructing signal-flow graph as shown here,



Number of forward paths is 3 having path gains,

$$P_1 = \frac{1}{s^2},$$
$$P_2 = 1,$$
$$P_3 = \frac{1}{s}$$

There are no loops, then, transfer function is given as,

$$T = \frac{P_1 \Delta_1 + P_2 \Delta_2 + P_3 \Delta_3}{\Delta} = \frac{1}{s^2} + 1 + \frac{1}{s}$$
$$= \frac{s^2 + s + 1}{s^2} \qquad [\Delta_1 = \Delta_2 = \Delta_3 = \Delta = 1]$$

67. Characteristic equation for this system is,

|sI - A| = 0or  $\begin{vmatrix} s & -2 \\ -2 & s \end{vmatrix} = 0$ or  $s^{2} - 4 = 0$ or  $s = \pm 2.$ 

68. 
$$T(s) = \frac{C(s)}{R(s)} = \frac{k/s(s+2)}{1 + \frac{k(1+sP)}{s(s+2)}} = \frac{k}{s^2 + (2+Pk)s + k}$$
  
Now,  $k = \omega_n^2$ ,  
and  $(Pk+2) = 2\pi\omega_n$ 

Then, 
$$k = 25$$
  
and  $P = 0.2$ 

**69.** 
$$C(s) = \frac{12.5 \times 8}{(s+6)^2 + 8^2} = \frac{100}{s^2 + 12s + 100}$$

Then, Output for unit step respoure

$$= \frac{1}{s} \cdot \frac{100}{s^2 + 12s + 100}$$

and steady-state value = 
$$\lim_{s \to 0} sC(s)$$

$$= \lim_{s \to 0} \frac{100}{s^2 + 12s + 100} = 1$$

70. 
$$Y(s) = X(s) \cdot \frac{s}{s+1}$$
  
 $Y(s) = \frac{1}{(s^2+1)} \cdot \frac{s}{s+1}$   
 $= \frac{A}{(s+j)} + \frac{B}{(s-j)} + \frac{C}{(s+1)}$   
 $= \frac{1}{2(1+j)} \left(\frac{1}{s-j}\right) + \frac{1}{2(1-j)(s+j)} - \frac{1}{2(s+1)}$ 

Taking inverse Laplace transformation, we have

$$y(t) = \frac{1}{2(1+j)}e^{jt} + \frac{1}{2(1-j)}e^{-jt} - \frac{1}{2}e^{-t}$$
$$= \frac{1}{2}(\sin t + \cos t) - \frac{1}{2}e^{-t}$$
$$= \frac{1}{\sqrt{2}}\sin(t + 45^\circ) - \frac{1}{2}e^{-t}$$

Then, steady state value =  $\lim_{t \to \infty} y(t)$ 

$$= \frac{1}{\sqrt{2}} \sin(t + 45^\circ).$$

1

**71.** The value of  $\omega$  at which  $|G(j\omega)| = 1$  is given by,

$$= \frac{\sqrt{(\omega_c a)^2 + 1}}{\omega_c^2} \qquad .....(i)$$



72. 
$$|T(j\omega)| = \frac{|T(j\omega)^2 + 4|}{(j\omega + 1)(j\omega + 4)} = 0$$
  
 $\Rightarrow -\omega^2 + 4 = 0,$   
 $\Rightarrow \omega = 2 \text{ rad/sec}$ 

**73.**  $G(s) H(s) = \frac{K}{s^3}$ 

Characteristic equation is,

1 + G(s) H(s) = 0  $\Rightarrow \qquad s^3 + K = 0$   $\Rightarrow \qquad \frac{dK}{ds} = 0$   $\Rightarrow \qquad 3s^2 = 0$  $\Rightarrow \qquad s = 0, 0$ 

**Note :** In all other options, all breaking points are not at origin.

74. 
$$G(j\omega) = \frac{(j\omega+1)}{(j\omega)^2} = \frac{1+j\omega}{-\omega^2} = -\frac{1}{\omega^2} - \frac{1}{\omega}$$
  
For Gain Margin, complex part of  $G(j\omega) = 0$   
$$\Rightarrow \qquad \frac{1}{\omega} = 0$$

$$\therefore \text{ Gain margin} = 20 \log \frac{1}{|G(j\omega)|} = 20 \log \frac{1}{\left|\frac{1}{\infty}\right|} = \infty$$

75. 
$$1 + G(s)H(s) = 0$$
  
 $\Rightarrow (1-K)s + (1+K) = 0$   
 $\Rightarrow s'(1-K) 0 \ 1-K > 0, \ 1+K > 0$   
So  $|K| < 1$ 

76.  

$$R(s) \xrightarrow{+} M(s) \xrightarrow{2} Y(s)$$

$$M(s) = R(s) + [R(s) - Y(s)] \frac{3}{s}$$

and

77.

$$\begin{array}{l} \therefore \qquad \frac{Y(s)}{R(s)} = \frac{2(s+3)}{s^2+2s+6} \\ E(s) = R(s) - Y(s) = \left[1 - \frac{2(s+3)}{s^2+2s+6}\right] \\ E(s) = R(s) \frac{s^2}{s^2+2s+6} \\ e_{ss} = \lim_{s \to 0} E(s) = 0 \\ G(s) H(s) = \frac{\pi e^{-0.25s}}{s} \\ \pi \left[\cos(0.25\omega) - j\sin(0.25\omega)\right] \end{array}$$

 $\mathbf{Y}(s) = \frac{2}{s+2} \left[ \mathbf{R}(s) \left( 1 + \frac{3}{s} \right) - \frac{3}{s} \mathbf{Y}(s) \right]$ 

$$G(js) H(j\omega) = \frac{n[\cos(0.25\omega) - j\sin(0.25\omega)]}{j\omega}$$
$$= \frac{-\pi}{\omega}\sin(0.25\omega) - j\frac{\pi}{\omega}\cos(0.25\omega)$$

Imaginary part = 0  

$$\therefore \quad \frac{\pi}{\omega} \cos(0.25\omega) = 0$$

$$\Rightarrow \qquad \frac{\omega}{4} = \frac{\pi}{2}, \ \omega = 2\pi$$

$$\therefore \quad \left| G(j\omega) H(j\omega) \right|_{\omega = 2\pi}$$

$$= \left| -\frac{\pi}{2\pi} \sin\left(\frac{2\pi}{4}\right) \right| = \left| -\frac{1}{2} \right| = 0.5$$
78. Here,  

$$T = \frac{Y(s)}{R(s)} = \frac{K + 0.366s}{s(s+1)}$$
Since,  

$$P.M. = 180^{\circ} + \angle G(j\omega) + (j\omega)$$

$$60^{\circ} = 180^{\circ} + \tan^{-1}\left(\frac{0.366\omega}{K}\right) - 90^{\circ} - \tan^{-1}\omega$$

$$\Rightarrow \quad \tan^{-1}\left(\frac{0.366\omega}{K} - \omega}{1 + \frac{0.366\omega}{K}}\right) = -30^{\circ}$$

$$\Rightarrow \quad \frac{0.366\omega - \omega K}{K + 0.366\omega^2} = -\frac{1}{\sqrt{3}}$$
At cross-over frequency of 1 rad/sec,  
we get K + 0.366 = 1.732 K - 0.634
$$\therefore \qquad K = 1.366$$

The eigen-values corresponding to triangular matrix are  $\lambda = 3, -2$  and 1.

0 0 1

 $\mathbf{P} = \begin{bmatrix} 3 & -2 & 2 \\ 0 & -2 & 1 \end{bmatrix}$ 

For finding eigen vector, 
$$[A - \lambda 1]\hat{x} = 0$$
  

$$\begin{bmatrix} 3 - \lambda & -2 & 2 \\ 0 & -2 - \lambda & 1 \\ 0 & 0 & 1 - \lambda \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix}$$
For  $\lambda = -2$ ,  

$$\begin{bmatrix} 5 & -2 & 2 \\ 0 & 0 & 1 \\ 0 & 0 & 3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix}$$
or,  $5x_1 - 2x_2 + 2x_3 = 0$   
 $x_3 = 0$   
 $3x_3 = 0$   
If  $x_2 = k$  (Assume),  
 $5x_1 - 2k = 0$ ,  
 $x_1 = \frac{2}{5}k$   
Then eigen vectors  

$$\begin{bmatrix} x_1 \\ x_2 \end{bmatrix} = \begin{bmatrix} 2 \\ 5 \\ 1 \end{bmatrix} = \begin{bmatrix} 2 \\ 5 \end{bmatrix}$$

 $\begin{bmatrix} x_3 \end{bmatrix} = \begin{bmatrix} 0 \end{bmatrix}$ 

0

6.34 Control Systems

80. 
$$F(z) = \frac{1}{z+1} = \frac{z+1-z}{z+1} = 1 - \frac{z}{z-(-1)}$$
  
∴  $z^{-1}[F(z)] = \delta(t) - (-1^n) \qquad \left[ \because z^{-1} \left( \frac{z}{z-a} \right) = a^n \right]$   
81.  $(sI - A)^{-1} = \begin{bmatrix} s & 1 \\ 0 & s+3 \end{bmatrix}^{-1} = \frac{Adj(sI - A)}{|sI - A|}$   
 $= \frac{\begin{bmatrix} s+3 & -1 \\ 0 & s \end{bmatrix}}{\begin{bmatrix} s & 1 \\ 0 & s+3 \end{bmatrix}}$   
 $= \frac{\begin{bmatrix} s+3 & -1 \\ 0 & s \end{bmatrix}}{s(s+3)}$   
 $= \begin{bmatrix} \frac{1}{s} & \frac{-1}{s(s+3)} \\ 0 & \frac{1}{s+3} \end{bmatrix}$   
∴  $\phi(t) = L^{-1}(sI - A)^{-1} = \begin{bmatrix} 1 & \frac{1}{3}(1 - e^{-3t}) \\ 0 & e^{-3t} \end{bmatrix}$   
82. Zero state response  $= \mathcal{L}^{-1}\phi(s)$  B U(s)

$$= \mathcal{L}^{-1} \begin{bmatrix} \frac{1}{s} & \frac{-1}{s(s+3)} \\ 0 & \frac{1}{s+3} \end{bmatrix} \begin{bmatrix} 1 \\ 0 \end{bmatrix} \frac{1}{s}$$
$$= \mathcal{L}^{-1} \begin{bmatrix} \frac{1}{s^2} \\ 0 \end{bmatrix}$$
$$= \begin{bmatrix} t \\ 0 \end{bmatrix}$$

State transition equation = Zero input response + zero state response.

$$X(t) = \phi(X) X(0) + t$$
$$= \begin{bmatrix} -1 + 1 - e^{-3t} \\ 0 + 3e^{-3t} \end{bmatrix} + \begin{bmatrix} t \\ 0 \end{bmatrix}$$
$$= \begin{bmatrix} t - e^{-3t} \\ 3e^{-3t} \end{bmatrix}$$

**83.** It is state space representation using phase variable.

$$\begin{array}{c} {\rm Standard\ form} \left[ {\begin{array}{*{20}c} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -a_n & -a_{n-1} & -a_{n-2} \end{array} \right] \\ {\rm Thus\ ATQ} \quad a_n = 1, \, a_{n-1} \, = -2, \, a_{n-2} \, = 4 \end{array}$$

85. 
$$H(j\omega) = \frac{10^4 (1+j\omega)}{(10+j\omega)(100+j\omega)^2}$$
$$= \frac{10^4 (1+s)}{10 \left(1+\frac{s}{10}\right)(100)^2 \left(1+\frac{s}{200}\right)^2}$$
$$= \frac{0.1 (1+s)}{\left(1+\frac{s}{10}\right) \left(1+\frac{s}{100}\right)^2}$$
∴ Value of K = +20 log<sub>10</sub> (0.1)  
= -20 db.  
So plot will start from -20 db ft.  
87.  $F(s) = s^5 - 3s^4 + 5s^3 - 7s^2 + 4s + 20$ 

We can solve it by making Routh Hurwitz array.

| $s^5$ | 1   | 5    | 4  |
|-------|-----|------|----|
| $s^4$ | -3  | _7   | 20 |
| $s^3$ | 8/3 | 20/3 | 0  |
| $s^2$ | 5   | 20   | 0  |
| $s^1$ | 0   | 0    | 0  |
| $s^0$ | 20  | 0    | 0  |

We can replace 1st element of  $s^1$  by 10.

If we observe 1st column, sign is changing two times, so we have two poles on right half side of imaginary axis and  $5s^2 + 20 = 0$ .

So,  $s = \pm 2j$  and 1 pole on left side of imaginary axis.

- **88.** If GH plot incircles (-1, j0) that is the critical point, then the system becomes unstable. So option 1 is there which does not enclose the (-1, j0) other all are incircling the critical point.
- **89.** Transfer function for  $u_1$



Hence stable

#### Transfer function for $u_2$

$$TF_2 = \frac{\left(\frac{1}{s-1}\right)}{1 + \left(\frac{1}{s-1}\right) \cdot \left(\frac{s-1}{s+2}\right)}$$
$$= \frac{1}{(s+3)(s-1)}$$

Hence unstable, as it has pole at right side of s-plane.

90. Given, G (s) = 
$$\frac{1}{s(s+1)(s+2)}$$
  
G (j\omega) =  $\frac{1}{j\omega(1-j\omega)(2+j\omega)} = \frac{-j(1-j\omega)(2-j\omega)}{\omega(1-\omega^2)(4+\omega^2)}$   
=  $\frac{j(2-3j(\omega-\omega^2)}{\omega(1+\omega^2)(4+\omega^2)} = \frac{-3\omega-j(2-\omega^2)}{\omega(1+\omega^2)(4+\omega^2)}$   
=  $\frac{-3}{(1+\omega^2)(4+\omega^2)} - j \frac{(2-\omega^2)}{\omega(1+\omega^2)(4+\omega^2)}$   
 $\Rightarrow$   $x = \frac{3}{(1+\omega^2)(4+\omega^2)}$ ,  
 $y = -\frac{(2-\omega^2)}{\omega(1+\omega^2)(4+\omega^2)}$ ,  
At  $\omega \to 0, x \to \frac{3}{4}, y \to -\infty$   
92.  $G(z) = z^{-1} + z^{-2}$   
Characteristic equation is  
 $1 + KG(z) = 0$   
 $\Rightarrow 1 + K(z^{-1} + z^{-2}) = 0$   
 $\Rightarrow z^2 + Kz + I = 0$   
Using stability criteria,  
 $|K| < 1$   
i.e.  $-1 < K < 1$   
93. Characteristic equation is  
 $1 + \frac{K(s+3)}{(s+8)^2} = 0$   
 $\Rightarrow s^2 + (16+K) s + 3K + 64 = 0$   
Routh's array,  
 $s^2 = 1 -3K + 64$   
 $s - 16 + K$   
 $s^0 - 3K + 64$   
 $\Rightarrow$  No such K exist to make all element of a row equal to 0.

94. By block diagram, technique reduction method

95. 
$$Z = \sqrt{R^2 + \left(\omega L - \frac{1}{\omega C}\right)^2}$$
$$Y = \frac{1}{\sqrt{R^2 + \left(\omega L - \frac{1}{\omega C}\right)^2}}$$

By rationalization and varying frequency  $\boldsymbol{\omega},$  then locus is



**96.** Steady state value of z

$$= \underbrace{\ell t}_{s \to 0} \operatorname{SE}(s) \frac{k_i}{s}$$

$$= \underbrace{\ell t}_{s \to 0} \frac{s \cdot \frac{1}{s} \cdot \frac{k_i}{s}}{\left[ 1 + \left( k_p + \frac{k_i}{s} \right) \left( \frac{\omega^2}{s^2 + 2s\omega_r s + \omega^2} \right) \right]}$$

$$= \underbrace{\ell t}_{s \to 0} \frac{k_i / s}{1 + \left( k_p + \frac{k_i}{s} \right) \left( \frac{\omega^2}{s^2 + 2s\omega_r s + \omega^2} \right)}$$

$$= 1$$

97. By Clayey – Hamilton theorem, Every square matrix satisfies its own characteristic equation  $\alpha(\lambda) = \lambda^3 + \lambda^2 + 2\lambda + 1 = 0$   $\alpha(P) = P^3 + P^2 + 2P + I = 0$   $\Rightarrow I = -P^3 + P^2 + 2P$ Premultiplying by p<sup>-1</sup>, we get P<sup>-1</sup> = - [P<sup>2</sup> + P + 2I]

**98.** 
$$G(s) = \frac{y(s)}{U(s)} = \frac{1}{s^2 + 3s + 2}$$
  

$$\therefore \qquad y(s) = \frac{1}{s^2 + 3s + 2} \quad U(s)$$
  
But  $U(t) = \delta (t - 1)$   
and  $U(s) = e^{-s}$   

$$\therefore \qquad y(s) = \frac{e^{-s}}{s^2 + 3s + 2}$$

For steady state,

$$y(t)_{t \to \infty} = \operatorname{Lt}_{s \to 0} sy(s)$$
$$= \frac{e^{-s}}{s \to 0} \frac{e^{-s}}{s^2 + 3s + 2}$$
$$= \frac{1}{2} = 0.5$$



∴ lag compensator ∴ lead compensator Hence  $C_1$  is lead compensator and  $C_2$  is lag compensator.

#### Alternately

or

$$\angle \phi \text{ for } G = \tan^{-1} \omega - \tan^{-1} \frac{\omega}{10}$$
$$\angle \phi \text{ for } G = \tan^{-1} \frac{\omega}{10} \omega - \tan^{-1} \omega$$

Hence  $C_1$  is lead compensator and  $C_2$  is lag compensator.

100. Transfer function of given plot is of the form

$$T(s) = \frac{k (1 + s T_1) (1 + s T_2)}{s^2}$$

so, it will have two poles and two zeroes.

First-40 dB/decade slope indicates 2 pole in system. Next-20 dB/decade slope indicates 1 zero is added. Next - dB/decade slope indicates 1 more zero is added.

#### Alternately

At  $\omega = 0.1$ , slope changes from -40 dB/decade to -20 dB/decade, then, zero at  $\omega = 0.1$ 

At,  $\omega > 0.1$ , slope changes from – 20dB/decade to 0 dB/decade, then one more zero at  $\omega > 0.1$ 

Since, for  $\omega < 0.1$ , slope is – 40 dB/decade there is alveag two poles.

Hence (c) is the correct option.

**101.** K > 0

Characteristic equation of the system is

$$1 + G(s) H(s) = 0$$

$$\Rightarrow 1 + \frac{K}{s(s+3)(s+10)} = 0$$
  
$$\Rightarrow s(s^{2} + 13s + 30) + K = 0$$
  
$$\Rightarrow s^{3} + 13s^{2} + 30s + K = 0$$
  
Routh array  
$$s^{3} = 1 \qquad 30$$

$$s^{-1} = 1$$
 30  
 $s^{2} = 13$  K  
 $s^{1} = \frac{390 - K}{13} = 0$   
 $s^{0} = K$ 

For system to be stable, all element of first column must be positive.

$$\therefore \qquad \frac{390 - K}{13} > 0$$
  
$$\Rightarrow \qquad K < 390^{\circ}$$
  
and 
$$\qquad K > 0$$

Hence range of K is 0 < K < 390

102. 
$$M(s) = \frac{100}{s^2 + 20s + 100}$$

Comparing with standard form,

$$\begin{split} M(s) &= \frac{\omega_n^2}{s^2 + 2\xi \omega_n s + \omega_n^2} \\ \text{we get} \quad & 2\xi \omega_n = 20 \\ \Rightarrow \quad & \xi = \frac{20}{2 \times 10} = 1 \\ \text{and} \quad & \omega_n^2 = 100 \\ \Rightarrow \qquad & \omega_n = 10 \end{split}$$

For  $\xi = 1$ , system is critically damped.

103. The relative error of product or division of different quantities is equal to the sum of relative errors of individual quantities.

**105.** Two roots at  $s = \pm j$  and one in left halfs-plane

$$s^{3} 2 2$$

$$s^{4} 4 4$$
Characteristic equation is
$$A(s) = 2s^{3} + 4s^{2} + 2s + 4 = 0$$

$$\Rightarrow s^{3} + 2s^{2} + s + 2 = 0$$

$$\Rightarrow (s + 2) (s^{2} + 1) = 0$$

$$\Rightarrow s = -2, \quad s = \pm j$$
**106.** At  $\omega = 0.1, 2$  poles
At  $\omega = 2, 1$  pole
At  $\omega = 5, 1$  zero
At  $\omega = 25, 1$  pole
$$80dB \underbrace{-40dB/dec}_{0.1} - 60 dB/dec}_{0.1} \underbrace{-40dB/dec}_{0.1} \underbrace{-40 dB/dec}_{0.1} \underbrace{-40$$

$$T(s) = \frac{k(s+5)}{s(s+2)(s+25)}$$
  

$$\therefore \qquad 80 = 20 \log \frac{5k}{(0.1)^2 \times 2 \times 25}$$
  

$$\Rightarrow \qquad 4 = \log \frac{5k}{0.5}$$
  

$$\Rightarrow \qquad 10k = 10^4$$
  

$$\Rightarrow \qquad k = 10^3 = 1000$$

$$\mathbf{T}(s) = \frac{1000(s+5)}{s^2(s+2)(s+25)}$$

*.*..

 $\Rightarrow$ 

*.*..

107. Given : 
$$G(s) = \frac{K}{(s+1)(s+2)}$$
  
From curve, steady state error  $= 1 - 0.75 = 0.25$   
 $\therefore$   $E(s) = \frac{R(s)}{1 + H(s)G(s)} = \frac{1/s}{1 + 1.\frac{k}{(s+1)(s+2)}}$   
 $\therefore$   $e_{ss} = \lim_{s \to 0} sE(s)$   
 $= \lim_{s \to 0} s.\frac{1/s}{1 + \frac{k}{(s+1)(s+2)}} = 0.25$   
 $\Rightarrow \frac{1}{1 + \frac{k}{2}} = 0.25$   
 $\Rightarrow 1 = 0.25 + \frac{0.25k}{2}$   
 $\Rightarrow k = \frac{0.75 \times 2}{0.25} = 6$   
108. Given :  $G(s) = \frac{e^{-0.1s}}{s}$   
 $\Rightarrow G(j\omega) = \frac{e^{-0.1j\omega}}{(j\omega)}$   
Phase crossover frequency :  
 $-90 - 0.1 \omega \times \frac{180}{\pi} = -180$   
 $\Rightarrow \frac{18}{\pi} \omega = 90$   
 $\Rightarrow \omega = 5\pi = 15.7 \text{ rad/sec}$   
Gain margin :  
 $a = |G(j\omega)|_{\omega = 18.7} = \frac{1}{\omega} = \frac{1}{15.7}$   
 $\therefore$  Gain margin = 20 log a  
 $= 20 \log(\frac{1}{15.7}) = -23.9 \text{ dB}.$   
109.  $G(s) = \frac{9}{100}$   
 $R(s) \stackrel{+}{\longrightarrow} O(s) = O(s)$ 

$$\overline{\mathbf{R}(s)} = \overline{\mathbf{1} + \mathbf{G}(s)\mathbf{H}(s)}$$

When G(s) = 100 + 10 = 110,

$$\frac{C(s)}{R(s)} = \frac{110}{1+110 \times \frac{9}{100}} = \frac{1100}{109}$$
$$= 10.091 = 10 + 0.1$$
$$= 10 + 1\% \text{ of } 10 = 10.1$$

When G(s) = 100 - 10 = 90, 90  $\mathbf{C}(s)$ 900  $\frac{\Theta(s)}{R(s)} = \frac{-1}{1+90 \times \frac{9}{100}}$ 91  $= 9.89 \approx 9.9$ =  $10 - 0.1 \neq 10 - 1\%$  of 10 Hence overall system gain =  $10 \pm 1\%$ **110.** Given : G(s) =  $\frac{2}{s+1} = \frac{A}{Ts+1}$  $\therefore$  Time constant, T = 1 For 98%, time required = 4T = 4 sec  $\frac{\mathcal{C}(s)}{\mathcal{R}(s)} = \frac{2}{s+1}$  $\mathcal{C}(s) = \frac{2}{(s+1)} \cdot \frac{1}{s}$ *.*.. *:*..  $\mathbf{C}(s) = 2\left[\frac{1}{s} - \frac{1}{s+1}\right]$  $\Rightarrow$  $C(t) = 2 [1 - e^{-t}] = 2 \times 0.98$  $\Rightarrow$  $1 - e^{-t} = 0.98$  $\Rightarrow$  $e^{-t} = 0.02$  $\Rightarrow$  $t = \ln 0.2 = 3.91 \text{ sec}$  $\Rightarrow$ 111. Tank (2) Tank (1)  $\overline{\mathbf{Q}}_{2}$  $\mathbf{R}_{1}$  $\dot{R}_2$  $\mathbf{Q}_{3}$  $C_1 = capacity of tank(1)$  $C_2 = capacity of tank (2)$  $h_1, h_2$  = heights of water  $Q_1, Q_2$  = flow rates  $R_1, R_2$  = resistances of flow  $\mathbf{Q}_1 = \mathbf{Q}_2 + \mathbf{C}_1 \frac{dh_1}{dt}$  $\mathbf{Q}_2 = \mathbf{Q}_3 + \mathbf{C}_2 \frac{dh_2}{dt}$  $Q_2 = \frac{h_1 - h_2}{R_1}, Q_3 = \frac{h_2}{R_2}$ *.*.. Equivalent circuit :



B, D = resistances

112.

$$G(s) = \frac{1}{s(s+1)(s+2)}$$

$$G(j\omega) = \frac{1}{j\omega(j\omega+1)(j\omega+2)}$$

$$M = \frac{1}{\omega\sqrt{\omega^2+1}\sqrt{\omega^2+4}}$$

$$\angle \phi = -90 - \tan^{-1}\omega - \tan^{-1}\frac{\omega}{2}$$

$$-\frac{3}{4}\omega = \sqrt{2} \operatorname{rad}/\sec$$

$$\int \sqrt{2} \operatorname{rad}/\sec$$
For  $\omega = 0$ ,  $M = \infty$ ;  $\angle \phi = -90$   
For  $\omega = \infty$ ,  $M = 0$ ;  $\angle \phi = -90 - 90 - 90 = -270$   
So Cutting Real Axis

So Cutting Real Axis  
Imaginary part of 
$$G(j\omega) = 0$$
  
i.e. Img  $\{G(j\omega)\} = 0$   
 $\Rightarrow Im\left\{\frac{1}{j\omega(-\omega^2 + 3)\omega + 2}\right\} = 0$   
 $\Rightarrow Im\left\{\frac{1}{-j\omega^3 - 3\omega^2 + 2j\omega}\right\} = 0$   
 $\Rightarrow Im\left\{\frac{1}{-3\omega^2 + j(2\omega - w^3)}\right\} = 0$   
 $\Rightarrow Im\left\{\frac{-3\omega^2 - j(2\omega - \omega^3)}{(3\omega^2)^2 + (2\omega - w^3)^2}\right\} = 0$   
 $\therefore 2\omega - \omega^3 = 0$   
 $\Rightarrow \omega^2 = 2 \cdot \omega = 0$ 

Neglecting  $\omega = 0$ , we have

$$\omega = \sqrt{2}$$
  
$$\therefore \quad \mathbf{M}\Big|_{at \, \omega = \sqrt{2}} = \frac{1}{\sqrt{2} \sqrt{3} \sqrt{2+4}}$$
$$= \frac{1}{\sqrt{36}} = \frac{1}{6} < \left(\frac{3}{4}\right)$$

\_

**113.** Given :  $\dot{x} = Ax + Bu$ ;

$$\mathbf{A} = \begin{bmatrix} -1 & 2\\ 0 & 2 \end{bmatrix}, \ \mathbf{B} = \begin{bmatrix} 0\\ 1 \end{bmatrix}$$

For stability, characteristic equation

$$= \left| \begin{bmatrix} (s\mathbf{I} - \mathbf{A})^{-1} \end{bmatrix} \right| = \left| \begin{bmatrix} s+1 & 2 \\ 0 & s-2 \end{bmatrix}^{-1} \right|$$
$$= \frac{1}{\mathbf{M}} \left| \begin{pmatrix} s-2 \\ 0 & s+1 \\ \end{pmatrix} \right| = 0$$

$$\therefore \qquad \mathbf{M} = (s+1) \times (s-2) = 0$$
$$\Rightarrow \qquad (s-2) (s+1) = 0$$

 $\Rightarrow \qquad s = 2, s = -1$ 

For controllability : 
$$\mathbf{B} = \begin{bmatrix} 0\\1 \end{bmatrix}$$
  
 $\mathbf{AB} = \begin{bmatrix} -1 & 2\\0 & 2 \end{bmatrix} \begin{bmatrix} 0\\1 \end{bmatrix} = \begin{bmatrix} 2\\2 \end{bmatrix}$   
 $|[\mathbf{B}:\mathbf{AB}]| = \begin{bmatrix} \begin{bmatrix} 0 & 2\\1 & 2 \end{bmatrix}|$   
 $= 0 - 2 = -2 \neq 0$ 

So controllable.

**114.** Given : 
$$s(s + 1) (s + 3) + k(s + 2) = 0$$
;  $k > 0$   

$$\Rightarrow \qquad 1 + \frac{k(s + 2)}{s(s + 1)(s + 3)} = 0$$

But 1 + G(s) H(s) = 0

:. 
$$G(s) H(s) = \frac{k(s+2)}{s(s+1)(s+3)}$$

Roots s = 0, s = -1, s = -3 (poles); s = -2 (zero) It has real pole or zeros.

$$\phi_{A} = \frac{2q+1}{n-m} \times 180^{\circ}$$

At 
$$s = 0$$
, Asymptotes

$$\begin{split} \varphi_0 &= \frac{(2*0+1)180}{2} = 90^{\circ} \\ \text{At } s = 1, \qquad \qquad \varphi_1 &= \frac{(2*1+1)*180}{2} = 270^{\circ} \end{split}$$

$$(-\sigma_{A}) = \frac{\Sigma \text{ real parts of poles} - \Sigma \text{ real poles of zeros}}{\text{number of poles -number of zeros}}$$
$$= \frac{-1 - 3 + 2 + 0}{2} = -1$$
$$\therefore \text{ Re}[s] = -1$$

**115.** At 
$$\angle G(j\omega) = -180$$
, magnitude M = 0.5

$$\therefore \quad \text{G.M} = 20 \log\left(\frac{1}{0.5}\right) = 6\text{dB}$$
  
At  $|\text{G}(j\omega)| = 1$ , phase angle  $\angle \text{G}(j\omega) = -150$   
 $\therefore \quad \text{P.M} = 180 + (-150) = 30^{\circ}$ 

**116.** For step input,  $e_{ss} = 0.1 = \frac{1}{1+k}$   $\Rightarrow \qquad K = 9$  $G(S) = \frac{9}{S+1}$ 

...(A)

Now input is pulse r(t) = 10 [u(t) - u(t - 1)]

$$R(S) = 10 \left[ \frac{1 - e^{-s}}{s} \right]$$

$$e_{ss} = \lim_{S \to 0} \frac{S.R(s)}{1 + G(S)H(S)}$$

$$= \lim_{S \to 0} \frac{S \frac{10[1 - e^{-s}]}{S}}{\frac{S + 10}{S + 1}} = \frac{0}{10} = 0$$

**117.** |Z| < 1, so |Y| > 1

Z is having +ve real part and positive imaginary part (from characteristics)

So,Y should have +ve real part and negative imaginary part.

118. Since, open loop system stability is depends only on pole locations, hence system is stable.

There is one zero on right half of s-plane so system is non-minimum phase.

**119.** 
$$F_2(t) = L\{f(t - \tau)\} = e^{-S\tau} F_1(S)$$

$$G(S) = \frac{e^{-s\tau} F_{1}(s).F_{1}^{*}(s)}{|F_{1}(s)|^{2}} = e^{-s\tau}$$
$$G(t) = L^{-1} \{G(S)\} = \delta(t - \tau)$$

**120.** Centroid, 
$$\sigma = \frac{-2 - \left(-\frac{-3}{3}\right)}{3 - 1} = \frac{-6 + 2}{6} = \frac{-4}{6} = -\frac{2}{3}$$



$$\begin{bmatrix} 2 & 1 \\ 4 & -1 \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ l_{21} & 1 \end{bmatrix} \begin{bmatrix} u_{11} & u_{12} \\ 0 & u_{22} \end{bmatrix}$$
  
We have  $u_{11} = 2, u_{12} = 1$   
 $l_{21} u_{11} = 4$   
 $\Rightarrow \quad l_{21} = 2$   
and  $l_{21} u_{12} + u_{22} = -1$   
 $\Rightarrow \quad u_{22} = -1 - l_{21} u_{12} = -3$ 

then,  $\begin{bmatrix} 2 & 1 \\ 4 & -1 \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ 2 & 1 \end{bmatrix} \begin{bmatrix} 2 & 1 \\ 0 & -3 \end{bmatrix}$ eqn (A) does not signfy any choice

Now, from crout's decomposition,

$$\begin{bmatrix} 2 & 1 \\ 4 & -1 \end{bmatrix} = \begin{bmatrix} l_{11} & 0 \\ l_{21} & l_{22} \end{bmatrix} \begin{bmatrix} 1 & u_{12} \\ 0 & 1 \end{bmatrix}$$
  
We get,  $l_{11} = 2, l_{11} u_{12} = 1$   
 $u_{12} = \frac{1}{2} = 0.5$   
and,  $l_{21} = 4, l_{21} u_{12} + l_{22} = -1$   
 $\Rightarrow \qquad l_{22} = -3$   
Then,  $\begin{bmatrix} 2 & 1 \\ 4 & -1 \end{bmatrix} = \begin{bmatrix} 2 & 0 \\ 4 & -3 \end{bmatrix} \begin{bmatrix} 1 & 0.5 \\ 0 & 1 \end{bmatrix}$ 

Hence the choice is (d).

122. 
$$\frac{Y(s)}{R(s)} = \frac{1}{s^2 + (k+1)s + 1}$$
$$2\xi \omega_n = k + 1$$
$$\Rightarrow \xi = \left(\frac{k+1}{2}\right);$$

Peak over shoot =  $e^{-\pi\xi/(1-\xi^2)}$ 

123. 
$$|G(s)| = \frac{(9 - \omega^2)\sqrt{4 + \omega^2}}{\sqrt{\omega^2 + 1}\sqrt{\omega^2 + 9}\sqrt{16 + \omega^2}} = 0$$
  
$$\therefore \qquad \omega^2 = 9$$
  
$$\Rightarrow \qquad \omega = 3 \text{ rad/s}$$

124. 
$$L[f(t)] = F(s) = \frac{1}{s^2 + s + 1};$$
$$L[tf(t)] = (-1)\frac{dF(s)}{ds}$$
$$= (-1)\left[\frac{-(2s+1)}{s^2 + s + 1}\right] = \frac{2s+1}{s^2 + s + 1}$$

**125.** 
$$\mathbf{Q}_{c} = [\mathbf{B} \ \mathbf{A} \mathbf{B} \ \mathbf{A}^{2} \mathbf{B}]$$
  
 $\mathbf{A} = \begin{bmatrix} 0 & a_{1} & 0 \\ 0 & 0 & a_{1} \\ a_{3} & 0 & 0 \end{bmatrix}; \ \mathbf{B} = \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix};$   
 $\mathbf{A} \mathbf{B} = \begin{bmatrix} 0 \\ a_{1} \end{bmatrix}; \ \mathbf{A}^{2} \mathbf{B} = \begin{bmatrix} a_{1} a_{2} \\ 0 \end{bmatrix}$ 

$$AB = \begin{bmatrix} a_2 \\ 0 \end{bmatrix}, AB = \begin{bmatrix} 0 \\ 0 \end{bmatrix}$$
$$Q_c = \begin{bmatrix} 0 & 0 & a_1 a_2 \\ 0 & a_2 & 0 \\ 1 & 0 & 0 \end{bmatrix}$$
$$Crapk of \Omega = 3 = order of matrix, for the set of matrix of the set of the se$$

If rank of  $Q_c = 3 =$  order of matrix, then  $\begin{array}{c} a_{1}\neq 0\\ \mathbf{Q}_{c} \text{ is controllable } \begin{array}{c} a_{2}\neq 0\\ a_{3}\neq 0 \end{array}\right\}, \text{ then } |\mathbf{Q}_{c}|\neq 0$  126. Characterisitic equation is

$$\begin{aligned} 1 + G(s)H(s) &= 0\\ \therefore & 1 + \frac{k(s+1)}{s^3 + as^2 + 2s + 1} = 0\\ \Rightarrow & s^3 + s^2a + s(k+2) + (k+1) = 0\\ & s^3 &| 1 & k+2\\ & s^3 &| 1 & k+2\\ & s^1 &| \frac{k+2a}{a & k+1}\\ & s^0 &| \frac{(k+2)a - (k+1)}{a}\\ as^2 + (k+1) &= 0; s = j\omega; s^2 = -\omega^2; \omega = 2\end{aligned}$$

 $-a\omega^2 + (k+1) = 0; a\omega^2 = k+1; 4a = k+1;$ From options. k = 2, a = 0.75

$$127. y(t) = \int_{-\infty}^{t} x(\tau) \cos(3\tau) d\tau$$

Since y(t) and x(t) are related with some function of time, so they are not time-invariant. Let x(t) be bounded to some finite value k.

$$y(t) = \int_{-\infty}^{t} K\cos(3\tau) d\tau < \infty$$

y(t) is also bounded. Thus system is stable.



By voltage division rule

$$\begin{aligned} v_2(s) &= \frac{\mathrm{R} + \frac{1}{cs}}{\mathrm{R} + \frac{1}{cs} + \frac{1}{cs}} \mathrm{V}_1(s) \\ \frac{v_2(s)}{v_1(s)} &= \frac{10 \times 10^3 + \frac{1}{100 \times 10^{-6}s}}{10 \times 10^3 + \frac{2}{100 \times 10^{-6}s}} \\ &= \frac{10^4 + \frac{10^4}{s}}{10^4 + 11} \\ \frac{v_2(s)}{v_1(s)} &= \frac{(s+1)}{(s+2)} \end{aligned}$$

129. 
$$u(s)$$
   
 $\frac{y(s)}{u(s)} = \frac{1}{s}$   
 $y(s) = \frac{1}{s} \times u(s)$   
For unit step 1/s  
 $u(s) = \frac{1}{s}$ 

y(s) =

y(t) = t u(t)

As 
$$h(t) =$$

input response  

$$\delta(t) \rightarrow t \ a \ (t)$$
  
 $u(t) \rightarrow tu(t)dt = u = \int_{0}^{t} t dt = \frac{t^{2}}{2}u(t)$ 

t u(t)

$$u(t-1) \rightarrow \frac{(t-1)^2}{2}u(t-1)$$

**131.Consider option** (*a*): In which all the poles lie on the left of  $j\omega$  axis which satisfy casual stable LTI system.

**Option** (*b*): For a stable casual system, there are no restriction for the position of zeroes on s plane. **Option** (*c*): text true.

**Option** (*d*): Roots of characteristic equation are all closed loop poles and they all line on the left side of the  $j\omega$  axis.

**132.** Two systems with impulse responses  $(h_1)(t)$  and  $h_2(t)$  are connected in cascade



Then the overall impulse response of the Cascaded system in given by convolution  $h_{\rm 1}(t)$  and  $h_{\rm 2}(t)$ 

Overall Impulse Respone = Convolution of  $h_1(t), h_2(t)$ 

= Product of  $H_1(s), H_2(s)$ 

 $\label{eq:Where H_1(s), H_2(s) are transfer function is $S$ domain. $$133. For max power$}$ 

$$R_{\rm L} = \sqrt{R_s^2 + X_s^2}$$
$$= \sqrt{4^2 + 3^2} = 5$$



$$\Rightarrow 20 \log k = 32$$
$$k = 10^{1.6} = 39.8$$

 $\Rightarrow$  As slope is – 40db/decade so two pols at ons'm

V(S)

140.

so  $T(s) = \frac{39.8}{52}$ 135. V(S)  $S^{-1}$   $S^{-1}$ Forward path  $P_1 = S^{-1}, S^{-1} = S^{-2}$   $\Delta_1 = 1$  $P_2 = S^{-1}$   $\Delta_2 = 1$ 

Loop

$$\begin{split} \mathbf{L}_{1} &= -4\mathbf{S}^{-1} \\ \mathbf{L}_{2} &= -2\mathbf{S}^{-1}\mathbf{S}^{-1} = -2\mathbf{S}^{-2} \\ \mathbf{L}_{3} &= -2\mathbf{S}^{-1} \\ \mathbf{L}_{4} &= -4 \\ \mathbf{T}(\mathbf{S}) &= \frac{\mathbf{S}^{-2} + \mathbf{S}^{-1}}{1 + 4 + 2\mathbf{S}^{-1} + 2\mathbf{S}^{-2} + 4\mathbf{S}^{-1}} \\ &= \frac{\mathbf{S}^{-1} + \mathbf{S}^{-2}}{\mathbf{S} + 6\mathbf{S}^{-1} + 2\mathbf{S}^{-2}} \\ &= \frac{S + 1}{5\mathbf{S}^{2} + 6\mathbf{S} + 2} \end{split}$$

136. Apply Laplace transform



for input step voltage  $\rightarrow$ 

$$y(s) = h(s)\frac{1}{s}$$
  
=  $\left[e^{-s} + e^{-3s}\right]1/s$   
 $y(t) = u(t-1) + u(t-3)$   
 $u(t) = 1$  for t 30 = 0 prt < 0  
 $\therefore$  O/P in  $y(2) u(t-1) + u(t-3) = (4-1)$   
=  $4(z-1) + 4(2-3)$   
=  $4(1) + (4-1)$   
=  $0 + 1 = 1$ 

**137.** When all elements of row have zero values which leads to auxiliary equation. This premature termination of the array indicates the presence of imaginary roots.

138. 
$$\frac{C(s)}{R(s)} = \frac{k}{(s+1)(s+2)-k}$$
 will give the root locus

given the diagram.

139. State transition matrix,

$$\begin{split} \phi(t) &= \mathcal{L}^{-1} [sI - A]^{-1} \\ &= \mathcal{L}^{-1} \begin{bmatrix} s - 1 & 0 \\ -1 & s - 1 \end{bmatrix}^{-1} \\ &= \mathcal{L}^{-1} \left\{ \frac{1}{(s - 1)^2} \begin{bmatrix} s - 1 & 0 \\ 1 & s - 1 \end{bmatrix} \right\} \\ \phi(t) &= \mathcal{L}^{-1} \begin{bmatrix} \frac{1}{s - 1} & 0 \\ \frac{1}{(s - 1)^2} & \frac{1}{s - 1} \end{bmatrix} = \begin{bmatrix} e^t & 0 \\ te^t & e^t \end{bmatrix} \\ G(s) &= \frac{5(s + 4)}{s(s + 0.25)(s^2 + 4s + 25)} \end{split}$$

$$G(j\omega) = \frac{5(j\omega+4)}{j\omega(j\omega+0.25)[(j\omega)^2+4j\omega+25]}$$
$$= \frac{5\times4\times4\left(\frac{j\omega}{4}+1\right)}{j\omega(4j\omega+1)\left[\left(\frac{j\omega}{5}\right)^2+\frac{4}{25}j\omega+1\right]\times25}$$
$$= \frac{80}{25}\frac{\left(\frac{j\omega}{4}+1\right)}{j\omega(4j\omega+1)\left[\left(\frac{j\omega}{5}\right)^2+\frac{4(j\omega)}{25}+1\right]}$$

Constant gain term =  $\frac{80}{25}$  = 3.2

Corner frequencies are  $\omega = 4$ ,  $\omega = 0.25$ ,  $\omega = 5$ Then highest corner frequency  $\omega = 5$  rad/sec. **141.** For controllability,

$$\mathbf{M}_{\mathrm{C}} = \begin{bmatrix} \mathbf{Q} & \mathbf{P}\mathbf{Q} \end{bmatrix}$$
$$\mathbf{M}_{\mathrm{C}} = \begin{bmatrix} 0 & 1 \\ 1 & -3 \end{bmatrix}$$
$$\Delta \mathbf{M}_{\mathrm{C}} = \begin{vmatrix} 0 & 1 \\ 1 & -3 \end{vmatrix} \neq 0$$

Therefore, the system is Controllable. Now, for observability.

$$\mathbf{M}_{o} = \begin{bmatrix} \mathbf{R}^{\mathrm{T}} & \mathbf{R}^{\mathrm{T}} \mathbf{p}^{\mathrm{T}} \end{bmatrix} = \begin{bmatrix} \mathbf{0} & \mathbf{0} \\ \mathbf{1} & -\mathbf{3} \end{bmatrix}$$
$$\Delta \mathbf{M}_{o} = \begin{vmatrix} \mathbf{0} & \mathbf{0} \\ \mathbf{1} & -\mathbf{3} \end{vmatrix} = \mathbf{0}$$

Therefore, the system is not observable.

**142.** From the signal flow graph,  $G(s) = \frac{C(s)}{U(s)}$ 

By mason's gain relation,

Transfer function = 
$$\frac{P_1\Delta_1 + P_2\Delta_2 + \dots}{\Delta}$$

$$P_{1} = \frac{h_{1}}{S} ; P_{2} = \frac{h_{0}}{s_{2}}$$

$$\Delta_{1} = \left[1 + \frac{a_{1}}{s}\right]; \Delta_{2} = 1;$$

$$\Delta = 1 + \frac{a_{1}}{s} + \frac{a_{0}}{s^{2}}$$
Transfer function 
$$= \frac{\frac{h_{1}}{s} \left[1 + \frac{a_{1}}{s}\right] + \frac{b_{0}}{s^{2}}}{1 + \frac{a_{1}}{s} + \frac{a_{0}}{s^{2}}} = \frac{b_{1}s + b_{0}}{s^{2} + a_{1}s + a_{0}}$$
144.
$$\underset{\mathbf{R}(s) + \mathbf{A} = \frac{1}{s} + \underbrace{\mathbf{A} = \frac{1}{s} + \frac{a_{0}}{s^{2}}}_{\mathbf{R}(s) = \mathbf{S}}$$

$$\underset{\mathbf{R}(s) = \mathbf{S}}{\frac{C(s)}{\mathbf{R}(s)}} = \frac{\mathbf{S}}{s^{2} + s + 2}$$
145. Transfer function
$$\underset{\mathbf{R}(s) = \mathbf{A} = \frac{1}{s} + \underbrace{\mathbf{R} = \frac{1}{s} - \frac{1}{s}$$

 $\Rightarrow C[SI - A]^{-1} \cdot B \cdot = \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} S & -1 \\ 1 & (s+1) \end{bmatrix}^{T} \begin{bmatrix} 0 \\ 1 \end{bmatrix}$ Transfer function  $= \frac{1}{s^{2} + s + 1}$  $\frac{G(s)}{1 + G(s)} = \frac{1}{s^{2} + s + 1}$  $\Rightarrow \qquad G(s) = \frac{1}{s^{2} + s}$ 

Steady state error for unit step

$$e_{ss} = \frac{A}{1+K_{p}} = \frac{1}{1+\lim_{s\to 0} G(s)}$$

$$e_{ss} = \frac{1}{1+\lim_{s\to 0} \frac{1}{s^{2}+s}} = \frac{1}{1+\infty}$$

$$e_{ss} = 0$$

$$G(s) = k \cdot \frac{(1+3s)}{(1+s)}$$

$$G(s) = \frac{3k \cdot \left(s + \frac{1}{3}\right)}{(s+1)}$$

146.

$$k = 1$$

$$\frac{1}{T} = \frac{1}{3} \Rightarrow \frac{1}{\alpha T} = 1$$

$$\omega_m = \frac{1}{\sqrt{3}}; \alpha = \frac{1}{3}$$

$$G(s)|_{\omega = \frac{1}{\sqrt{3}}} = \frac{\sqrt{4}}{\sqrt{\frac{4}{3}}} \Rightarrow \sqrt{3}$$

$$G_m|_{indB} = 20 \log \sqrt{3} = 4.77 \, dB$$

$$\begin{split} \phi_{m} &= \sin^{-1} \left[ \frac{1-\alpha}{1+\alpha} \right] \\ \alpha &= \frac{1}{3} = \sin^{-1} \left[ \frac{1-\frac{1}{3}}{1+\frac{1}{3}} \right] = \sin^{-1} \left( \frac{1}{2} \right) \\ \phi_{m} &= 30^{\circ} \\ \mathbf{147.} \\ \chi_{1}(s) &= 0 \\ \psi_{m} &= 10^{\circ} \\ \chi_{1}(s) &= 0 \\ \psi_{m} &= 10^{\circ} \\ \chi_{1}(s) &= 0 \\ \psi_{m} &= 10^{\circ} \\ \chi_{2}(s) &= 0 \\ \psi_{m} &= 10^{\circ} \\ \chi_{2}(s) &= 0 \\ \psi_{m} &= 10^{\circ} \\ \chi_{2}(s) &= 0 \\ \chi_{$$



151. System is second order

It means, number of poles = 2

System has perfectly flat magnitude response.

It means, It is all pass system

In all pass system,

Poles and zeros are mirror image about Imaginary axis,

One pole = (2 - j3)Other pole = (2 + j3)



Therefore

poles at 
$$(2 \pm j3)$$

zeros at (–2  $\pm j3$ )

**152.** Open loop transfer function of a third order unity feedback system

$$G(s) = \frac{K}{s(s+1)(s+2)}$$

Now zero is introduce at -3. Modified transfer function,

$$G(s) = \frac{K(s+3)}{s(s+1)(s+2)}$$

Closed loop characteristic equation,

$$1 + G(s) = 0$$

$$\begin{split} 1 + \frac{\mathrm{K}(s+3)}{s(s+1)(s+2)} &= 0\\ s(s+1)(s+2) + \mathrm{K}(s+3) &= 0\\ s^3 + 3s^2 + 2s + \mathrm{K}s + 3\mathrm{K} &= 0\\ s^3 + 3s^2 + (\mathrm{K}+2)s + 3\mathrm{K} &= 0 \end{split}$$

Routh array corresponding to characteristic equation,

$$\begin{array}{c|cccc} s^{3} & 1 & (K+2) \\ s^{2} & 3 & 3K \\ s^{1} & 2 & 0 \\ s^{0} & 3K & 0 \end{array}$$

For every value of K > 0

 $I^{st}$  column of the array are positive.

Therefore, Root locus of modified system never transits to unstable region.



$$G_1(s) = \frac{1}{s}$$
 and  $G_2(s) = s$ 

From the Nquist plot,

*:*.

 $\mathbf{G}_{1}\mathbf{G}_{2}(s) = \frac{1}{s} \times s = 1$ 

**154.** The characteristic equation of given function is 1 + G(s) H(s) = 0

: 
$$s(s^2 + 3s + 2) + K = 0$$
 (::  $H(s) = 1$ )  
:  $-k = s^3 + 3s^2 + 2s$ 

In order to find break away point

We have 
$$\frac{dK}{ds} = 0$$
  
 $\therefore 3s^2 + 6s + 2 = 0$ 

 $\therefore$  S = -0.42 is the solution that makes k > 0

**155.** Here 
$$A = \left| \frac{j\omega}{j\omega + 2} \right|_{\omega=2} = \frac{2}{\sqrt{2^2 + 2^2}} = \frac{2}{2\sqrt{2}} = \frac{1}{\sqrt{2}}$$
  
 $\phi = \left| \frac{j\omega}{j\omega + 2} \right|_{\omega=2}$   
 $= 90^\circ - \tan^{-1}\frac{2}{2}$   
 $= 90^\circ - \tan^{-1}(1)$   
 $= (90^\circ - 45^\circ) = 45^\circ$   
So, the value of A and  $\phi$  are  $\frac{1}{\sqrt{2}}$ , +45° respectively

#### 6.44 Control Systems

**156.** Given transfer function  $G(s) = \frac{100}{(s+1)^3}$ 

$$\therefore \quad \angle \frac{100}{(j\omega+1)^3} = -180^{\circ} \bigg|_{\omega=\omega_{pc}}$$
$$-3 \tan^{-1} \omega_{pc} = -180^{\circ}$$
$$\therefore \quad \omega_{pc} = \sqrt{3}$$

So, the phase cross-over frequency of the given transfer function is  $\sqrt{3}$  .

**157.** From the given Bode plot the corner frequencies are 2 rad/sec and 4 rad/sec respectively.

Transfer function (TF) = 
$$\frac{\text{Ks}}{\left(1 + \frac{s}{2}\right)\left(1 + \frac{s}{4}\right)^2}$$
  
20logK + 20 log $\omega$  = 0 dB  
At  $\omega$  = 0.5, we get

$$20 \log K + 20 \log 0.5 = 0$$
$$20 \log K = -20 \log \left(\frac{1}{2}\right)$$
$$20 \log K = 20 \log (2)$$
$$\therefore K = 2$$

$$\therefore \text{ TF} = \frac{2s}{(1+0.55)(1+0.25s)^2}$$

**158.** Given transfer function G(s). $H(s) = \frac{s+3}{s^2(s-3)}$ 

$$CE = 1 + G(s) \cdot H(s) = 1 + \frac{s+3}{s^2 - 3s^2} = 0$$
  

$$s^3 - 3s^2 + s + 3 = 0$$
  

$$S^3 | 1 1$$
  

$$S^2 | -3 3$$
  

$$S^1 | 2$$
  

$$S^0 | 3$$

System is unstable with two right half of s-plane poles

 $\therefore Z = 2, P = 1$ N = P - Z

$$N = 1 - 2$$

N = 1 - 2 = -1 (it shows once in clockwise direction.

 $I = 2 \int_{-\infty}^{\infty} \left( \frac{\sin 2\pi t}{\pi t} \right) dt$ 

...(i)

#### 159. Let

As per the fourier transform of

$$\frac{2\sin(t\tau/2)}{t} \longrightarrow 2\pi \operatorname{rect}\left(\frac{\omega}{t}\right)$$
$$\frac{\sin(2\pi t)}{\pi t} \longrightarrow \operatorname{rect}\left(\frac{\omega}{4\pi}\right)$$

So, 
$$\int_{-\infty}^{\infty} \frac{\sin(2\pi t)}{\pi t} e^{-j\omega t} dt = \operatorname{rect}\left(\frac{\omega}{4\pi}\right)$$
 ...(*ii*)

Putting  $\omega = 0$  in above equation, we get

$$\int_{-\infty}^{\infty} \frac{\sin(2\pi t)}{\pi t} dt = 1$$

Now from equation (i)

$$I = 2 \int_{-\infty}^{\infty} \frac{\sin(2\pi)}{\pi t} dt = 2 \times 1 = 2$$
$$\mathbf{p} = \begin{bmatrix} 3 & 1 \end{bmatrix}$$

1 - |1 | 3|

**160.** Given

*.*..

According to question

$$P\left[\frac{x}{y}\right] = \left[\frac{a}{b}\right]$$
$$\begin{bmatrix}3 & 1\\1 & 3\end{bmatrix}\begin{bmatrix}x\\y\end{bmatrix} = \begin{bmatrix}a\\b\end{bmatrix}$$
$$3x + y = a$$
$$x + 3y = b$$
$$a^2 + b^2 = 1$$
$$\Rightarrow 10x^2 + 10y^2 + 12xy = 1$$
Ellipse with major axis along 
$$\begin{bmatrix}1\\1\end{bmatrix}.$$

161. Through open loop transfer function

G(s) = 
$$\frac{K(s+1)}{s(1+Ts)(1+2s)}$$
, K > 0, and T > 0

For closed loop system stability, characteristic equation is

$$1 + G(s) H(s) = 0$$
  
1 +  $\frac{K(s+1)}{s(1+Ts)(1+2s)} \cdot 1 = 0$ 

s(1 + Ts) (1 + 2s) + k(s + 1) = 0 $2Ts^{3} + (2 + T)s^{2} + (1 + k)s + k = 0$ Using Routh's criteria

$$\begin{array}{c|c} s^{3} \\ s^{2} \\ s^{1} \\ s^{0} \\ s^{0} \\ \end{array} \begin{array}{c} 2T & (1+k) \\ (2+T) & k \\ \hline (2+T) & k \\ \hline (2+T) & 0 \\ k \\ \end{array}$$

For stability, k > 0and (2 + T) (1 + k) - 2Tk > 0k(2 + T - 2T) + (2 + T) > 0or -(T - 2)k + 2(2 + T) > 0 $-k > -\frac{(2 + T)}{(T - 2)}$ 

Product of distances from all

$$\Rightarrow K = \frac{\text{the poles to break away point}}{\text{Product of distance from all}}$$
the zeros to break away point

Gain, 
$$K = \frac{1 \times 2}{2} = 1$$

Hence the value of K is 1.

164. The characteristic equation is given by

$$s^3 + Ks^2 + (K+2)s + 3 = 0$$

For this system to be stable, using Routh's criterion, we get

$$3 < K(K+2)$$

 $K^2 + 2K - 3 > 0$  $K^2 + 3K - K - 3 > 0$ K(K + 3) - 1 (K + 3) > 0(K+3)(K-1) > 0

So from the given option, K > 1 conditions should be satisfied

165. The transfer function of a system is given by

$$\frac{V_0(s)}{V_1(s)} = \frac{1-s}{1+s} = H(s)$$

For the minimum and maximum values of "\op"

$$H(j\omega) = \frac{1-j\omega}{1+j\omega}$$

$$\angle H(j\omega) = -2 \tan^{-1} \omega$$

At 
$$\omega = 0$$
;  $\angle H(j\omega) = 0^{\circ}$ 

*.*...

and

At 
$$\omega = \infty$$
;  $\angle \mathbf{H}(j\omega) = -\pi$ 

Therefore the minimum and maximum value of  $\phi$  are  $-\pi$  and 0.

 $\dot{X} = \begin{bmatrix} 1 & 2 \\ 2 & 0 \end{bmatrix} X + \begin{bmatrix} 1 \\ 2 \end{bmatrix} u$ 166. Given,  $Y = \begin{bmatrix} 1 & 0 \end{bmatrix} X$ 

Transfer function = 
$$C[sI - A]^{-1}B + D$$

$$[sI - A]^{-1} = \begin{bmatrix} \frac{s}{s^2 - s - 4} & \frac{2}{s^2 - s - 4} \\ \frac{2}{s^2 - s - 4} & \frac{s - 1}{s^2 - s - 4} \end{bmatrix}$$
$$[sI - A]^{-1} \cdot B = \begin{bmatrix} \frac{s + 4}{s^2 - s - 4} \\ \frac{s - 1}{s^2 - s - 4} \end{bmatrix}$$
$$\frac{Y(s)}{U(s)} = C[sI - A]^{-1} \cdot B = \frac{s + 4}{s^2 - s - 4}$$

So the transfer function of the system  $\frac{Y(s)}{U(s)}$  is

$$\frac{s+4}{s^2-s-4}$$

*:*..

or

$$<\frac{T+2}{(T-2)}$$

Hence the closed loop system will be stable, if

k

$$0 < k < \frac{\mathrm{T}+2}{\mathrm{T}-2}$$

162. Damping ratio

 $\xi = 0.5$ Undamped natural frequency  $\omega_{n} = 10 \text{ rad/sec}$ Steady state output to a unit step input  $C_{ss} = 1.02$ Hence steady state error = 1.02 - 1.00 $e_{00} = 0.02$ : Required characteristics equation is,  $s^2 + 2\xi\omega_n s + \omega_n^2 = 0$  $s^2 + 2 \times 0.5 \times 10s + 100 = 0$  $s^2 + 10s + 100 = 0$ From options, if we take option (b) $C_{ss} = \lim_{s \to 0} s.C(s)$ then  $= \lim_{s \to 0} s \times \frac{1}{s} \times \frac{102}{s^2 + 10s + 100}$  $C_{ss} = 1.02$ Hence the transfer function of a system 1.02is  $\frac{1}{s^2 + 10s + 100}$ 163. Given, the gain at breakaway point Open loop transfer function  $\mathbf{G}(s) = \frac{\mathbf{K}s}{(s-1)(s-4)}$  $OLTF \Rightarrow$ Now, characteristics equation is 1 + G(s) H(s) = 0 $\frac{\mathrm{K}s}{(s-1)(s-4)} + 1 = 0$  $Ks + (s^2 + 5s + 4) = 0$  $\Rightarrow$ 

For break away point :

$$-\frac{d\mathbf{K}}{ds} = 0$$
$$-\frac{d\mathbf{K}}{ds} = -\left[1 - 0 - \frac{4}{s^2}\right] = 0$$
$$s = \pm 2$$

K =  $-\frac{(s^2 - 5s + 4)}{s} = \left[s - 5 + \frac{4}{s}\right]$ 

we get

Therefore valid break away point is

$$s = 2,$$

Now gain at s = 2 is

#### 6.46 Control Systems

167. From the given fig.

Transfer function, 
$$\left(\frac{Y(s)}{U_1(s)}\right) = \frac{\frac{K_1}{LJs^2}[1]}{1 - \left[-\frac{R}{LS} - \frac{K_1K_2}{LJs^2}\right]}$$
  

$$= \frac{K_1}{LJs^2 + RJs + K_1K_2}$$
168.  $V_1 \angle \delta$  jx  $V_2 \angle \delta$   
Assuming loss-less line  
 $P = \frac{V_1 V_2}{X} \sin \delta$   
Max. power =  $P_{max} = \frac{V_1 V_2}{X}$   
For Lossy-Line,  
 $V_1 \angle \delta$   $K$  jx  $V_1$   
 $\Theta$  Because of Resistance there is a power loss  
Power Loss = 3I^2R  
 $\therefore$  Lossy Line  $P_{max} < \frac{V_1 V_2}{X}$   
169.  $\frac{15}{s^2 + 5s + 15} = \frac{\omega_{n_1}^2}{s^2 + 2\xi_1 \omega_{n_1} s + \omega_{n_1}^2}$   
 $\omega_{n_1} = \sqrt{15} = 3.872$   
 $2G_1 \omega_{n_1} = 5$   
 $\Rightarrow$   $G_1 = \frac{5}{2 \times 3.872} = 0.64$  under damped  
 $\frac{25}{s^2 + 10s + 25} = \frac{\omega_{n_1}^2}{s^2 + 2\xi_2 \omega_{n_2} s + \omega_{n_2}^2}$   
 $\omega_{n_2}^2 = 25 \Rightarrow \omega_{n_2} = \pm 5 \text{ rad / sec}$   
 $2\xi_2 \omega_{n_2} = 10$   
 $\Rightarrow$   $\xi_2 = \frac{10}{2 \times 5} = 1$  critically damped  
170.  $s^7 + s^6 + 7s^5 + 14s^4 + 31s^3 + 73s^2 + 25s + 200 = 0$   
 $s^7$  1 7 31 25  
 $s^6$  1 14 73 200  
 $s^5$   $-7$   $-42$   $-175$  200

Auxiliary equation  $8s^4 + 48s^2 + 200$ A.E. =  $s^4 + 6s^2 + 25 = 0$  $\frac{d(A \cdot E)}{2} = 4s^3 + 12s$ ds A.E:  $s^4 + 6s^2 + 25 = 0$  $s = -3 \pm j4 \rightarrow 2$  poles in LHS = + 3  $\pm$  j4  $\rightarrow$  2 poles in RHS. 2 sign changes in above  $A.E \Rightarrow 2$  poles in RHS Total poles = 7Poles in RHS = 2 + 2 = 4Poles in LHS = 7 - 4 = 3[No poles on Imaginary axis]

171. The given characteristic equation of LTI system is

$$\Delta(s) = s^4 + 3s^3 + 3s^2 + s + k$$
, for

According to Routh hurwitz criteria,

$$\begin{array}{c|ccccc} s^{4} & 1 & 3 & k \\ s^{3} & 3 & 1 & 0 \\ s^{2} & 8 & k \\ s^{1} & \frac{8/3 - 3k}{8/3} & k \\ s^{0} & k \end{array}$$

For stability all elements of 1st column should be positive.

$$\frac{\frac{8}{3} - 3k}{8/3} > 0 \text{ and } k > 0$$
  
$$\therefore \quad k < \frac{8}{9} \text{ and } k > 0$$
  
$$\therefore \quad 0 < k < \frac{8}{9}$$

172. It is given that

$$H(s) = \frac{a_1 s^2 + b_1 s + c_1}{a_2 s^2 + b_2 s + c_2}$$
  
If  $a_1 = b_1 = 0$ , then H(s) becomes  
$$H(s) = \frac{0 + 0 + 4}{a_2 s^2 + b_2 s + c_2}$$
  
$$H(s) = \frac{c_1}{a_2 s^2 + b_2 s + c_2}$$

Now H(0) =  $\frac{c_1}{c_2}$  (i.e., as low frequency s  $\rightarrow 0$  $\Rightarrow \omega \rightarrow 0)$ 

 $H(\infty) = 0$ (i.e., as high frequency  $s \rightarrow \infty \Rightarrow \omega \rightarrow \infty$ ) So the system passes low frequency and blocks high frequency. So it works as a low pass filter.

- 1 42  $s^4$ 8 48 200  $s^3$ 4 12 $\rightarrow Row$  of zeros  $s^2$ 24200 $s^1$ -170 $s^0$ 200

16

**3.** 
$$G(s) = \frac{\pi e^{-0.25s}}{s}$$

Now when the Nyquist Plot psses through negative real axis then in phase become  $(-180^{\circ})$ 

 $\therefore \qquad -180^\circ = -90^\circ - 0.25\omega \frac{180^\circ}{\pi}$  $\therefore \qquad \omega = 2\pi$ 

Now Magnitude at this frequency

$$G(2\pi) = \left| \frac{\pi e^{-0.25j(2\pi)}}{j2\pi} \right| = \frac{1}{2} = 0.5$$

At negative real axis the co-ordinate becomes (-0.5, j0)

- **174.**  $\rightarrow$  From the given bode-plot, we can say
  - $\rightarrow$  At origin, there is a pole at origin, since the initial slope is 20 db/dec.
  - → At  $\omega = 20$ , the change in slope is -60 (-40)= -20 db/dec, so it imply one pole at  $\omega = 20$ .
  - → At  $\omega$  = 1, the change in slope is 40 (– 20) = –20 dB/sec, so it imply one pole at  $\omega$  = 1.
  - → So in total the transfer function has 3 poles, hence at  $\omega = \infty$ , the net phase contributed by 3 poles is  $-270^{\circ}$
  - $\rightarrow$  Hence statement I is false and II is true.

175.



The transfer function of a lead compensator is

$$D(s) = 3 \left[ \frac{s + \frac{1}{3T}}{s + \frac{1}{T}} \right]$$

The frequency at which phase is maximum is given by geometric mean of pole and zero location,

$$\omega_{\rm m} = \sqrt{\left(\frac{1}{3T}\right) \left(\frac{1}{T}\right)} = \sqrt{\frac{1}{3T^2}}$$

176. From the given state space model, we can say that

$$\mathbf{A} = \begin{bmatrix} \mathbf{0} & \mathbf{1} \\ -\alpha & -2\beta \end{bmatrix}, \ \mathbf{B} = \begin{bmatrix} \mathbf{0} \\ \alpha \end{bmatrix}, \ \mathbf{C} = \begin{bmatrix} \mathbf{1} & \mathbf{0} \end{bmatrix}$$

Transfer function

Now  $T(s) = C(sI - A)^{-1}B$ 

$$= \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} s & 0 \\ 0 & s \end{bmatrix} - \begin{pmatrix} 0 & 1 \\ -\alpha & -2\beta \end{bmatrix}^{-1} \begin{bmatrix} 0 \\ \alpha \end{bmatrix}$$

$$= \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} s & -1 \\ \alpha & s - 2\beta \end{bmatrix}^{-1} \begin{bmatrix} 0 \\ \alpha \end{bmatrix}$$

$$= \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} \frac{1}{s(s+2\beta) + \alpha} \begin{bmatrix} s+2\beta & 1 \\ -\alpha & s \end{bmatrix} \end{bmatrix} \begin{bmatrix} 0 \\ \alpha \end{bmatrix}$$

$$= \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} \frac{1}{s^2 + 2\beta s + \alpha} \end{bmatrix} \begin{bmatrix} \alpha \\ \alpha s \end{bmatrix} = \frac{\begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} \alpha \\ \alpha s \end{bmatrix}$$

$$= \frac{\alpha}{s^2 + 2\beta s + \alpha}$$
 by comparing with standard equation  $+ \omega_n^2$  we get

$$\omega_n = \sqrt{\alpha} \text{ and } \xi \omega_n = \beta$$

 $\therefore \quad \xi = \frac{P}{\sqrt{\alpha}}$  **177.** Apply the Mason's gain formula to the

given, SFG T.F =  $\frac{ad}{1 - ade - dc}$  ...(1)

apply the Mason's gain formula to the optionn (c)

$$T.F = \frac{a \frac{d}{1 - cd}}{1 - \frac{ade}{1 - cd}} = \frac{ad}{1 - ade - dc} \qquad ...(2)$$
  
(1) = (2)



**178.** Given that the final value to a step of 5 V is 10 and T = 0.5s

$$\begin{split} \frac{C(s)}{R(s)} &= \frac{K}{1+TS} \\ \therefore \quad T.F = \frac{2}{1+0.5s} \\ Final value \ & \underset{s \rightarrow 0}{LtS}(TF) \frac{5}{S} = S \bigg( \frac{2}{1+0.5s} \bigg) \cdot \frac{5}{S} = 10 \end{split}$$

$$T.F = \frac{2}{1+0.5s}$$

**179.** Characteristic Equation = 
$$1 + G(s) H(s) = 0$$

$$\therefore \quad 1 + \frac{1}{(s+1)} \frac{1}{s^2} \frac{20}{(s+20)} = 0$$
$$(s^3 + s^2) (s+20) + 20 = 0$$
$$s^4 + 21s^3 + 20s^2 + 20 = 0$$



#### 6.48 Control Systems

The given system is of 4<sup>th</sup> order

+s1 2020  $+s^3$ 21 0 0  $+s^2$ 20 20 0 -21 $-s^1$  $+s^0$ 20

Number of sign changes in the first column = 2  $\therefore$  two roots are present in the right half of s-plane, and system is unstable.

**180.** Given transfer function =  $\frac{s^2 + s + 1}{s^3 + 2s^2 + 2s + k} = 0$ 

Now, characteristic equation is given by

$$\label{eq:G} \begin{array}{l} 1+G(s)+H(s)=0\\ \\ \therefore \ \ s^{3}+3s^{2}+3s+(k+1)=0 \end{array}$$

Now according to Routh Hurwitz Criteria,

$$egin{array}{ccc} & {
m s}^3 & 1 & & 3 \ {
m s}^2 & {
m s}^3 & {
m s}^3 & & 3 \ {
m s}^1 & {
m s}^{-(k+1)} & {
m k}+1 \ {
m s}^0 & {
m k}+1 \end{array}$$

For marginal stability,

$$\therefore 9 - (k+1) = 0$$
  
$$\therefore k = 8$$

**181.** As given circuit:



 $\frac{C(s)}{R(s)}$ G (standard negative feedback So, 1+GH system)

$$\begin{split} \mathbf{E}(\mathbf{s}) &= \mathbf{R}(\mathbf{s}) - \mathbf{C}(\mathbf{s})\mathbf{H} \\ &= \mathbf{R}(\mathbf{s}) - \frac{\mathbf{GH}}{\mathbf{1} + \mathbf{GH}}\mathbf{R}(\mathbf{s}) = \mathbf{R}(\mathbf{s}) \bigg[ \mathbf{1} - \frac{\mathbf{GH}}{\mathbf{1} + \mathbf{GH}} \bigg] \\ &= \mathbf{R}(\mathbf{s}) \bigg[ \frac{\mathbf{1} + \mathbf{GH} - \mathbf{GH}}{\mathbf{1} + \mathbf{GH}} \bigg] = \mathbf{R}(\mathbf{s}) \frac{\mathbf{1}}{\mathbf{1} + \mathbf{GH}} \\ &\Rightarrow \frac{\mathbf{E}(\mathbf{s})}{\mathbf{R}(\mathbf{s})} = \frac{\mathbf{1}}{\mathbf{1} + \mathbf{GH}} \end{split}$$

Hence, option (c) is the correct answer. 182. As given function;

$$H(s) = \frac{As + B}{s^{2} + Cs + D}$$
  
At low = freq. (s = 0)

$$H(0) = \frac{B}{D} \text{ (exist)}$$
  
At high - freq. (s =  $\infty$ )  
 $H(\infty) = 0 \text{ (No output)}$ 

Since,  $H(\infty)$  should be non-zero for HPF. So, the above system cannot be operated as HPF because for high pass filter should pass high-frequency component i.e  $H(\infty)$  should be non-zero.

Hence, the system cannot operate as high pass filter.

183. For, the first order system, there exist one finite pole, by observing magnitude of plot, it is maintaining constant magnitude, this exist for all pass system.

So, Possibilities, 
$$T.F = \frac{s-1}{s+1} \text{ or } \frac{1-s}{1+s}$$
  
Observe phase angle at  $w \to \infty$   
 $\phi = -180^{\circ}$   
Number of poles,  $P = 1$   
Number of zeros,  $Z = 1$   
It is having one pole at left and one zero at right.  
**184.** As we know that  
Characteristics equation,  $1 + G(s) = 0$ 

So, 
$$1 + \frac{K}{s^2 + 2s - 5} = 0$$
  
Now,  $s^2 + 2s + (K - 5) = 0$ 

By Routh table analysis,

$$K - 5 > 0, K > 5 and K > 0$$

For stability K > 5

185. In given figure;

Given Nyquist contour S-plane



Corresponding to the Nyquist contour, the Nyquist plot is drawn, and is shown in figure below,



From the plot Number of Encirclement (N) = 1

186. Transfer function of given system is

$$T.F = \frac{C(s)}{R(s)} = \frac{100}{s^2 + 10s + 100}$$
 By compair it to standard transfer function

$$T.F = \frac{\omega_n^2}{s^2 + 2\xi\omega_n s + \omega_n^2}$$
$$2\xi\omega_n = 10$$
So, 
$$\omega_n^2 = 100$$
$$\Rightarrow \qquad \omega_n = 10 \text{ rad/sec}$$
$$\xi = \frac{10}{2 \times 10}$$
$$\xi = 0.5$$

187. As given figure :



From signal flow graph;



Now by using Masson's graph formula.

$$\frac{Y(s)}{R(s)} = \left[\frac{P_1\Delta_1 + P_1\Delta_2}{(1 - L_1)}\right] = \frac{3 + \frac{2}{s}}{1 - \left(\frac{1}{s}\right)} = \frac{3s + 2}{(s - 1)}$$

188. As given transfer function;



Now for mapping  $C_2$ ,

$$G(s) \ H(s) = \lim_{R \to \infty} \left[ \frac{3 \, R e^{j\theta} + 5}{R e^{j\theta} - 1} \right] \qquad (\, . : \ S = R e^{i\theta})$$

So, G(s) H(s) = 3189. As given that,

$$G(s) = \frac{1}{s-1}$$
$$H(s) = 1$$

$$\frac{R(s)}{3 + \frac{1}{s}} \xrightarrow{Y_1(s)} Y(s)$$
Open loop transfer function =  $\left[\frac{(3s+1)}{s(s-1)}\right]$ 
Closed loop transfer function =  $\left[\frac{Y(s)}{R(s)}\right]$ 

$$= \left[\frac{3s+1}{s^2+2s+1}\right]$$

Final value of plant :

$$\begin{split} Y_{ss} &= \lim_{s \to 0} s Y(s) \\ Y_{ss} &= \lim_{s \to 0} s \times \left[ \frac{3s+1}{s^2+2s+1} \right] \times \frac{1}{s} = +1 \qquad \left[ \therefore R(s) = \frac{1}{s} \right] \\ As now, & Y(s) = Y_1(s) \times \left[ \frac{1}{s-1} \right] \\ & Y_1(s) = (s-1) Y(s) \\ & Y_1(s) = (s-1) \times \left[ \frac{3s+1}{s(s^2+2s+1)} \right] \\ & (Y_1)_{ss} = \lim_{s \to 0} s Y_1(s) = -1 \end{split}$$

**190.** As given transfer fucntion;

$$\begin{split} \mathrm{K}(\mathrm{s}) &= \frac{1 + \frac{\mathrm{s}}{\alpha}}{1 + \frac{\mathrm{s}}{\beta \alpha}}, \alpha > 0, \ \beta > 1 \\ \mathrm{K}(\mathrm{s}) &= \frac{\mathrm{s} + \alpha}{\left(\mathrm{s} + \beta \alpha\right)} \end{split}$$

Also given that, the maximum phase lead occur at,

$$(\omega_m) = \sqrt{\alpha(\alpha\beta)} = 4$$
 rad/sec  
 $\alpha\sqrt{\beta} = 4$  ...(i)

#### 6.50 Control Systems

As we know that

Amplification, 
$$M = -20 \log_{10} \left[ \frac{\alpha \sqrt{\beta}}{\alpha} \right]$$
  
 $6 = 20 \log_{10} \left[ \sqrt{\beta} \right] \left( \begin{array}{c} \therefore \text{ From (i)} \\ \therefore M = 6 \end{array} \right)$   
 $6 = 20 \log_{10} \left[ \frac{4}{\alpha} \right]$   
So,  $\alpha = 2$   
Now, from eq. (i);  
 $2\sqrt{\beta} = 4$   
 $\Rightarrow \qquad \beta = 4$ 

#### NUMERICAL TYPE QUESTIONS

1. The characteristic equation is given by 1 + GH = 0or  $1 + \frac{(s-5)}{s+4} K = 0$ or s + 4 + (s-5) K = 0or s(K+1) + 4 - 5K = 0This single root is given by  $s = \frac{5K-4}{K+1} \le 0 \text{ for stability, since } K \ge 0$ or  $5 K - 4 \le 0, \qquad \text{since } K \ge 0,$ 

or 
$$K \le \frac{4}{5}$$
 for stability

- **2.** True
- **3.** One

$$f(s) \equiv s^{3} - 2s + 2 = 0$$
  
$$f(-2) = -8 + 4 + 2 = -ve$$
  
$$f(-1) = -1 + 2 + 2 = +ve$$

There is one real root between -2 and -1



$$\therefore f(s) \text{ is maximum when } s = -\sqrt{\frac{2}{3}}$$
  
and minimum when  $s = \sqrt{\frac{2}{3}}$   
Maximum value of  $f(s) = 2 + \sqrt{\frac{32}{27}}$   
Minimum value of  $f(s) = 2 - \sqrt{\frac{32}{27}} = + \text{ ve}$ 

Graph cuts the X-axis at one point only. No other real root.

4. Roots

5. From the characteristic equation

1 + G(s) H(s) = 0

$$1 + \frac{(s+\alpha)}{s^3 + (1+\alpha)s^2 + (\alpha-1)s + (1-\alpha)} = 0$$

: 
$$s^3 + (1 + \alpha)s^2 + \alpha s + 1 = 0$$

By Routh Hurwitz criteria

$$(1 + \alpha) \alpha > 1$$
  
 $(\alpha^2 + \alpha - 1) > 0$   
∴ α = 0.618 & -0.618

But for system to be stable  $\alpha = 0.618$ 

$$G(s) = \frac{K\left[1 + \frac{S}{2}\right]\left[1 + \frac{S}{4}\right]}{S\left[1 + \frac{S}{8}\right]\left[1 + \frac{S}{24}\right]\left[1 + \frac{S}{36}\right]}$$

Now comparing with given transfer function

$$a = \frac{1}{4}; b = \frac{1}{24}$$

For finding K:



 $\mathbf{K} = (\omega)n$ : where *n* is no. of poles from the given plot  $\therefore \qquad \mathbf{K} = (8)^1$ 

Now, 
$$\frac{a}{bk} = \left(\frac{1}{24} \times 8\right) = 0.75$$

7. 
$$T(s) = \frac{C(s)}{R(s)} = \frac{4}{s^2 + 0.4s + 4}$$
  
 $\Rightarrow \frac{R(s) - C(s)}{R(s)} = \frac{E(s)}{R(s)} = \frac{s^2 + 0.4s}{s^2 + 0.4s + 4}$ 

Steady-state error

$$e_{ss} = \lim_{s \to 0} s E(s) = \lim_{s \to 0} \frac{s R(s)(s^2 + 0.4s)}{s^2 + 0.4s + 4}$$
  
e, 
$$R(s) = \frac{1}{s}$$

Here,

 $e_{ss} = \lim_{s \to 0} \frac{s^2 + 0.4s}{s^2 + 0.4s + 4} = 0$ **8.** Characteristic equation is, 1 + G(s)H(s) = 0 $\Rightarrow$ 1 + G(s) = 0(:: H(s) = 1)

$$\Rightarrow 1 + \frac{k}{s(s+2)(s^2+2s+2)} = 0$$

 $\Rightarrow s^4 + 4s^3 + 6s^2 + 4s + k = 0$ 

Constructing Routh- array, we have

$$\begin{array}{c|cccc} s^4 & 1 & 6 & k \\ s^3 & 4 & 4 & 0 \\ s^2 & 5 & k \\ s^1 & \frac{20-4k}{5} & 0 \end{array}$$

For the closed loop system to be marginally 20 - 4k = 0stable, k = 5 $\Rightarrow$ 

 $g(t) = e^{-2}[\sin 5t + \cos 5t]$ 9. Taking lapcase transform of g(t),

$$G(s) = \frac{5}{(s+2)^2 + 5^2} + \frac{s+2}{\{s+2\}^2 + 5^2}$$

 $G(0) = \frac{5}{2^2 + 5^2} + \frac{2}{2^2 + 5^2} = \frac{7}{29}$ 

For DC gain,  $|G(s)|_{s=0}$ 

10.

*:*..

$$H(s) = \frac{1}{(s+1)}$$
Put  $s = j\omega$ ,  $H(j\omega) = \frac{1}{j\omega+1}$   
 $|H(j\omega)| = \frac{1}{\sqrt{\omega^2+1}}$   
 $\because$  input  $x(t) = \cos(t)$   
Here  $\omega = 1$  rad/sec  
and  $|x(t)| = 1$   
Hence, steady state output  
 $y(t) = |x(t)| \times |H(j\omega)|_{\omega=1} \cos[t + \angle H(j\omega)]$   
 $A = |x(t)| \times |H(j\omega)|_{\omega=1}$ 

$$A = \frac{1}{\sqrt{2}} = 0.707$$

So, the value of A is 0.707.

11. 
$$E = \int_{C} \overline{F} \cdot d\overline{r}$$
  
Here  $\overline{F} = 5xz\hat{i} + (3x^{2} + 2y) - \hat{j} + x^{2}2\hat{k}$   
 $= \int_{C} 5xzdx + (3x^{2} + 2y)dy + x^{2}zdz$   
Put  $x = t$ ,  
 $y = t^{2}$ ,  
 $z = t$ ,  
 $t = 0$  to 1  
 $dx = dt$   
 $dy = 2t dy$ ,  
 $dz = dt$   
 $= \int_{0}^{1} 5t^{2}dt + (3t^{2} + 2t^{2})2tdt + t^{3}dt$   
 $= \int_{0}^{1} (5t^{2} + 11t^{3})dt$   
 $= \left[\frac{5t^{3}}{3} + \frac{11t^{4}}{4}\right]_{0}^{1}$   
 $= \frac{5}{3} + \frac{11}{4}$   
 $= \frac{53}{12} = 4.41$ 

12. From the given fig.

Forward path transfer function,

$$G(s) = \frac{Ke^{-s}}{s}$$
  
Given, Phase margin = 30°

Phase margin =  $180^\circ + \phi$ or  $30^\circ = 180^\circ + \phi$  $\phi = -150^{\circ}$ *.*..  $\phi = \angle \mathbf{G}(j\omega)_{\omega = \omega_{gc}}$ Now [where,  $\omega_{gc}$  = gain crossover frequency]  $\angle G(j\omega) = -90^{\circ} - 57.3 \ \omega^{\circ}$ At  $\omega = \omega_{gc}$ 

$$|\mathbf{G}(j\omega)| = 1$$

or 
$$\frac{K \times 1}{\omega} = 1 \Rightarrow \omega = \omega_{gc} = K \text{ rad/sec.}$$
  
 $\therefore \qquad \angle G(j\omega)_{\omega = \omega_{gc}} = -90^{\circ} - 57.3 K$ 

$$-90^{\circ} - 57.3 K = -150^{\circ}$$
  
-57.3 K = -60°  
∴ K =  $\frac{60}{57.3} = 1.047$ 

Hence the value of K is 1.047

#### 6.52 Control Systems

13.

$$G(s) = \frac{-s+1}{s+1}$$

System output,

$$Y(s) = G(s) \cdot \frac{1}{s}$$
  
=  $\frac{-s+1}{s+1} \cdot \frac{1}{s}$   
=  $\frac{1}{s} - \frac{2}{s+1}$   
 $y(t) = u(t) - 2e^{-t}u(t)$   
 $\therefore$   $y(1.5) = 1 - 2e^{-1.5}$   
=  $1 - 0.446 = 0.554$ 

So the value of the response of the system at *t* =1.55 *e* is 0.554.

14. 
$$G(s) = \frac{1}{(s+1)(s+2)}$$
  
H(s) = 1

Steady state error due to unit step input is

$$\begin{split} \mathbf{e}_{ss} &= \frac{A}{1 + \mathop{\mathrm{Lt}}_{s \to 0} \mathbf{G}(s)} \\ &= \frac{1}{1 + \mathop{\mathrm{Lt}}_{s \to 0} \frac{1}{(s+1)(s+2)}} = \frac{1}{1 + \frac{1}{(0+1)(0+2)}} \\ &= \frac{1}{1 + \frac{1}{2}} = \frac{2}{3} = 0.66 \end{split}$$

**15.** Steady-state error =  $e_{ss} = 1 - 0.8 = \frac{1}{1 + K_p}$ 

$$1 + K_{p} = \frac{1}{0.2} = 5 \Longrightarrow K_{p} = 4$$
$$K_{p} = \underset{s \to 0}{\text{Lt}} G(s) = \underset{s \to 0}{\text{Lt}} \frac{K}{(s+1)^{2}(s+2)}$$
$$4 = \frac{K}{(0+1)^{2}(0+2)} \Longrightarrow K = 8 = 4 \times 2$$

16. As given that

$$O.L.T.F = G(s) = \frac{K}{(s + a)(s - b)(s + c)}$$

$$P = 1$$

$$N = -1$$

$$N = P - Z$$

$$Z = P - N = 1 - (-1)$$

$$Z = 2$$

Hence,  $\frac{G(s)}{1+G(s)}\,$  has two poles in the right half of s-plane.

17. For the following RLC circuit, the gain at frequency 2000 rad/sec is given 26dB. We need to obtain value of R.



When the operating frequency is 2000 rad/sec  $Z_L = jwL = j \times 2000 \times 1 \times 10^{-3} = j2$ 

$$Z_{c} = \frac{-j}{\omega C} = \frac{-1}{200 \times 250 \times 10^{-6}} = -j2$$
  
In general

$$V_{o}(j\omega) = \frac{Z_{C}}{R + Z_{L} + Z_{C}} V_{i}(j\omega)$$

$$\Rightarrow \frac{V_{o}(j\omega)}{V_{i}(j\omega)} = \frac{-j2}{R + j2 - j2} (at \ \omega = 2000 \text{ r/sec})$$

$$\Rightarrow \left| \frac{V_{o}(j\omega)}{V_{i}(j\omega)} \right| = \frac{2}{R} \qquad \dots (i)$$
As given that

As given that,

$$20 \log \left| \frac{V_{o}(j\omega)}{V_{i}(j\omega)} \right| = 26$$
  

$$\Rightarrow \left| \frac{V_{o}(j\omega)}{V_{i}(j\omega)} \right| = 10^{\frac{26}{20}} \text{ (at } \omega = 2000 \text{ rad/sec)}$$
  

$$\Rightarrow \left| \frac{V_{o}(j\omega)}{V_{i}(j\omega)} \right| = 19.95 \qquad \dots (ii)$$

Equating equation (i) and (ii)  $\frac{2}{R} = 19.95$ 

$$\Rightarrow R = \frac{2}{19.95} \Rightarrow R = 0.1\Omega$$

18. As given block diagram:



As given that,

For unit step input response has damped oscillation we need to obtain damped natural frequency.

Now, we get,

The closed loop transfer function for the given diagram.

$$\frac{C(s)}{R(s)} = \frac{G(s)G(s)}{1 + G_c(s)Gp(s)} = \frac{\frac{14.4}{s(1 + 0.1s)}}{1 + \frac{14.4}{s(1 + 0.1s)}}$$
$$= \frac{14.4}{s(1 + 0.1s) + 14.4} = \frac{14.4}{s + 0.1s^2 + 14.4}$$
$$= \frac{144}{s^2 + 10s + 144} = \frac{\omega_n^2}{s^2 + 2\xi\omega_n s + \omega_0^2}$$

Now, by comparison  $\omega_n^2 = 144$ 

$$\Rightarrow \omega_{n} = 12 \text{ rad/sec}$$

$$2\xi\omega_{n} = 10$$

$$\Rightarrow \xi = \frac{10}{2\omega_{n}} = \frac{10}{24} = 0.416$$

So, damped natural frequency :

$$\omega_{\rm d} = \omega_{\rm n} \sqrt{1-\xi^2} = 12\sqrt{1-(0.416)^2} = 10.90 \,\text{rad/sec}$$
  
**19.** According to the given block diagram :



$$G_{p}(s) = \frac{2.2}{(1+0.1s)(1+0.4s)(1+1.2s)}, G_{c}(s) = k \left(\frac{1+T_{1}s}{1+T_{2}s}\right)$$

When D(s) is unit step the steady state error should be at maximum 0.1, we need to obtain minimum value of k.

Since there are 2 inputs  $(R(s),\,D(s)),\,we\,want\,E(s)$  due to  $D(s),\,so\;R(s)$  should be made 0

We know that,

$$\frac{E(s)}{D(s)} = \frac{G_p(s)}{1 + G_p G_c(s)}$$
Now,  $E(s) = \frac{-G_p(s)}{s(1 + G_p(s)G_0(s))}$   $\left(\because D(s) = \frac{1}{s}\right)$ 
Steady state error  $e(\infty) =$ 

$$\lim_{s \to 0} SE(s) = \lim_{s \to 0} \left[\frac{-G_p(s)}{1 + G_p(s)G_c(s)}\right]$$

$$\Rightarrow e(\infty) = \lim_{s \to 0} \left[ \frac{\frac{-2.2}{(1+0.1s)(1+0.4s)(1+1.2s)}}{1+k\left(\frac{1+T_1s}{1+T_2s}\right)\frac{2.2}{(1+0.1s)(1+0.4s1+1.2s)}} \right]$$
$$= \frac{+2.2}{1+2.2k}$$
Now, We need  $e(\infty) \le 0.1$   
So,  $\frac{+2.2}{1+2.2k} \le 0.1$ 
$$\Rightarrow k \ge 9.54$$
$$\Rightarrow k_{\min} = 9.54$$
Hence, the Minimum value of k is 9.54.

**20.** As given that

$$\dot{\mathbf{x}} = -\mathbf{x} + \mathbf{u}$$

y = x

u = -kx

According to the question we need to obtain value of such that closed loop pole will be at s = -2.

Closed loop pole is roots of characteristic equation, Characteristic equation in terms of state space is |sI - A| = 0

$$\begin{split} \dot{x} &= -x + u = -x - kx \quad (\because u = -kx) \\ \Rightarrow \ sX(s) &= -X(s) - kX(s) \\ \Rightarrow \ \dot{x} &= x(-k-1) \\ \Rightarrow \ X(s) \ [s+1-k] &= 1 \\ \Rightarrow \ \dot{x} &= Ax + BU \\ By \ comparison \ A &= -k-1 \\ As \ we \ know \ that, \\ Characteristic \ equation : \ |sI - A| &= 0 \\ |s - (-k-1) &= 0| \\ |s + k + 1| &= 0 \\ s + k + 1 &= 0 \\ (Determinant \ of \ a \ constant \ is \ constant) \\ k + 1 &= 2 \\ (closed \ loop \ pole \ is \ k + 1 \ and \ if \ should \ be \ at \ 2) \end{split}$$

k = 1